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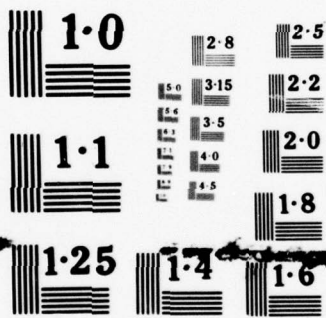
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THESIS

ADAPTATION OF THE IMPROVED ANTIAIRCRAFT  
ARTILLERY SIMULATION COMPUTER PROGRAM (P001)  
FOR USE AT THE NAVAL POSTGRADUATE SCHOOL  
IN AIRCRAFT COMBAT SURVIVABILITY STUDIES

by

Carl Frederick Swenson

March 1978

Thesis Advisor:

R. E. Ball

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A complete P001/PIP package and user's guide for an aircraft attrition study in the NPS Course AE 3251, Aircraft Combat Survivability, are presented.

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⑥ Adaptation of the Improved  
Antiaircraft Artillery Simulation Computer Program (Pp/p1)  
for Use at the Naval Postgraduate School  
in Aircraft Combat Survivability Studies.

by

⑩ Carl Frederick Swenson  
Lieutenant Commander, United States Navy  
B.S., Iowa State University, 1967

Submitted in partial fulfillment of the  
requirements for the degree of

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⑪ March 1978

Author:

C. Swenson

Approved by:

R. E. Ball

Thesis Advisor

E. R. Bell

Chairman, Department of Aeronautics

G. J. Haltiner

Dean of Science and Engineering

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## ABSTRACT

↙ The Air Force Armament Laboratory Antiaircraft Artillery Simulation Computer Program (P001), as modified by Calspan Corporation, was adapted for use on the Naval Postgraduate School IBM 360/65 computer and a preprocessor program (PIP) for P001 was written to facilitate data input to P001 and to complement the P001 output.

The modifications required to convert the Calspan modified P001 from a Control Data Corporation computer program to an NPS IBM 360/65 computer program are described herein. In addition, aircraft characteristics and P001 scenario assumptions, as well as the various P001 Input Program (PIP) options and capabilities, are discussed.

A complete P001/PIP package and user's guide for an aircraft attrition study in the NPS Course AE 3251, Aircraft Combat Survivability, are presented. ↗

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## I. INTRODUCTION

The Air Force Armament Laboratory (AFATL) has developed an antiaircraft artillery (AAA) simulation computer program called P001 which is the present standard program for conducting survivability assessments of aircraft in a hostile AAA environment. P001 is used throughout the aircraft industry and is the aircraft attrition program required by the Department of the Navy MIL-STANDARD-2072(AS), SURVIVABILITY, AIRCRAFT; ESTABLISHMENT AND CONDUCT OF PROGRAMS FOR, August 1977.

Briefly, P001 computes the probability of kill of a target aircraft flying a user-input flight path, as a result of its being fired upon by user-selected antiaircraft artillery located at user-input locations. The technique used by P001 to accomplish this task involves:

- Computation of an aim point with consideration of the errors that can arise therein.
- Simulation of the firing process and the sources of error in the firing process.
- Combination of all the effects of random error into one total projectile trajectory distribution.
- Location of the user-input vulnerable area of the aircraft within the total trajectory distribution.
- Computation of the probability of kill.

P001 has been used in NPS Course AE 3251, Aircraft Combat Survivability, to illustrate the interaction of the various



elements that comprise the aircraft combat survivability problem in a hostile AAA environment. The scenario consists of a typical Naval aircraft on an attack mission. The aircraft's target is the bridge shown in Fig. 1. The student must select a flight path to the target and also the location of the defending AAA. P001 is used to determine the probability of survival of the aircraft.

Use of P001 as an educational tool in aircraft combat survivability studies is very effective since it requires a knowledge of the techniques for calculating aircraft vulnerable areas, as well as the basics of the interaction between the threat, the environment and the target aircraft. Some of the interaction parameters include aircraft vulnerable area, speed, altitude, location and aspect angle with respect to the threat, and aircraft maneuver characteristics; the effect of terrain, target altitude and range on projectile performance; and the antiaircraft artillery threat envelope.

The input to P001 requires many time consuming, tedious computations and a significant amount of keypunching, a use of time that does not profitably contribute to the aircraft combat survivability learning experience. In addition, the realism of the input data has a significant effect on the validity of the result and, up to now, it has not been possible to evaluate input data accuracy. Consequently, a preprocessor computer program that would significantly reduce the time required for a student to prepare the input data, as well as provide an indication as to the realism of the input data, is very desirable.

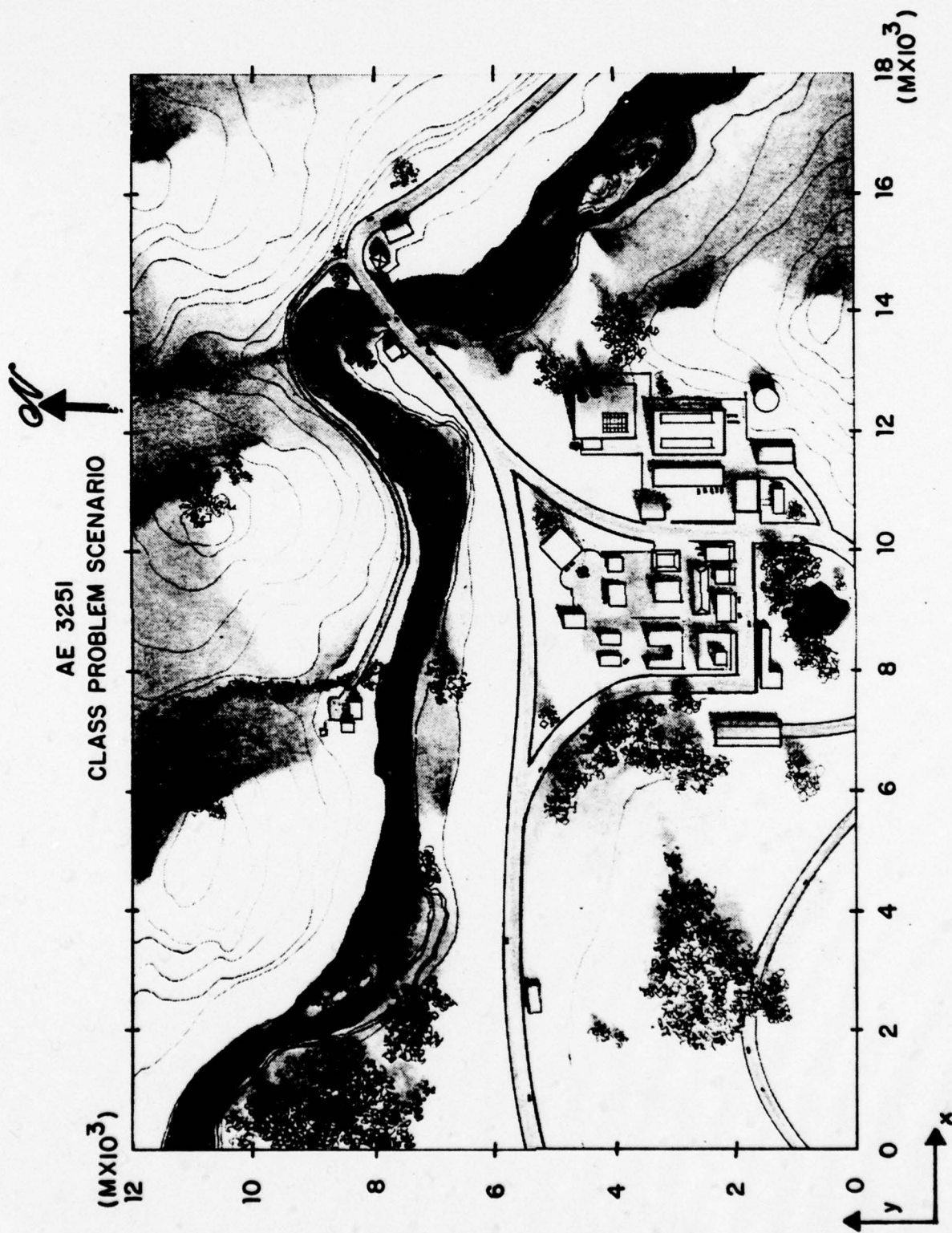


Figure 1

While the preprocessor program was being developed for P001, a version of P001 that was significantly modified by Calspan was obtained from the Air Force Flight Dynamics Laboratory. This improved version of P001 is capable of including the effects of self-contained airborne electronic countermeasures (ECM) on the acquisition/tracking process, of detection system anti-jam capabilities, and of radar beam multipath characteristics. The output of P001 was also expanded. The improved P001 was written for a Control Data Corporation computer system.

The problem solved by this thesis effort is two-fold:

- The adaptation of the improved P001 to the NPS IBM 360/65.
- The development of a P001 input program (PIP) to reduce student time required to prepare the input data required by P001 and to provide an indication of the realism of the input data.

The main body of this thesis describes the adaptation of the improved P001 to the IBM 360/65 and the details of the development of PIP. Appendix A contains a complete package for a problem in aircraft attrition to be used in AE 3251.

The following sources were heavily relied upon for information throughout the entire thesis development process:

- Antiaircraft Artillery Simulation Computer Program - AFATL Program P001 Vol. I User Manual, Air Force Armament Laboratory, Eglin AFB, Florida (Joint Aircraft Attrition Program Advanced Planning Group, September 1973).



- Antiaircraft Artillery Simulation Computer Program -  
Program P001 Program Update, Joint Technical Coordinating  
Group for Aircraft Survivability (Survivability Assessment  
Subgroup, April 1976).

- M. E. Ramaccia, ATS Working Paper No. 9, Calspan Modification to Antiaircraft Artillery Simulation, AFATL Program P001 (Calspan Corporation, Buffalo, New York, 11 August 1977).

- G. Gary Maxwell, The Development of Class Problems for a Course in Aircraft Combat Survivability (Naval Postgraduate School Master's Thesis, 1978).

## II. APPROACH

### A. ADAPTATION OF THE IMPROVED P001 TO THE NPS IBM 360/65

The adaptation of P001 from a CDC computer to the NPS IBM computer involved several hundred computer-unique alterations; that is, changes that had to be made due to the inherent differences between the CDC and IBM computer systems. These changes involved:

- Elimination of the FTNBIN, SECOND, TIMREM and DATE subroutines contained in the CDC system, but not available on the IBM 360 system.

- Substitution of the IBM "REREAD" command for the CDC "DECODE" function.

- Substitution of the IBM "A4" format field descriptor for the CDC "A10" descriptor.

- Substitution of the IBM "" format delimiter for the CDC "\*" delimiter.

- Substitution of IBM-NPS job control language (JCL) cards to accomplish the required input, output, tape usage and core size requirements.

- Extensive reconstruction of the P001 main program by creating the subroutine TOOBIG using the input, output and exits sections of the P001 main program. This permitted a scalar map size small enough for the IBM system to compile and thus avoided the "ROLL SIZE EXCEEDED" error given by the NPS compiler.

## B. DEVELOPMENT OF A PREPROCESSOR FOR P001 (PIP)

In the early stages of AE 3251 it is not desirable for the student to become heavily involved in the many and various facets and options of the P001 program. For this reason, a simple preprocessor computer program to P001 that will provide punched cards for all of the input in the proper format, requiring only a minimum of student involvement, has been developed. For example, the Calspan-modified P001 program requires an input of 196 cards with over 1,400 entries and 860 calculations for a 50 milestone flight scenario involving 7 ground weapons of 4 threat classes. On the other hand, PIP requires an input of 59 cards with 160 entries and no calculations for the same 50 milestone flight scenario and automatically punches the output that fulfills all P001 input requirements in the proper format and order. This significantly reduces the student involvement in the input process.

PIP requires only that the student provide the X, Y and Z coordinates of each of the aircraft flight path milestones as input data, avoiding the many tedious calculations required by full P001 input. In addition, the preprocessor checks the flight path to determine if it exceeds any aircraft performance limits or scenario guidelines. The aircraft built into PIP has performance characteristics approximating those of the A-7 Corsair. The cruise velocity, climb/dive schedules, acceleration/deceleration schedules, dirty/clean velocity limitations, stall velocity and "G" loading limitation values are only representative quantities and are not intended to

accurately describe the performance characteristics of an actual A-7 since the intent of PIP is only to provide representative flight characteristics that demonstrate the principles of aircraft combat survivability in an AAA scenario.

### C. FLIGHT PATH PREPROCESSING

#### 1. FORTTRAN Distance Equations

The three cartesian milestone distance components, DX, DY and DZ, are calculated in PIP by taking the difference between the X, Y and Z components of adjacent milestone locations.

Milestone distance component FORTRAN equations:

$$DX = X(I+1) - X(I)$$

$$DY = Y(I+1) - Y(I)$$

$$DZ = Z(I+1) - Z(I)$$

where,

I = ith milestone

X = milestone x-coordinate location

Y = milestone y-coordinate location

Z = milestone z-coordinate location

DX = milestone x-coordinate distance difference

DY = milestone y-coordinate distance difference

DZ = milestone z-coordinate distance difference.

#### 2. Milestone Distance Equations

The distance between successive milestones is the square root of the sum of the squares of the milestone distance components.



Milestone distance FORTRAN equations:

$$DX2 = DX**2$$

$$DY2 = DY**2$$

$$DZ2 = DZ**2$$

$$DIST = SQRT(DX2 + DY2 + DZ2)$$

where,

DIST = milestone distance

### 3. Heading and Climb Angle Equations

The aircraft heading and climb angle between milestones are calculated in PIP using standard geometrical considerations based on the relative locations of the X, Y and Z coordinates of adjacent flight path milestones.

Aircraft heading FORTRAN equation:

$$HDG(I+1) = ATAN2(DY,DX)$$

where,

HDG = aircraft heading

Aircraft climb angle FORTRAN equation:

$$CA(I) = ATAN2(DZ, SQRT(DX2 + DY2))$$

where,

CA = aircraft climb angle

### 4. Aircraft Speed Equations

The aircraft is assigned a cruise speed of between 206 and 257 meters per second (400 and 500 knots, respectively) at milestone 1 by the user. Aircraft speed at successive locations is calculated based on the altitude change between milestones, since an increase/decrease in altitude decreases/increases aircraft speed proportionately, and from a schedule



based on the present aircraft speed as compared with the initially assigned cruise speed. If the aircraft speed is found to be less/more than the assigned cruise speed, a slow acceleration/deceleration to the assigned cruise speed is assumed. The X, Y and Z components of the velocity at each milestone are then calculated based on the heading and climb angle at the milestone.

Aircraft velocity FORTRAN equations:

$$VEL(1) = CVEL$$

$$VEL(I+1) = VEL(I) - TAN(CA(I))*DIST/100 \\ + (CVEL - VEL(I))*(DIST/VEL(I))/30$$

$$VAVG = (VEL(I) + VEL(I+1))/2$$

where,

CVEL = aircraft cruise speed

VEL = aircraft velocity

VAVG = average aircraft velocity

Aircraft velocity component FORTRAN equations:

$$XYVEL = VEL(I)*COS(CA(I))$$

$$XDOT(I) = XYVEL*COS(HDG(I))$$

$$YDOT(I) = XYVEL*SIN(HDG(I))$$

$$ZDOT(I) = VEL(I)*SIN(CA(I))$$

where,

XYVEL = horizontal velocity component

XDOT = x-coordinate velocity component

YDOT = y-coordinate velocity component

ZDOT = z-coordinate velocity component.

## 5. Flight Time Equations

The time interval between flight path milestones is calculated by dividing the distance between the milestones by the average velocity between the milestones. The individual milestone time intervals are summed to provide the total time for the scenario.

Flight time FORTRAN equations:

$$T(I+1) = T(I) + \text{DIST}/\text{VAVG}$$

$$\text{DT}(I) = T(I) - T(I-1)$$

where,

$T$  = flight time (cumulative)

$\text{DT}$  = flight time between milestones (I) and (I+1)

## 6. Turn Rate, Roll Angle, "G" Loading

The turn rate required between milestones is calculated from the heading change and the time interval between the milestones. This turn rate and the aircraft speed are used to calculate the "G" loading on the aircraft caused by the turn. The roll angle required for a level turn is calculated from the turn rate and the aircraft speed.

Turn angle FORTRAN equation:

$$\text{TNANG} = \text{HDG}(I+1) - \text{HDG}(I)$$

where,

$\text{TNANG}$  = milestone turn angle.

Turn rate FORTRAN equation:

$$\text{TNRT}(I) = \text{TNANG}/\text{DT}(I)$$

where,

$\text{TNRT}$  = milestone turn rate.

Roll angle FORTRAN equation:

$$RA(I) = ATAN(TNRT(I)*VAVG/9.81)$$

where,

RA = milestone aircraft roll angle.

"G" loading FORTRAN equations:

$$ABSRT(I) = ABS(TNRT(I))$$

$$G(I) = SQRT(ABSRT(I)**2*VEL(I)**2/9.81**2 + 1)$$

where,

ABSRT = absolute value of turn rate

G = "G" loading

#### 7. P001 Stored Time Increments

The equal time increments between successive "P001 stored" positions along the flight path (not milestones) is equal to the total scenario flight time (T) divided by 1000. This time increment is required in the P001 input data card 02.

#### 8. Probability of Kill Accumulation Periods

The total scenario flight time (T) is divided into ten equal time segments in which the probability of kill will be computed for each segment. These ten increments are required on P001 input card 06.

The values computed above are printed and punched on cards that can be used as part of the P001 input data. All data are punched in the specified order to be input to the P001 program.

D. AIRCRAFT PERFORMANCE LIMITATIONS,  
FLIGHT PATH REQUIREMENTS AND ERROR MESSAGES

PIP provides several checks on the performance requirements of the aircraft as it traverses the user-input flight path. It also checks the bombing run portion of the flight path to see if it satisfies the requirements for successful bomb drop. These checks are as follows:

1. Cruise Speed

The aircraft cruise speed is initially input by the user at a value between 206 and 257 meters per second and is changed as dictated by altitude changes and the acceleration/deceleration schedule presented in II.C. If the aircraft is assigned a cruise speed outside of the range from 206 to 257 meters per second, the following cruise speed error message is generated:

Error message: "CRUISE SPEED IS \_\_\_\_ METERS PER SECOND WHICH IS NOT WITHIN THE GIVEN LIMITS OF BETWEEN 206 AND 257 METERS PER SECOND."

2. Stall Speed

If the aircraft speed falls below 90 meters per second (175 knots), the following stall error message is generated, identifying the error, the milestone and the velocity value causing the error message:

Error message: "MILESTONE \_\_\_\_ VELOCITY IS \_\_\_\_ METERS PER SECOND. STALL OCCURS AT 90 METERS PER SECOND. DECREASE THE CLIMB ANGLE PRIOR TO MILESTONE \_\_\_\_."



### 3. "Red Line"

Prior to the bomb release point, the aircraft is "drag limited" to 260 meters per second (500 knots). After ordnance release, the drag limitation eases to permit a speed of 310 meters per second (600 knots).

### 4. "G" Loading

If the maximum "G" loading of 6 is exceeded, the following turn rate error message is generated, identifying the error, the milestone, the "G" loading and turn rate that caused the error message and denoting the corrective turn angle which will eliminate the error:

Error message: "MILESTONE \_\_\_\_ TURN RESULTS IN A TURN RATE OF \_\_\_\_ DEGREES PER SECOND WHICH RESULTS IN A G LOADING OF \_\_\_\_ WHICH IS IN EXCESS OF THE 6 G MAX LOADING. DECREASE THE TURN ANGLE AT MILESTONE \_\_\_\_ TO BELOW \_\_\_\_ DEGREES."

### 5. Minimum Altitude

If the aircraft descends to an altitude less than 61 meters prior to bomb release, the following error message is generated identifying the error, the milestone and the milestone altitude that caused the error:

Error message: "ALTITUDE AT MILESTONE \_\_\_\_ IS \_\_\_\_ METERS WHICH IS BELOW THE MIN ALT OF 61 METERS."

### 6. Maximum Altitude

If the aircraft attains an altitude greater than 457 meters prior to the "pop-up" maneuver, the following error message is generated identifying the error, the milestone and the milestone altitude that caused the error:

Error message: "ALTITUDE AT MILESTONE \_\_\_\_ IS \_\_\_\_ METERS WHICH IS ABOVE THE MAX ALT PRIOR TO POP UP OF 457 METERS."

7. Overall Maximum Altitude

If, at any time, the aircraft exceeds the overall maximum altitude of 2134 meters, the following error message is generated identifying the error, the milestone and the milestone altitude that caused the error:

Error message: "ALTITUDE AT MILESTONE \_\_\_\_ IS \_\_\_\_ METERS WHICH IS ABOVE THE MAX ALT OF 2134 METERS."

8. Minimum "pop up" Altitude

If, during the "pop up" maneuver, the aircraft fails to attain a minimum altitude of 1219 meters, the following error message is generated identifying the error and the altitude attained during the "pop up" maneuver:

Error message: "MAX ALTITUDE DURING POP UP WAS \_\_\_\_ METERS WHICH IS LESS THAN THE MINIMUM POP UP ALTITUDE OF 1219 METERS."

9. Bomb Release Heading

If the aircraft heading at the time of bomb release is greater than 5 degrees from the true heading to the target, the following error message is generated identifying the error, the aircraft heading and the true heading to the target at the time of bomb release:

Error message: "THE AIRCRAFT HEADING INTO THE BOMB RELEASE POINT IS \_\_\_\_\_. THE HEADING TO THE TARGET IS \_\_\_\_\_."

THE HEADING DIFFERENCE IS \_\_\_\_\_ WHICH IS GREATER THAN THE 5 DEGREE MAXIMUM DIFFERENCE LIMIT."

10. Target Acquisition Time

If the aircraft does not hold a heading of less than 5 degrees from the true heading to the target for a time period of at least 2 seconds on the leg immediately prior to the bomb release point, the following error message is generated identifying the error and the time duration of the leg that caused the error:

Error message: "THE LENGTH OF THE LEG IMMEDIATELY PRIOR TO THE BOMB RELEASE POINT IS \_\_\_\_\_ SECONDS IN DURATION WHICH IS LESS THAN THE MINIMUM OF 2 SECONDS."

11. Bomb Release Altitude

If the ordnance is released outside of an altitude envelope of from 305 to 914 meters, the following error message is generated identifying the error and the altitude at bomb release:

Error message: "THE BOMB RELEASE ALTITUDE IS \_\_\_\_\_ METERS WHICH IS NOT IN THE BOMB RELEASE ALTITUDE RANGE OF BETWEEN 305 TO 914 METERS."

12. Bomb Release Range

If the ordnance is released at a distance greater than 1000 meters from the target, the following error message is generated identifying the error and the distance from the target at the time of bomb release:

Error message: "THE BOMB WAS RELEASED AT A DISTANCE OF \_\_\_\_\_ METERS FROM THE TARGET WHICH IS IN EXCESS OF THE 1000 METER MAXIMUM BOMB RELEASE RANGE."



13. Gun Location Input Error

If the option is chosen to input the gun locations, but no gun location information is part of the input data or not all of the gun locations are specified, the following error message is generated identifying the error. Program execution terminates after the error message is printed.

Error message: "GUN EMPLACEMENT DATA WAS SPECIFIED AS PART OF THE INPUT DATA; HOWEVER, EITHER NO GUN EMPLACEMENT DATA IS PART OF THE INPUT OR ALL SIX GUN LOCATIONS WERE NOT SPECIFIED. EXECUTION TERMINATES."

14. Anti-jam Error

If the anti-jam option is specified, but no jammer is in operation, the following error message will be generated and the anti-jam function will be "turned off":

Error message: "THE ANTI-JAM FUNCTION IS SPECIFIED; HOWEVER, THE JAM FUNCTION IS NOT SPECIFIED. THE ANTI-JAM FEATURE HAS BEEN TURNED OFF."

15. Type 3 Gun Range Error

If a type 3 weapon is located within 3,000 meters of the center of the bridge, the following error message will be generated, identifying the error, the position of the gun that caused the error and the actual distance from the target of the gun:

Error message: "GUN TYPE 3 LOCATED AT X: \_\_\_\_ Y: \_\_\_\_ IS \_\_\_\_ METERS FROM THE TARGET WHICH IS LESS THAN THE MINIMUM DISTANCE OF 3000 METERS."



16. Zero Power Jammer Error

If the jamming function has been specified, but the jammer has been assigned a power of zero, the jammer function is "turned off" and the following error message is generated, identifying the error and the fact that the jammer has been "turned off":

Error message: "THE JAM FUNCTION IS SPECIFIED, BUT THE JAMMER POWER IS SPECIFIED AS ZERO. THE JAM FUNCTION HAS BEEN TURNED OFF."

17. Maximum Power Jammer Error

If the jammer has been assigned a power greater than 1000 watts, the following error message is generated, identifying the error. The jammer power will then be limited to 1000 watts.

Error message: "THE SPECIFIED JAMMER POWER IS GREATER THAN 1000 WATTS AND HAS BEEN LIMITED TO 1000 WATTS."

E. PROGRAM OPTIONS

PIP provides the following electronic warfare options:

1. Jamming Option

An airborne jammer aboard the target aircraft is utilized to degrade radar acquisition/tracking capabilities.

2. Anti-jam Option

Ground weapons that have an anti-jam capability use it to partially nullify the effects of the airborne jammer.

3. Multipath Option

The performance of all radar units which are susceptible to multipath effects is appropriately degraded.

PIP also provides for any combination of the following input/output options:

4. List the P001 Input Deck

A complete listing of all required cards for input to the P001 program is provided. The green "JOB" card and the orange final "EOF" card are not part of this listing. These two cards are the only cards that must be provided by the student to run the P001 program with the PIP output.

5. Punch the P001 Input Deck

A complete punched input deck in the proper format and order to run P001 is provided by this option. Again, no "JOB" or "EOF" card is provided.

6. Plot the P001 Scenario

A plot of the basic geographical features of the scenario, the aircraft flight path and milestone locations, the bomb release point, the gun emplacement locations and the threat radius for each weapon (coded as to weapon type) are provided by this option. Fig. 2 shows a typical PIP scenario plot.

7. Extended Printout Option

An extended printout of the results of the P001 analysis will be provided as output.

8. Gun Location Option

The locations of six of the seven guns in the scenario may be input to the program or preset gun locations may be utilized, as desired.

# AE 3251 P001 SCENARIO

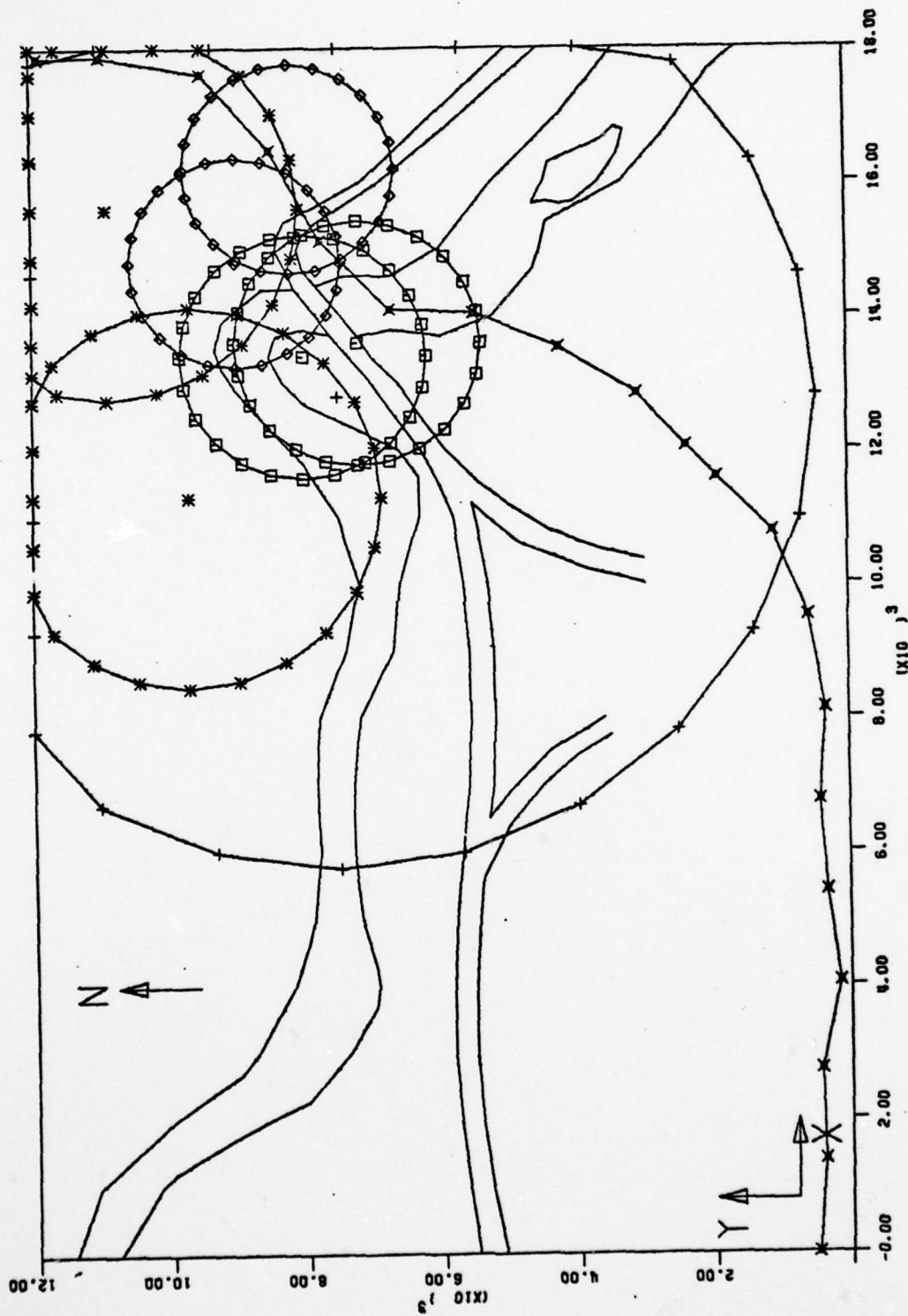


Figure 2 Typical PIP Scenario Plot

In all cases, messages are generated specifying which options were or were not chosen for each execution of PIP.

#### F. PIP INPUT DATA DEFAULT VALUES

In order to significantly reduce student involvement in the P001 input process, PIP assigns predetermined values to many of the options available under the full P001 input. These default values, over which the student has little or no control, were chosen to provide values that are representative of the typical attack situation simulated by the class problem scenario. The following is a list of these default values as they pertain to each P001 input card:

##### 01 Card: Output Header Information

The output header information is assigned "Aircraft Combat Survivability Scenario" by PIP.

##### 02 Card: Initial Flight Path Data

JMODE = 0: The milestone data are read from cards for one milestone at a time.

KMODE = 12: Flight path stored position data will be printed at every 12th position along the flight path.

TMIN = 0: The time at the beginning of the flight path is assigned a value of zero.

TMAX: The time at the end of the flight path is computed by PIP.

DTFPA: The time increment between successive stored positions along the flight path is calculated by PIP as  $TMAX/1000$ .



XR, YR = 0: An x, y reference location in the Flight Path Coordinate System. XR and YR are coordinates in the Flight Path Coordinate System of the point located at XT, YT and ZT in the General Reference Coordinate System, as shown in Fig. 3.

XT, YT = 0: The x, y coordinates in the General Reference Coordinate System of the point located at XR, YR in the Flight Path Coordinate System, as shown in Fig. 3.<sup>1</sup>

PSI = 0: The rotational angle required to rotate the Flight Path Coordinate System into the General Reference Coordinate System (positive for counter-clockwise rotation).

ZT = 0: Vertical correction factor to be added to each point of the flight path, as shown in Fig. 3.<sup>1</sup>

#### 2A Card: Flight Path Milestone Input

All data on Card 2A is calculated by PIP based on the cruise speed and milestone coordinates provided by student input.

#### 03 Card: Ground Weapon Complex Coordinates

If the preset weapon location option is chosen, the seven ground weapons used in the class scenario are assigned in the following locations by PIP:

---

<sup>1</sup> Setting XR, YR, XT, YT, ZT and PSI equal to zero results in the coincidence of the Flight Path Coordinate System and the General Reference Coordinate System.

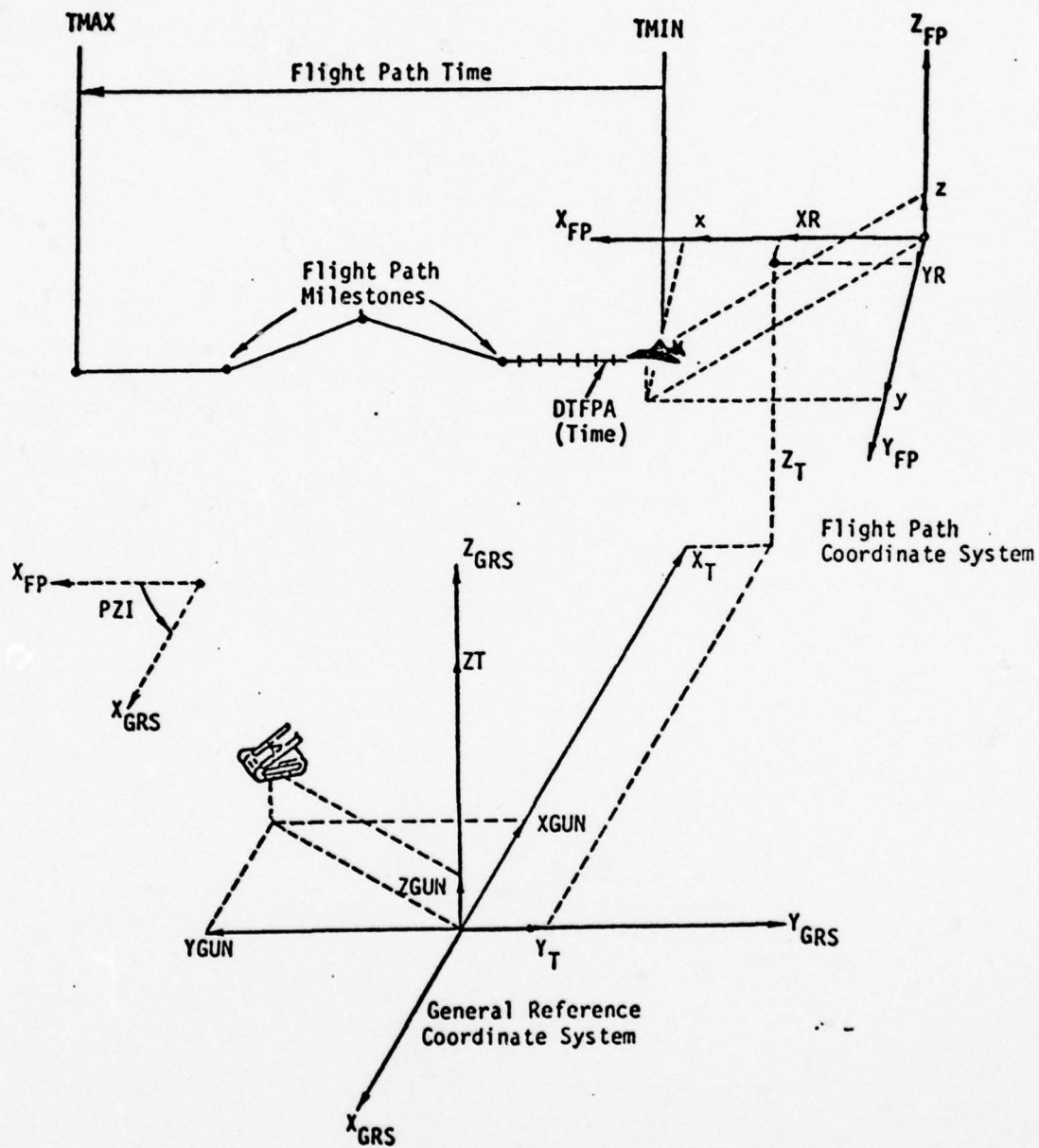


Figure 3 Relationship of the General Reference and Flight Path Coordinate Systems

| <u>Gun</u> | <u>XGUN</u> | <u>YGUN</u> | <u>ZGUN</u> |
|------------|-------------|-------------|-------------|
| 1          | 14,800      | 9,000       | 40          |
| 2          | 16,200      | 8,200       | 40          |
| 3          | 13,600      | 7,200       | 20          |
| 4          | 13,400      | 8,000       | 20          |
| 5          | 11,300      | 9,700       | 50          |
| 6          | 15,600      | 10,900      | 90          |
| 7          | 12,800      | 7,500       | 20          |

where XGUN, YGUN and ZGUN are as shown in Fig. 3. Fig. 2 shows the weapon locations.

If the preset weapon location option is not chosen, all values on Card 03 are assigned by student input.

#### 04 Card: Ground Weapon Characteristics

Each ground weapon is defined by six parameters. These parameters are:

IGT: Ground weapon type.

IEM: Fire control operation mode.

ICB: Number of barrels of the ground weapon to be fired in a simultaneous manner, where  $ICB \times ISB$  is the number of barrels per weapon.

IGL: Number of ground weapons located in the ground weapon complex. ( $ICB \times ISB \times IGL$  is the number of barrels at one location.)

CIRCLE: Radius of the circle of the ground weapon complex. If there is only weapon in the ground weapon complex, CIRCLE = 0.0.

The default values selected for the six parameters for each of the seven weapons are as follows:

|        | <u>IGT</u> | <u>IEM</u> | <u>ICB</u> | <u>ISB</u> | <u>IGL</u> | <u>CIRCLE</u> |
|--------|------------|------------|------------|------------|------------|---------------|
| Gun 1: | 1          | 1          | 1          | 1          | 1          | 0.0           |
| Gun 2: | 1          | 1          | 1          | 1          | 1          | 0.0           |
| Gun 3: | 2          | 1          | 1          | 1          | 1          | 0.0           |
| Gun 4: | 2          | 1          | 1          | 1          | 1          | 0.0           |
| Gun 5: | 3          | 4          | 4          | 1          | 1          | 0.0           |
| Gun 6: | 3          | 3          | 4          | 1          | 1          | 0.0           |
| Gun 7: | 5          | 3          | 2          | 1          | 1          | 0.0           |

05 Card: Ground Weapon Complex Density Factors

IF5 = 0: Ground weapon complex density factors are not printed.

NRHOS = 1: The number of ground weapon density factors equals one.

RHO(1) = 1.0; RHO(2) through RHO(9) = 0.0: Generally, RHO = (number of possible ground weapon complexes in the engagement divided by the number of possible ground weapon complex locations in the scenario).

06 Card: Flight Path  $P_k$  Accrual Time Intervals ( $P_k$  = probability of kill)

IF6 = 1: Flight path  $P_k$  accrual time intervals are printed.

NTINTS = 9: One less than the total number of flight path time intervals to be considered. P001 adds one additional time interval for  $P_k$ 's accumulated from NTINTS to infinity.



TINTER(1) through TINTER(9): TINTER values are assigned by PIP. Each TINTER is an increment representing 1/10th of the total flight path time.

07 Card: Aircraft Vulnerable Area Table Title

- ICARD = a. "Vulnerable Area Table vs Type 1 and 2 Weapons",  
b. "Vulnerable Area Table vs Type 3 Weapons",  
c. "Vulnerable Area Table vs Type 5 Weapons"

7A Card: Aircraft Vulnerable Area Tables

The values assigned by PIP to the three aircraft vulnerable area tables representing the three senario threat classes are given in Tables I, II and III respectively.

08 Card: Ground Weapon Reaction and Tracking Times

Card 08 is omitted by PIP. Values are assigned within P001.

09 Card: Ground Weapon Parameters

Card 09 is omitted by PIP. Values are assigned within P001.

9A Card: Ground Weapon Parameters

Card 9A is omitted by PIP. Values are assigned within P001.

10 Card: Ground Weapon Projectile Parameters

Card 10 is omitted by PIP. Values are assigned within P001.

11 Card: Logical Unit Input Option

Card 11 is omitted by PIP. Logical Unit 5 is assigned for input within P001.

[illegible]

TABLE I. Aircraft Vulnerable Area Table (Type 1 and 2 Weapons)

[illegible]

TABLE II. Aircraft Vulnerable Area Table (Type 3 Weapons)



[illegible]

TABLE III. Aircraft Vulnerable Area Table (Type 5 Weapon)



### 13 Card: Radar Multipath Parameters

IMUL: Assigned by the user on the PIP option card. If IMUL = 0, no multipath effects are considered. If IMUL = 1, multipath radar effects are taken into consideration in the P001 analysis.

IRMP: Radar type identification assigned by PIP, specifying the tracking radar. The value of IRMP indicates appropriate radar parameters within P001. The value of IRECM, assigned by PIP on Card 14, dictates the value of IRMP. The radar parameters and the relationship between IRMP and IRECM are as follows:

|        | <u>IRECM</u> | <u>IRMP</u> | <u>Beamwidth<br/>(deg)</u> | <u>Squint<br/>Angle (deg)</u> | <u>Calibration<br/>Constant</u> |
|--------|--------------|-------------|----------------------------|-------------------------------|---------------------------------|
| Gun 6: | 1            | 1           | 1.4                        | 0.5                           | 0.759                           |
| Gun 7: | 2,3          | 2           | 1.8                        | 0.6                           | 1.060                           |

Multipath radar effects do not apply to guns 1 through 5.

REFC = 0.35: Reflection coefficient. 0.35 is a typical value for terrain with vegetation.

### 14 Card: ECM (Jamming) Parameters

IJAM: Assigned by the user on the PIP option card. 0 = no jamming. 1 = jamming effects considered in P001 analysis.

IP = 5: Print every 5th value in J/S printout.

IJ = 0. Therefore, GAINJ is the antenna gain of the jammer.

GAINJ = 1.0: The antenna gain of the jammer is 1.0.

PJW: Jammer power, assigned by the user on the PIP option card.

PLEN = 1.0E-06: The length of the jammer cover pulse is 1 microsecond, a standard value.

IX = 1: A radar cross section table is provided by PIP. (Table IV).

XSEC = 0: XSEC is not used if IX = 1. If used (IX = 0), a constant cross section of XSEC m<sup>2</sup> is used.

CALX = 1: The radar cross section table is not scaled.

IRECM: The value of IRECM defaulted by PIP depends upon the gun type and mode. IRECM calls up certain radar parameters from a data statement within P001. IRECM values and the relationship with the gun type and mode are as follows:

| <u>Gun Type</u><br>(IGT) | <u>Mode</u><br>(IEM) | <u>Radar ID</u><br>(IRECM) | <u>Gain</u><br>(RGDB) | <u>Power</u><br>(PRW) | <u>Frequency</u><br>(FREQ) | <u>SJTMAX</u> |
|--------------------------|----------------------|----------------------------|-----------------------|-----------------------|----------------------------|---------------|
| 1                        | 1                    | N/A                        | N/A                   | N/A                   | N/A                        | N/A           |
| 2                        | 1                    | N/A                        | N/A                   | N/A                   | N/A                        | N/A           |
| 3                        | 3,4                  | 1                          | 40.0 dB               | 150,000               | 15.1E9                     | 3.0 dB        |
| 5                        | 3                    | 2 (no AJ)                  | 38.5 dB               | 175,000               | 9.38E9                     | 1.5 dB        |
| 5                        | 3                    | 3 (AJ)                     | 38.5 dB               | 175,000               | 9.38E9                     | 17 dB         |

SJTMAX: Assigned by PIP as indicated above. SJTMAX is the threshold where tracking errors become significant.

RGDB, PRW, FREQ: Assigned within P001.

#### 12 Card: Print Options for Output

IPRINT(1) through IPRINT(7): If the extended output option is chosen by the user on the PIP option card, IPRINT(1) through IPRINT(7) = 1 and an extended printout of the result of the P001 analysis is obtained. If the extended output

|          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          |          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TABLE IV. Aircraft Radar Cross-section Table

option is not chosen, IPRINT(1) through IPRINT(7) = 0 and a summary of the P001 analysis is printed as output from P001.



### III. SUMMARY AND CONCLUSIONS

The use of the P001 Input Program (PIP) to provide all required input to P001 greatly reduces the student involvement in the aircraft combat survivability scenario computer input procedure, freeing him from time consuming, tedious computations and keypunching which do not contribute profitably to the aircraft combat survivability learning experience. In addition, PIP provides an indication as to the realism of the input data, thus contributing to the validity of the result of the P001 analysis.

Introduction of the Calspan Improved P001 Computer Program into aircraft combat survivability studies provides the class problem in survivability assessment with ECM (jamming), ECCM (anti-jam) and radar multipath features which are realistic parameters to be considered in any current aircraft combat survivability situation.

As developed, the PIP target aircraft performance parameters are those of a "typical" Navy attack aircraft, having flight characteristics that are realistic, but which can not be used to describe the performance of any specific aircraft. As a future project, specific aircraft flight performance parameters and equations could be added to PIP in the form of a flight path generator program to give the input program the added capability of simulating the flight path of a specific aircraft.

APPENDIX A

AIRCRAFT COMBAT SURVIVABILITY PROBLEM

This Appendix contains a complete package for a class problem in aircraft attrition in a hostile AAA environment for AE 3251, Aircraft Combat Survivability.

AE 3251

AIRCRAFT COMBAT SURVIVABILITY

A STUDY

of

AIRCRAFT ATTRITION

in a

HOSTILE AAA ENVIRONMENT

NAVAL POSTGRADUATE SCHOOL  
MONTEREY, CALIFORNIA

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## I. INTRODUCTION

This aircraft attrition study is designed to present the student with an opportunity to see first hand how the survivability of an aircraft can be evaluated in a given combat scenario. The methods employed in this study are those used by both industry and government when making crucial decisions in the survivability design of an aircraft weapon system. In this study, a computer program named P001 (AFATL Antiaircraft Artillery Simulation Computer Program) will be used to (1) simulate the flight of a typical Naval attack aircraft through a hostile antiaircraft artillery (AAA) environment and (2) compute the aircraft probability of survival.<sup>1</sup>

Section II describes all of the steps necessary to complete this study. Note the flow of the survivability assessment process from a physical description of the aircraft to a determination of its capabilities to withstand certain threat levels (i.e., its vulnerability), to a scenario in which both offensive and defensive strategies must be employed, to the final phase of simulating flight through the hostile environment and computing probabilities of survival using a modern, state-of-the-art computer program.

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<sup>1</sup> The Navy specifies the use of P001 in all non-nuclear survivability assessments in MIL-STANDARD-2072(AS), SURVIVABILITY, AIRCRAFT; ESTABLISHMENT AND CONDUCT OF PROGRAMS FOR, August 1977.

The student should develop a good appreciation for the magnitude of the survivability problem by keeping the above survivability assessment process in mind when working each part of the analysis.

## II. PROBLEM DEFINITION

- A. You are going to conduct a survivability assessment of a familiar Naval aircraft, shown in Figs. 1 and 2, on a typical attack mission to destroy the bridge shown in Fig. 3.
- B. The class will be divided into groups of four, with two members in each group on the blue team and two members on the red team.
- C. Each team will independently determine the vulnerable areas of the aircraft to the specified threat in the six major views.
- D. Each team will use P001 to determine the survivability of the aircraft in the class problem scenario, as follows:
  - 1. Each team will select a flight path to the bridge according to the rules of the scenario given in Section IV. Keep this path a secret.
  - 2. Each team will also select the locations of six AAA emplacements that will defend the bridge against an air attack. Locate the weapons according to the order of battle given in Section IV. Keep these locations secret, also.
  - 3. Each team will conduct an attack against the other team in the group.
  - 4. The input data cards for the computer run for the blue team attacking the bridge defended by the red team

will consist of the flight path of the blue aircraft flying through the AAA emplacements selected by the red team.

5. The input data cards for the computer run for the red team against the blue team will consist of the flight path of the red aircraft flying through the AAA emplacements selected by the blue team.
- E. May the best team win. A small prize will be awarded to the team whose aircraft has the highest probability of survival against their opponent's weapon distribution.
- F. Additional runs will be made against a preset AAA distribution to investigate the effects of ECM, ECCM, jinking, etc., on the survivability of the aircraft.



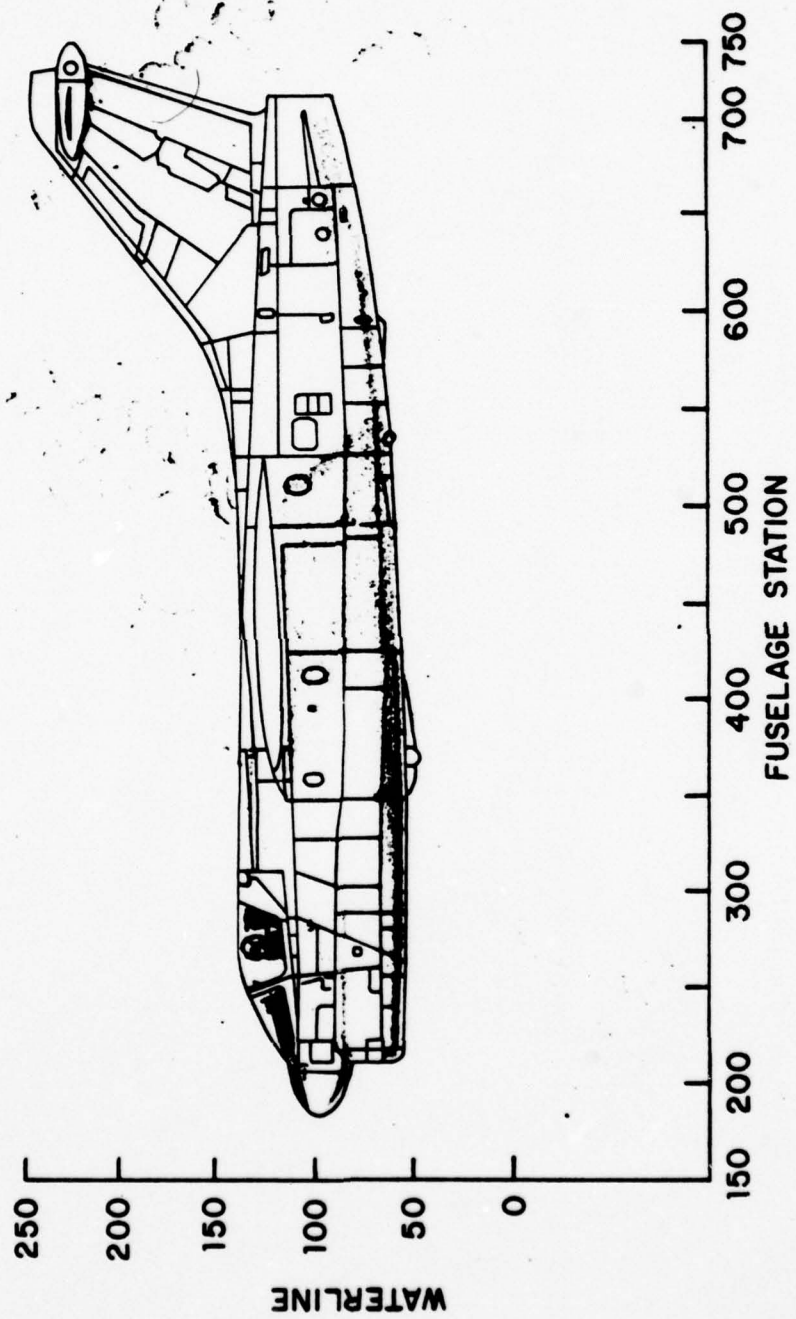


Figure 1

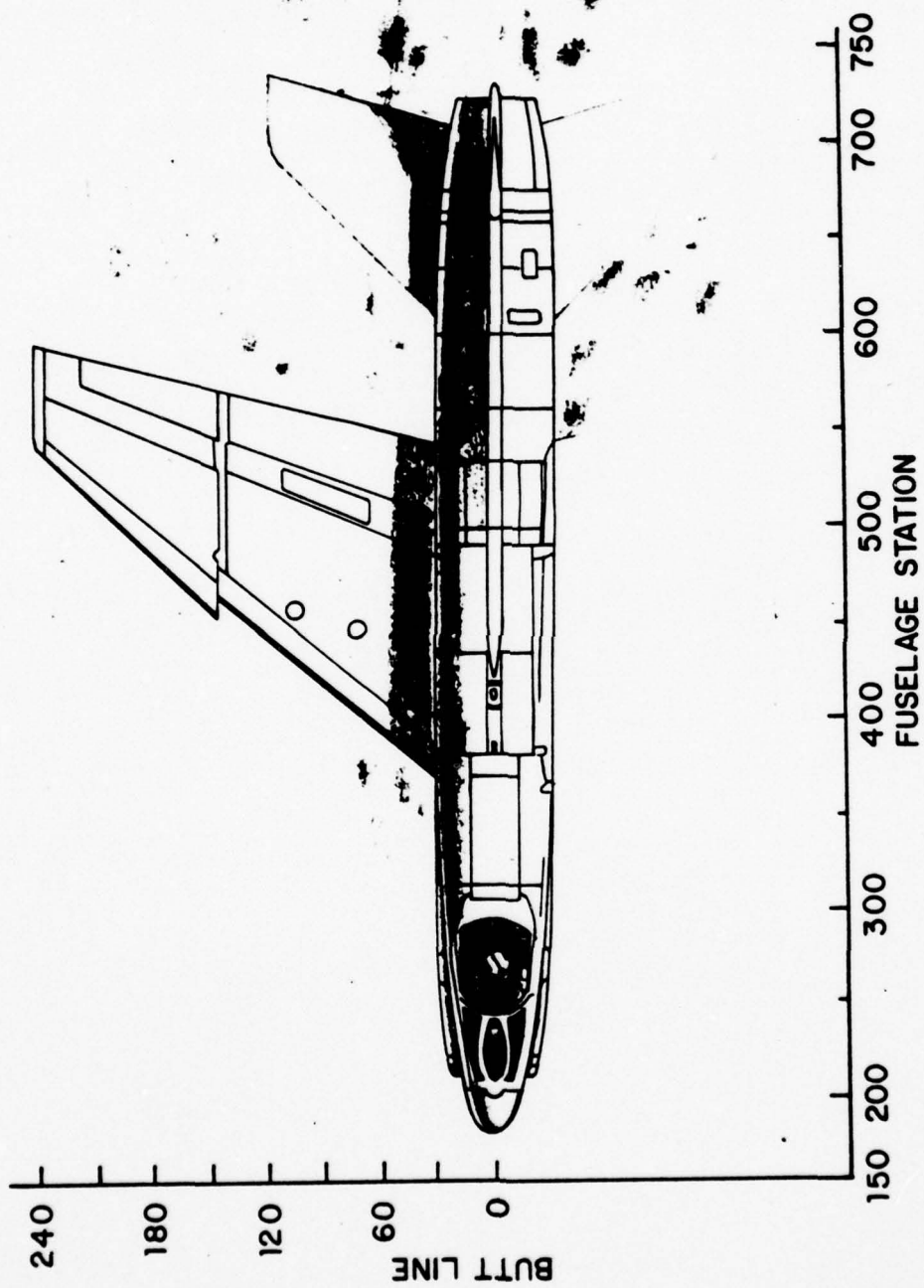


Figure 2

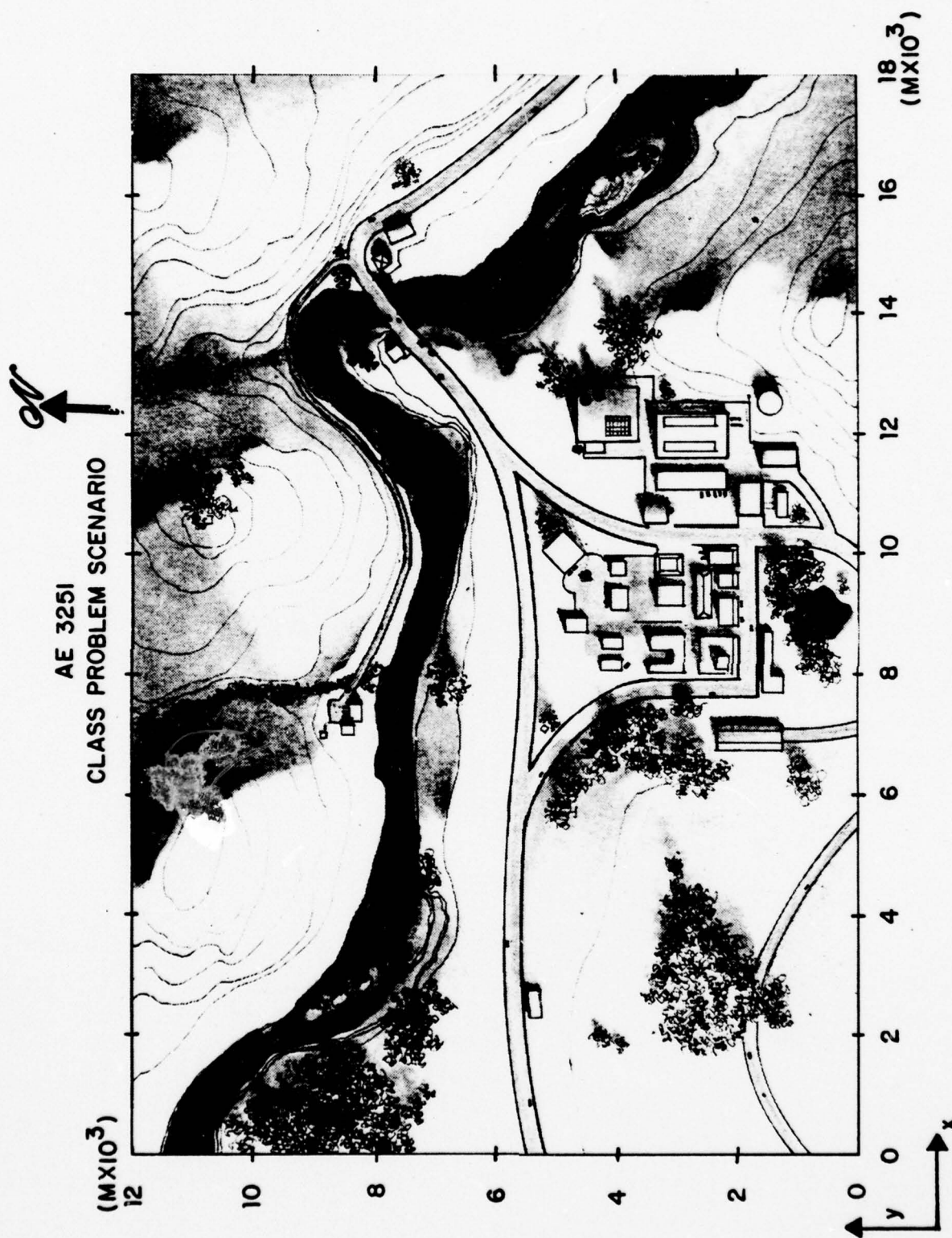


Figure 3

### III. VULNERABILITY ASSESSMENT DESCRIPTION

#### A. GENERAL METHODOLOGY

The six general requirements for a vulnerability assessment are discussed in detail in Chapter IV of the class text. The following data are given in support of this assessment procedure:

1. Kill Category = "A" Kill
2. Technical Description of the Aircraft - Figs. 1 and 2
3. Critical Components - Pilot, engine and fuel tanks.

a. Each of these singly vulnerable components will make a contribution to  $A_{p_i}$ , the total presented area of the aircraft.

b. The total presented area is assumed to be a "shoe box" centered around the aircraft center of gravity.

4. Damage Analysis - Determination of  $P_{K/H}$  for each of the components will be discussed in class due to the classified nature of the material.

5. Threat Types - To be discussed in class; Types I, II, III and V threats will be used in the scenario.

6. Determine Vulnerable Area - Use the equation:

$$A_v = \sum_i A_{p_i} \cdot P_{K/H_i}$$

where  $i$  = pilot, engine and fuel tanks.



## B. SPECIFIC CALCULATIONS

The vulnerability assessment may now be completed in the following manner:

1. Measure the presented area of each critical component of the aircraft shown in Figs. 1 and 2 for the top/bottom, front/rear and left/right aspects and record in Table I.

2. Calculate the  $A_{V_i}$  for each component using the given  $P_{K/H_i}$  for the appropriate aspect and striking velocity and enter it into Table I.

3. In order to use the information compiled in Table I in P001, a more complete description of the aircraft  $A_V$  with changing aspect angle must be tabulated. This is normally done in a 26 view,  $\delta$  striking velocity vulnerable area table (VAT). Fig. 4 and Table II show how the aircraft is physically divided into these 26 different views. You have tabulated in Table I the total  $A_V$  for each of the striking velocities, but only for the six major aspects. The vulnerable area of the other views can be obtained by interpolating between these six aspects. The following is a summary of the six views you have done in Table I and their relationship to the 26 views needed to describe the aircraft:

| View      | View # | Long (AZ)<br>(degrees) | Lat (Elev)<br>(degrees) |
|-----------|--------|------------------------|-------------------------|
| Bottom    | 1      | 0                      | 0                       |
|           | 2-9    | 0-315                  | 45                      |
| Tail-on   | 10     | 0                      | 90                      |
|           | 11     | 45                     | 90                      |
| STBD Side | 12     | 90                     | 90                      |
|           | 13     | 135                    | 90                      |
| Head-on   | 14     | 180                    | 90                      |
|           | 15     | 215                    | 90                      |
| Port Side | 16     | 270                    | 90                      |
|           | 17     | 315                    | 90                      |
|           | 18-25  | 0-315                  | 135                     |
|           |        |                        |                         |
| TOP       | 26     | 0                      | 180                     |

The  $A_v$ 's you have calculated will not be used in the P001 analysis. Instead, predetermined VATS for each threat type will be used in order to provide a standardized aircraft for the attrition study.

4. Turn in Table I prior to initiating a computer run for the analysis.



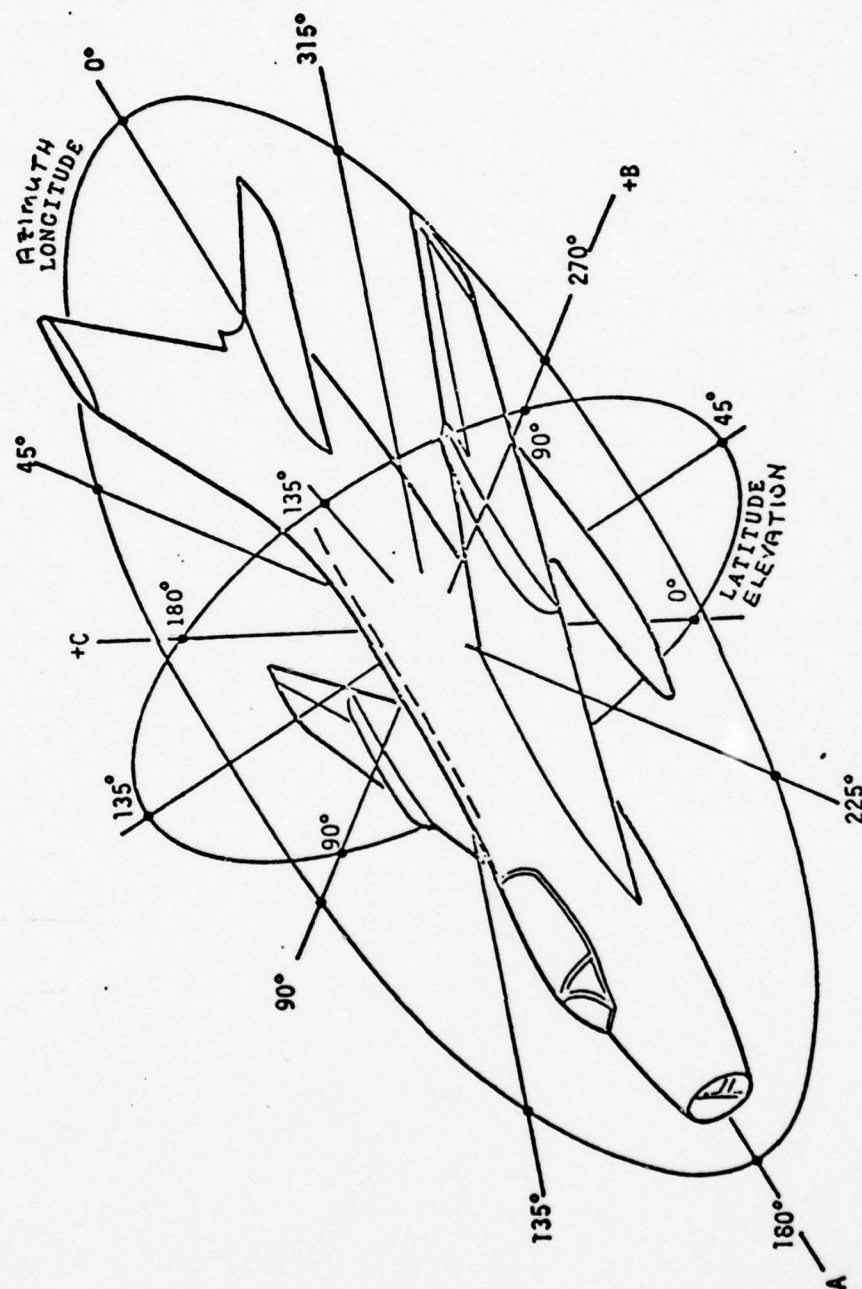


Figure 4 Longitude and Latitude of Aircraft for Vulnerable Area Computation



TABLE II VULNERABLE AREA TABLES

| Card Number | I | J | Aircraft View                 |
|-------------|---|---|-------------------------------|
| 1           | 1 | 1 | 0° Longitude, 0° Latitude     |
| 2           | 1 | 2 | 0° Longitude, 45° Latitude    |
| 3           | 2 | 2 | 45° Longitude, 45° Latitude   |
| 4           | 3 | 2 | 90° Longitude, 45° Latitude   |
| 5           | 4 | 2 | 135° Longitude, 45° Latitude  |
| 6           | 5 | 2 | 180° Longitude, 45° Latitude  |
| 7           | 6 | 2 | 225° Longitude, 45° Latitude  |
| 8           | 7 | 2 | 270° Longitude, 45° Latitude  |
| 9           | 8 | 2 | 315° Longitude, 45° Latitude  |
| 10          | 1 | 3 | 0° Longitude, 90° Latitude    |
| 11          | 2 | 3 | 45° Longitude, 90° Latitude   |
| 12          | 3 | 3 | 90° Longitude, 90° Latitude   |
| 13          | 4 | 3 | 135° Longitude, 90° Latitude  |
| 14          | 5 | 3 | 180° Longitude, 90° Latitude  |
| 15          | 6 | 3 | 225° Longitude, 90° Latitude  |
| 16          | 7 | 3 | 270° Longitude, 90° Latitude  |
| 17          | 8 | 3 | 315° Longitude, 90° Latitude  |
| 18          | 1 | 4 | 0° Longitude, 135° Latitude   |
| 19          | 2 | 4 | 45° Longitude, 135° Latitude  |
| 20          | 3 | 4 | 90° Longitude, 135° Latitude  |
| 21          | 4 | 4 | 135° Longitude, 135° Latitude |
| 22          | 5 | 4 | 180° Longitude, 135° Latitude |
| 23          | 6 | 4 | 225° Longitude, 135° Latitude |
| 24          | 7 | 4 | 270° Longitude, 135° Latitude |
| 25          | 8 | 4 | 315° Longitude, 135° Latitude |
| 26          | 1 | 5 | 0° Longitude, 180° Latitude   |

NOTE: Refer to Figure 4 for definition of longitude and latitude.

#### IV. SCENARIO DESCRIPTION

A. This scenario is purely for instructional purposes and is not based on any actual or planned combat attack situation. The target site, order of battle, attack heading, and outbound flight path parameter limits have been chosen only to provide guidelines for the class problem. As much realism has been introduced for the players as possible while retaining an unclassified scenario.

B. Your target is the bridge shown in Fig. 3. located at:

x: 14,100 meters

y: 7,900 meters

z: 20 meters

Heavy military supply traffic has been reported in this area. Your mission is to destroy this vital supply link.

C. The following order of battle has been gathered from intelligence reports of the target area:

Interceptor Aircraft - three airfields within striking distance

SAM - six sites within a 125 km radius.

AAA - two type 1 mode 1  
two type 2 mode 1  
one type 3 mode 4  
one type 3 mode 3  
one type 5 mode 3

(Note: Gun types and their relationship to AAA will be discussed in class.)

Ground Troops - regular infantry and civilian militia are numerous in the target area.

D. The SAM threat and the presence of enemy aircraft requires that the inbound approach to the target be made from the west at low level. A pop-up maneuver is required to visually identify the target followed by a dive bombing run to weapon delivery. Egress must be made to either the north or south, depending on individual strategy.

E. The following is a list of scenario limitations to be used in the development of your strategy:

1. Flight path milestones - specify at least one milestone for approximately 500 meters of flight path.
2. Aircraft cruise speed - 210 to 250 meters per second.
3. Inbound altitude - 70 to 450 meters.
4. Pop-up maneuver
  - a. Commence maneuver - 4,000 to 6,000 meters from the target.
  - b. Maneuver altitude - minimum 1,220 meters; maximum 2,130 meters.
5. Weapons delivery.
  - a. Alignment - the leg immediately prior to the bomb release point must be 600 meters in length (straight) and must have a heading within 5° of the heading to the target from the bomb release point.
  - b. Bomb release range - 1,000 meters maximum.
  - c. Bomb release altitude - 310 to 910 meters.

(Note: A typical 20° dive commenced from 1,000 meters of altitude at 2,500 meters from the target will release weapons at 400 meters of altitude about 700 meters from the target and will lose 160 meters in the pull-out.)

6. Maneuvering - if any turn along the flight path is greater than 28°, the maximum g loading of 6 will be exceeded.

7. Weapons placement.

a. Two type 1 mode 1, two type 2 mode 1, one type 3 mode 4, and one type 3 mode 3 weapons are available for defense placement.

b. One type 5 mode 3 weapon is placed at x: 12,800 meters, y: 7,500 meters, and z: 20 meters. You do not specify the location of this weapon.

c. Neither of the type 3 weapons may be placed within 3,000 meters of the center of the bridge.

8. Jammer power - if the jamming function is specified, the jammer power you select must be no more than 1,000 watts.

F. Begin the flight path at an entry point of your choosing along the western boundary and end it along the northern or southern boundary. Note the terrain features, anticipate the AAA placement for bridge defense and plan your flight path accordingly.

G. Locate the AAA weapons given in the order of battle to best defend the bridge against your opponent's attacking aircraft.



## V. INPUT DATA PREPARATION

A preprocessor for P001 has been developed at NPS that will punch all of the input cards for the execution of P001, with the exception of the green JOB card and the final orange END OF FILE card. This preprocessor is called PIP (P001 Input Program). The inputs to PIP are the x, y and z coordinates of your flight path milestones and your opponent's six AAA emplacement locations.

A. Milestone cards: The x, y and z coordinates of the aircraft (in meters) for up to 199 flight path milestones must be entered into PIP in 3F10.0 format, one milestone per card. (Milestone #1 will have an x coordinate of 0.0).

B. Milestone delimiter card: A card containing 99999., left justified, must be placed after the final milestone card.

C. Option control card: A control card follows the milestone delimiter card and specifies the aircraft cruise speed, the number of the bomb release milestone (count the initial position on the western border as milestone #1), eight input/output/scenario options and the jammer power. The data on the control card must be specified in F10.0,I2,8I1,F10.0 format and contains the following parameters:

(columns 1-10): Aircraft cruise speed in meters per second

(columns 11-12): Number of the bomb release milestone.

- (column 13): EW option - 0 for no jamming, 1 for jamming.
- (column 14): ECCM anti-jam option - 0 for no AJ, 1 for radar AJ.
- (column 15): Radar multipath option - 0 for no multipath effects, 1 for radar degradation caused by multipath effects.
- (column 16): Gun location option - 0 specifies PIP preset AAA locations, 1 requires user input of the six AAA locations.
- (column 17): List option - 0 for no listing of the P001 input deck, 1 for listing of P001 input deck.
- (column 18): Punch option - 0 for no punched P001 input deck, 1 for punched P001 input deck.
- (column 19): Plot option - 0 for no scenario plot, 1 for scenario plot.
- (column 20): Extended output option - 0 for no extended output, 1 for extended printout of P001 analysis results.
- (columns 21-30): Jammer power in watts (0 to 1,000 watts).

D. Gun emplacement location cards: If column 16 on the control card contains a 1, six gun location cards specifying the x, y and z coordinates of each of the gun emplacements (format 3F10.0) specified in the order given in the order of battle are required as input data.

E. Sample PIP input deck:

```

// ( Green JOB Card, TIME=2 )
// EXEC NVTECGO,NAME=PIP,REGION=200K
//STEPLIB DD DSN=F0559.PIP,UNIT=3330,VOL=SER=DISK02,DISP=SHR
//FT06F001 DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//GO.FT07F001 DD SYSOUT=B
//GO.SYSIN DD*
0.          500.          450.
500.         500.          440.
900.         600.          445.
(etc., until all milestones are described)
99999.
232.0        2511111111    500.0
14800.       9000.          20.
16200.       8200.          10.
(etc., until all six AAA locations are described)
/*

```

F. When the P001 input deck punched by PIP is received, put the green JOB card used for PIP on top of the deck and an orange /\* (EOF) card on the bottom of the deck and read it through the card reader. The output of this deck will be a combat survivability analysis for the given flight path and AAA emplacement locations.

# APPENDIX B

## PIP INPUT DECK LISTING

THE FOLLOWING IS AN EXAMPLE OF THE INPUT TO THE POOL INPUT PROGRAM (PIP) TO BE RUN FROM A LOAD MODULE (DISK 02). THE INPUT UTILIZES THE PRESET GUN EMPLOYMENT LOCATION OPTION AND THE ELECTRONIC WARFARE (JAMMING), ANT I-JAM AND MULTIPATH SCENARIO OPTIONS. THE CONTROL CARD ALSO SPECIFIES THAT THE OUTPUT BE LISTED PUNCHED AND PLOTTED. THERE ARE 20 MILESTONES, MILESTONE 15 IS THE BOMB RELEASE MILESTONE AND THE CRUISE SPEED IS SPECIFIED AS 232.0 METERS PER SECOND. THE EXTENDED OUTPUT OPTION IS SPECIFIED.

```
( GREEN JOB CARD )
// EXEC NVTEC GO, NAME=PIP
// STEPLIB DD DSN=F0559.PIP, UNIT=3330, VOL=SER=DISK02, DISP=SHR
// FT06FC01 DD SYSOUT=A, DCB=(RECFM=FBA, LRECL=133, BLKSIZE=3325)
// GO.FTC7F001 DD SYSOUT=B
// GO.SYSIN DD *
0 1386.
2742.
4054.
5406.
6762.
8129.
9508.
10767.
11588.
12054.
12844.
13525.
14050.
14097.
15127.
16479.
17619.
17881.
17872.
99999.
232.0
1511101111 500.0
/*
```



APPENDIX C  
PIP OUTPUT LISTING

```

// EXEC PGM=IEH,REGION=200K,VOL=SER=DISK02,DISP=SHR
//STEPLIB DD DSN=F0559.PIEM,UNIT=3330,LRECL=133,BLKSIZE=3325)
//FT06F001 DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//GO.FT04F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),VOL=SER=DISK02,DISP=SHR
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT07F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),VOL=SER=DISK02,DISP=SHR
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT08F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),VOL=SER=DISK02,DISP=SHR
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT09F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),VOL=SER=DISK02,DISP=SHR
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//FT11F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),VOL=SER=DISK02,DISP=SHR
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT05F001 DD *

```

## 01 AIRCRAFT COMBAT SURVIVABILITY SCENARIO

[illegible]

GC.FTC5F002 DD \* 9000. 40.  
03 14800.

|    |            |        |       |
|----|------------|--------|-------|
| 03 | 1211111111 | 34411  | VULNE |
| 03 | 1211111111 | 12.543 |       |
| 04 | 21111      | 9.853  |       |
| 03 | 1211111111 | 5.653  |       |
| 03 | 1211111111 | 12.640 |       |
| 03 | 1211111111 | 9.853  |       |
| 04 | 21111      | 5.633  |       |

[illegible]



[illegible]



1211111

GO.FTC5F003 DD \*

•

(NEW PAGE)

\*\*\*\*\*  
 POOL FLIGHT PATH SCENARIO SUMMARY  
 \*\*\*\*\*

THE FLIGHT PATH CONSISTS OF 20 MILESTONES WITH A TOTAL FLIGHT TIME OF 1114.5 SECONDS. BOMB RELEASE IS AT MILESTONE 15.

\*\*\*\*\*  
OPTION SUMMARY

CRUISE SPEED IS 232.0 METERS PER SECOND.  
A POOL INPUT LISTING IS PROVIDED AS OUTPUT.  
A SCENARIO PLOT IS PROVIDED AS OUTPUT.  
EXTENDED PRINTOUT IS PROVIDED AS OUTPUT.  
PRESET GUN EMPLOYMENT IS PROVIDED AS OUTPUT.  
AN AIRBORNE JAMMER IS BEING UTILIZED.  
JAMMER POWER IS 500.0 WATTS.  
JAM TYPATH RADAR EFFECTS ARE CONSIDERED WHERE APPROPRIATE.

## FLIGHT PATH ERRORS

```
***** NO FLIGHT PATH ERRORS IN THIS RUN *****
```

\*\*\*\*\*  
END OF POOL INPUT PROGRAM - SUMMARY COMPLETE  
\*\*\*\*\*

## APPENDIX D

### P001 INPUT GUIDE CHANGES

Revisions to P001:

1. Delete card 7A and 7B.
2. Insert cards 13 and 14 as given on the following pages.
3. The following list is given as an aid to facilitate the assignment of valid combinations of gun type, mode and operating characteristics to the ECM and multipath options:

| Gun Type<br>(IGT) | Mode<br>(IEM) | Radar ID<br>(IRECM) | IRMP | Option                   |
|-------------------|---------------|---------------------|------|--------------------------|
| 1                 | 1             | -                   | -    | -                        |
| 2                 | 1             | -                   | -    | -                        |
| 3                 | 1             | -                   | -    | -                        |
| 3                 | 2             | -                   | -    | -                        |
| 3                 | 3             | -                   | -    | -                        |
| 3                 | 3             | 1                   | -    | Jam                      |
| 3                 | 3             | 1                   | 1    | Jam, Multipath           |
| 3                 | 4             | -                   | -    | -                        |
| 3                 | 4             | 1                   | -    | Jam                      |
| 4                 | 1             | -                   | -    | -                        |
| 5                 | 1             | -                   | -    | -                        |
| 5                 | 2             | -                   | -    | -                        |
| 5                 | 3             | -                   | -    | -                        |
| 5                 | 3             | 2                   | -    | Jam                      |
| 5                 | 3             | 2                   | 2    | Jam, Multipath           |
| 5                 | 3             | 3                   | -    | Jam, Anti-jam            |
| 5                 | 3             | 3                   | 2    | Jam, Anti-jam, Multipath |
| 5                 | 3             | 4                   | -    | Jam                      |
| 5                 | 3             | 4                   | 3    | Jam, Multipath           |
| 5                 | 4             | -                   | -    | -                        |
| 5                 | 4             | 2                   | -    | Jam                      |
| 5                 | 4             | 3                   | -    | Jam, Anti-jam            |
| 5                 | 4             | 4                   | -    | Jam                      |

| Radar Multipath Input Parameters   |                    |                       |  |         | CARD: 13   |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
|--|--------------------|-----------------------|--|---------|--|-----------------------|-------------------------|---|-----|-----|-------|---|-----|-----|------|---|-----|-----|------|--|--|
| ID   | PARA               | UNITS                 | FORMAT   | COLUMNS | DESCRIPTION  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| A  | I                  | ND                    | I2   | 1-2     | Data group identification code. I=13 indicates that the remainder of the card contains radar multipath parameters. |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| B  | ICARD              | ND                    | 7A10,A8  | 3-80    | Seventy-eight columns of alphanumeric data to be decoded and assigned as follows:                                  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| B1   | IMUL               | ND                    | I3   | 3-5     | IMUL=0, no multipath. Turn off multipath if previously used. IMUL=1, multipath desired.                            |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| B2   | IRMP               | ND                    | I5   | 6-10    | Radar type ID. Specifies the tracking radar.   |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| B3   | REFC               | ND                    | F10.0  | 11-20   | Reflection coefficient. 0.35 is a typical value for terrain with vegetation.                                       |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| NOTE: Multipath effects can only be applied to a system with a Mode ID (IEM) of 3. |                    |                       |  |         |  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| The value of IRMP selects the appropriate radar parameters:                        |                    |                       |  |         |  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
|  |                    |                       | <table><tr><th>IRMP</th><th>Beamwidth<br/>(deg)</th><th>Squint<br/>Angle (deg)</th><th>Calibration<br/>Constant</th></tr><tr><td>1</td><td>1.4</td><td>0.5</td><td>0.759</td></tr><tr><td>2</td><td>1.8</td><td>0.6</td><td>1.06</td></tr><tr><td>3</td><td>4.5</td><td>1.4</td><td>2.74</td></tr></table> | IRMP    | Beamwidth<br>(deg)   | Squint<br>Angle (deg) | Calibration<br>Constant | 1 | 1.4 | 0.5 | 0.759 | 2 | 1.8 | 0.6 | 1.06 | 3 | 4.5 | 1.4 | 2.74 |  |  |
| IRMP   | Beamwidth<br>(deg) | Squint<br>Angle (deg) | Calibration<br>Constant  |         |  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| 1  | 1.4                | 0.5                   | 0.759  |         |  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| 2  | 1.8                | 0.6                   | 1.06   |         |  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |
| 3  | 4.5                | 1.4                   | 2.74   |         |  |                       |                         |   |     |     |       |   |     |     |      |   |     |     |      |  |  |

CARD: 13

CARD: 13

| Radar Multipath Input Parameters   |                 |             |        |         | CARD: 13    |  |                 |             |  |                |  |   |   |  |     |   |  |   |   |  |
|--|-----------------|-------------|--------|---------|-------------|--|-----------------|-------------|--|----------------|--|---|---|--|-----|---|--|---|---|--|
| ID   | PARA            | UNITS       | FORMAT | COLUMNS | DESCRIPTION |  |                 |             |  |                |  |   |   |  |     |   |  |   |   |  |
| <p>The following table gives the relationship between the Radar ID (IRECM) and the corresponding radar parameters (IRMP):</p> <table border="0"> <tr> <td></td> <td><u>Radar ID</u></td> <td><u>IRMP</u></td> </tr> <tr> <td></td> <td><u>(IRECM)</u></td> <td></td> </tr> <tr> <td>1</td> <td>1</td> <td></td> </tr> <tr> <td>2,3</td> <td>2</td> <td></td> </tr> <tr> <td>4</td> <td>3</td> <td></td> </tr> </table> |                 |             |        |         |             |  | <u>Radar ID</u> | <u>IRMP</u> |  | <u>(IRECM)</u> |  | 1 | 1 |  | 2,3 | 2 |  | 4 | 3 |  |
|  | <u>Radar ID</u> | <u>IRMP</u> |        |         |             |  |                 |             |  |                |  |   |   |  |     |   |  |   |   |  |
|  | <u>(IRECM)</u>  |             |        |         |             |  |                 |             |  |                |  |   |   |  |     |   |  |   |   |  |
| 1  | 1               |             |        |         |             |  |                 |             |  |                |  |   |   |  |     |   |  |   |   |  |
| 2,3  | 2               |             |        |         |             |  |                 |             |  |                |  |   |   |  |     |   |  |   |   |  |
| 4  | 3               |             |        |         |             |  |                 |             |  |                |  |   |   |  |     |   |  |   |   |  |

CARD: 13



| ECM (Jamming) Input Parameters |       |                |         |         | CARD: 14  |          |
|--------------------------------|-------|----------------|---------|---------|---|----------|
| ID                             | PARA  | UNITS          | FORMAT  | COLUMNS | DESCRIPTION   |          |
| A                              | I     | ND             | I2      | 1-2     | Data group identification code. I=14 indicates that the remainder of the card contains ECM parameters.  |          |
| B                              | ICARD | ND             | 7A10,A8 | 3-80    | Seventy-eight columns of alphanumeric data to be decoded and assigned as follows:   |          |
| B1                             | IJAM  | ND             | I3      | 3-5     | Jamming switch. 0=no jamming; 1=jamming; card must be reread with IJAM=0 to turn jamming off.   |          |
| B2                             | IP    | ND             | I3      | 6-8     | IP=0, no J/S printout. IP=n, print every nth value.   |          |
| B3                             | IJ    | ND             | I2      | 9-10    | IJ=0; then GAINJ is antenna gain of jammer.   |          |
| B4                             | GAINJ | ND             | F10.0   | 11-20   | IJ=1; a 37x37 5° jammer table follows. Gain is in dB.   |          |
| B5                             | PJW   | watts          | F10.0   | 21-30   | Jammer power in watts.  |          |
| B6                             | PLEN  | sec            | F10.0   | 31-40   | Length of jammer cover pulse. Needed when IRECM=3. 1 microsecond is a standard value.   |          |
| B7                             | IX    | ND             | I5      | 41-45   | IX=0; no cross section table is needed.   |          |
| B8                             | XSEC  | m <sup>2</sup> | F10.0   | 46-55   | A constant cross section of XSEC m <sup>2</sup> is used.  |          |
| B9                             | CALX  | ND             | F10.0   | 56-65   | IX=1, a cross section table will be read following the 14 card and a possible jammer table. The cross section table values will be multiplied by CALX. This allows the user to scale the cross section. If not used, it must be set to 1. | CARD: 14 |

| ECM (Jamming) Input Parameters |              |               |                  |            | CARD: 14   |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
|--------------------------------|--------------|---------------|------------------|------------|--|------------------|--------------|---------------|------------------|---------|-----|---------|---------|--------|------|-----|---------|---------|--------|------------|---|---------|---------|---------|------|
| ID                             | PARA         | UNITS         | FORMAT           | COLUMNS    | DESCRIPTION  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| B10                            | IRECM        | ND            | I5               | 66-70      | <p>Radar type. This index calls up the following constants from a data statement:</p> <table border="1"> <thead> <tr> <th>IRECM</th> <th>Gain (RGDB)</th> <th>Power (PRW)</th> <th>Frequency (FREQ)</th> <th>SJTMAX*</th> </tr> </thead> <tbody> <tr> <td>1</td> <td>40.0 dB</td> <td>150,000</td> <td>15.1E9</td> <td>3 dB</td> </tr> <tr> <td>2,3</td> <td>38.5 dB</td> <td>175,000</td> <td>9.38E9</td> <td>1.5, 17 dB</td> </tr> <tr> <td>4</td> <td>28.0 dB</td> <td>250,000</td> <td>2.838E9</td> <td>0 dB</td> </tr> </tbody> </table> <p>*SJTMAX must be entered as in B11.<br/>Threshold where tracking errors become significant. (dB)</p> | IRECM            | Gain (RGDB)  | Power (PRW)   | Frequency (FREQ) | SJTMAX* | 1   | 40.0 dB | 150,000 | 15.1E9 | 3 dB | 2,3 | 38.5 dB | 175,000 | 9.38E9 | 1.5, 17 dB | 4 | 28.0 dB | 250,000 | 2.838E9 | 0 dB |
| IRECM                          | Gain (RGDB)  | Power (PRW)   | Frequency (FREQ) | SJTMAX*    |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| 1                              | 40.0 dB      | 150,000       | 15.1E9           | 3 dB       |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| 2,3                            | 38.5 dB      | 175,000       | 9.38E9           | 1.5, 17 dB |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| 4                              | 28.0 dB      | 250,000       | 2.838E9          | 0 dB       |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| B11                            | SJTMAX       | dB            | F10.0            | 71-80      | <p>NOTE: The following table summarizes the valid combinations of radars, gun types and tracking modes:</p> <table border="1"> <thead> <tr> <th>Radar ID (IRECM)</th> <th>Gun ID (IGT)</th> <th>Mode ID (IEM)</th> </tr> </thead> <tbody> <tr> <td>1</td> <td>3</td> <td>3,4</td> </tr> <tr> <td>2,3</td> <td>5</td> <td>3,4</td> </tr> <tr> <td>4</td> <td>5</td> <td>3,4</td> </tr> </tbody> </table> <p>The two ID's on the second radar indicate the anti-jam capability.<br/>IRECM = 2, anti-jam off<br/>IRECM = 3, anti-jam on</p>   | Radar ID (IRECM) | Gun ID (IGT) | Mode ID (IEM) | 1                | 3       | 3,4 | 2,3     | 5       | 3,4    | 4    | 5   | 3,4     |         |        |            |   |         |         |         |      |
| Radar ID (IRECM)               | Gun ID (IGT) | Mode ID (IEM) |                  |            |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| 1                              | 3            | 3,4           |                  |            |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| 2,3                            | 5            | 3,4           |                  |            |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |
| 4                              | 5            | 3,4           |                  |            |  |                  |              |               |                  |         |     |         |         |        |      |     |         |         |        |            |   |         |         |         |      |

CARD: 14

APPENDIX E

JCL CARD SETUPS FOR REFERENCES

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE POOL INPUT PROGRAM (PIP)  
FROM BATCH, UTILIZING THE CALCOMP PLOTTER:

```
// ( GREEN JOB CARD )  
// EXEC FORTCLGP  
// FORTCO1 DD SYSCUT=B  
// FORT.SYSIN DD *  
// ( PROGRAM SOURCE CARDS GO HERE )  
/*  
// GO.SYSIN DD *  
// ( DATA DECK GOES HERE )  
/*
```

THIS PAGE IS BEST QUALITY PRACTICABLE  
FROM COPY FURNISHED TO DDG

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE POOL INPUT PROGRAM (PIP)  
FROM BATCH, UTILIZING THE VERSATEC PLOTTER:

```
// ( GREEN JOB CARD )  
// EXEC FORTCLGV  
//FCRT.SYSIN DD *  
// ( PROGRAM SOURCE CARDS GO HERE )  
/*  
//GC.FTCTF001 DD SYSOUT=8  
//GO.SYSIN DD *  
// ( DATA DECK GOES HERE )  
/*
```



THE FOLLOWING CARDS ARE THE SETUP TO CREATE A LOAD MODULE FOR THE PC01  
INPUT PROGRAM (PIP):

```
// ( GREEN JOB CARD )
// EXEC NVTECLNK
// FCRT.SYSIN DD *
// ( PROGRAM SOURCE CARDS GO HERE )
/*
//LINK.SYSLMOD DD DSN=F0559.PIP,SPACE=(CYL,(2,1,1)),
// UNIT=3330,VOL=SER=DISK02,DISP=(NEW,KEEP),
// LABEL=EXPDT=99360
//LINK.SYSIN DD *
ENTRY MAIN
NAME PIP(R)
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE POOL INPUT PROGRAM (PIP)  
FROM A LOAD MODULE, UTILIZING THE VERSATEC PLOTTER.

```
// ( GREEN JOB CARD )
// EXEC NVTECGO,NAME=PIP,REGION=200K
//STEPLIB DD DSN=F0559.PIP,UNIT=3330,VOL=SER=DISK02,DISP=SHR
//FT06F001 DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//GO.FT07F001 DD SYSOUT=B
//GO.SYSIN DD *
// ( DATA DECK GOES HERE )
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO REMOVE THE PC01 INPUT PROGRAM  
(PIP) FROM DISK 02:

```
/// JOB CARD )  
//SCRATCH EXEC PGM=IEHPRGM  
//SYSPRINT DD SYSCUT=A  
//DD1 DD UNIT=3330,VOL=SER=DISK02,DISP=CLD  
//SYSIN DD *  
SCRATCH DSN= F0559.PIP,VOL=3330=DISK02,PURGE  
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE IMPRCVD P001 AAA SIMULATION  
PROGRAM (PLEW) FRM BATCH:

```
// ( GREEN JOB CARD )
// EXEC FORTCLG,REGICN.GO=250K
// FORT.SYSIN DD *
// ( PROGRAM SOURCE CARDS GO HERE )
/*
// GO.FTC4F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC7F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC8F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC9F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FT11F001 DD UNIT=SYSDA, SPACE=(CYL,(1,1)),
// DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
// GO.FTC5F001 DD *
// ( DATA DECK GOES HERE )
/*
```



THE FOLLOWING CARDS ARE THE SETUP TO CREATE A LOAD MODULE FOR THE IMPROVED POOL AAA SIMULATION PROGRAM (PIEW):

```

// ( GREEN JOB CARD )
// EXEC FORTCL
//FCRT.SYSIN DD *
// ( PROGRAM SOURCE CARDS GO HERE )
/*
//LINK.SYSLMOD DD DSN=F0559.PIW,SPACE=(CYL,(2,1,1)),
// UNIT=3330,VOL=SER=DISK02,DISP=(NEW,KEEP),
// LABEL=EXPDT=99360
//LINK.SYSIN DD *
ENTRY MAIN
NAME PIEW(R)
/*

```

THE FOLLOWING CARDS ARE THE SETUP TO RUN THE IMPRVED POOL AAA SIMULATION  
PROGRAM (PIEW) FROM A LOAD MODULE:

```
// ( GREEN JOB CARD )
// EXEC PGM=PIEW,REGION=200K
//STEPLIB DD DSN=FO559.PIW,UNIT=3330,VCL=SER=DISK02,DISP=SHR
//FT06FC01 DD SYSCUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//GO.FT04F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT07F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT08F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT09F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GO.FT11F001 DD UNIT=SYSDA,SPACE=(CYL,(1,1)),
//DCB=(RECFM=VBS,LRECL=404,BLKSIZE=3236)
//GC.FT05F001 DD
// ( DATA DECK GOES HERE )
/*
```

THE FOLLOWING CARDS ARE THE SETUP TO REMOVE PLEW FROM DISK 02:

```
/// JOB CARD )  
/// SCRATCH EXEC PGM=IEHPRGM  
/// SYSPRINT DD SYSCUT=A  
/// DD1 DD UNIT=3330,VOL=SER=DISK02,DISP=CLD  
/// SYSPRINT DD *  
/// SCRATCH DSN=DSNAME=F0559.PLEW,VOL=3330=DISK02,PURGE  
/*
```

APPENDIX F

P001 ANALYSIS OUTPUT

MASK ANGLE FOR THIS RUN = 0.0 DEG.



[illegible]

PAR, F ?

AFAIL P-001 AAASIM " AIRCRAFT CCMUAT SURVIVABILITY SCENARIO

[illegible]

|     |     |     |     |     |     |     |     |     |     |     |     |
|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|
| XR= | 0.0 | YR= | 0.0 | XL= | 0.0 | YL= | 0.0 | PL= | 0.0 | PL= | 0.0 |
|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|





### 10 TIME INTERVALS FOR PK ACCUMULATION

| VIEW | 0 | 152 | 305 | 457 | 610 | 762 | 914 | 1067 | 1219 |
|------|---|-----|-----|-----|-----|-----|-----|------|------|
| VIEW | 0 | 152 | 305 | 457 | 610 | 762 | 914 | 1067 | 1219 |

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PAGE 5

AFATL P-001 AAASIM = AIRCRAFT COMBAT SURVIVABILITY SCENARIO

ECM INPUTS (INITIAL OR CHANGED)

IP = 5  
IJ = 0  
GAINJ(OBJ) = 1.00  
PJM(J) = 500.00  
PLEN(S) = 0.0  
IX = 1  
JSEC(SOM) = 0.0  
JALC = 1.000  
JTXMAX(08) = 3.00

JAMMER ANTENNA GAIN 1.000 DB

AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX = 1.00

TABLE DATA

19 ELEMENTS FROM 8.0 IE 189.00 EV 19.00 AL  
 7 ELEMENTS FROM 50.00 IE 1000.00  
 ELSEWHERE TABLE 15

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[illegible]



VULNERABLE AREA (SQ. METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SFC) AND ASPECT VIEW VULNERABLE AREA TABLE VS TYPE 2 WEAPONS

[illegible]

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AFATL P-001 AAASIM " AIRCRAFT-CHEAT SURVIVABILITY-SCENARIO

PAGE 7

MULTIPATH INPUTS (INITIAL CR-CHANGED

ISMP =  
REFC = 0.350

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## MULTIPATH INPUTS (INITIAL CR CHANGED)

1AMP = 2  
REFC = 0.350

## ECM INPUTS (INITIAL OR CHANGED)

IP = 5  
 IJAJIN(JOB) = 1.00  
 IJAJIN(M) = 500.00  
 IJAJIN(S) = 0.0  
 IJAJIN(XSEC(SCM) = 0.0  
 IJAJIN(CALX) = 1.00  
 IJAJIN(MAX(OB) = 17.00

JAMMER ANTENNA GAIN 1,000 DB.

~~AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALC.CATX= 1.00~~

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THIS PAGE IS BEST QUALITY PRACTICABLE  
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TABLE DATA

19 ELEMENTS FROM  
17 ELEMENTS FROM  
ELSEWHERE TABLE IS

0-0 16 180-00 14 10-00 47  
-50-00 16 90-00 14 30-00 47  
1000-00



AD-A057 907

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF  
ADAPTATION OF THE IMPROVED ANTI-AIRCRAFT ARTILLERY SIMULATION CO--ETC(U)  
MAR 78 C F SWENSON

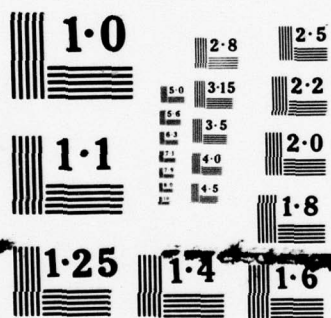
F/G 19/5  
CO--ETC(U)

UNCLASSIFIED

NL

2 OF 3  
ADA  
057907





NATIONAL BUREAU OF STANDARDS  
MICROCOPY RESOLUTION TEST CHART

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[illegible]

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| KEY   | 130    | 140    | 150    | 160    | 170    | 180    |
|-------|--------|--------|--------|--------|--------|--------|
| 120.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 130.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 140.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 150.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 160.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 170.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 180.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |





AFATL P-001 AAASIM " AIRCRAFT-COMBAT-SURVIVABILITY SCENARIO

COLLECT AREA (SQ. METERS) AS A FUNCTION OF  
COLLECTIBLE AREA TABLE VS TYPE 5 WEAPON

[illegible]

PAGE 10

AFATL P-001 AASIM - AIRCRAFT-COMBAT SURVIVABILITY SCENARIO

| LOG | P KILL 1  | RCNDS | FIRE TIME | XGUN     | YGUN     | ZGUN  | RADIUS GL | T REACT | T TRACK | T CH | T IN |
|-----|-----------|-------|-----------|----------|----------|-------|-----------|---------|---------|------|------|
| 1   | 0.000000  | 17    | 2.25      | 14800.00 | 9000.00  | 40.00 | 0.0       | 0.0     | 0.0     | 0.0  | 0.0  |
| 2   | 0.001094  | 17    | 2.25      | 14200.00 | 8700.00  | 40.00 | 0.0       | 0.0     | 0.0     | 0.0  | 0.0  |
| 3   | 0.0003691 | 17    | 6.80      | 13400.00 | 7200.00  | 20.00 | 0.0       | 0.0     | 0.0     | 0.0  | 0.0  |
| 4   | 0.0       | 42    | 0.0       | 11400.00 | 6000.00  | 20.00 | 0.0       | 0.0     | 0.0     | 0.0  | 0.0  |
| 5   | 0.01235   | 132   | 21.42     | 11300.00 | 1000.00  | 20.00 | 0.0       | 0.0     | 0.0     | 0.0  | 0.0  |
| 7   | 0.0884375 | 132   | 31.42     | 12800.00 | 17500.00 | 20.00 | 0.0       | 0.0     | 0.0     | 0.0  | 0.0  |

ATTRITION ACCRUED AS A FUNCTION OF TIME OF FIRE

| TIME<br>SEG | CUM FOR<br>CLASS 1 | DENSITY<br>CLASS 1 | DENSITY<br>CLASS 2 | DENSITY<br>CLASS 3 | DENSITY<br>CLASS 4 | DENSITY<br>CLASS 5 | DENSITY<br>CLASS 6 | DENSITY<br>CLASS 7 | DENSITY<br>CLASS 8 | DENSITY<br>CLASS 9 |
|-------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|
| 1           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 2           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 3           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 4           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 5           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 7           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 8           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 9           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 10          | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |

ATTRITION ACCRUED AS A FUNCTION OF TIME AT INTERCEPT

| TIME<br>SEG | CUM FOR<br>CLASS 1 | DENSITY<br>CLASS 1 | DENSITY<br>CLASS 2 | DENSITY<br>CLASS 3 | DENSITY<br>CLASS 4 | DENSITY<br>CLASS 5 | DENSITY<br>CLASS 6 | DENSITY<br>CLASS 7 | DENSITY<br>CLASS 8 | DENSITY<br>CLASS 9 |
|-------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|
| 1           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 2           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 3           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 4           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 5           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 7           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 8           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 9           | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |
| 10          | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                | 0.0                |

TOTALS 0.0802243

TOTAL PK FOR DENSITY CLASS 1 AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.

| SECTION | AZIMUTH<br>BEAR=00 | ELEV<br>DOWN=00 | 0-152 | 152-305 | 305-457 | 457-610 | 610-762   | 762-914 | 914-1067 | 1067-1219 | TOTAL PK  |
|---------|--------------------|-----------------|-------|---------|---------|---------|-----------|---------|----------|-----------|-----------|
| 1       | 000-045            | 000-045         | 0.0   | 0.0     | 0.0     | 0.0     | 0.0       | 0.0     | 0.0      | 0.0       | 0.0       |
| 2       | 045-090            | 000-045         | 0.0   | 0.0     | 0.0     | 0.0     | 0.0       | 0.0     | 0.0      | 0.0       | 0.0       |
| 3       | 090-135            | 000-045         | 0.0   | 0.0     | 0.0     | 0.0     | 0.0001340 | 0.0     | 0.0      | 0.0       | 0.0001340 |

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AFATL P-001 AAASIM " AIRCRAFT-CCPBAT SURVIVABILITY SCENARIO

| TOTAL PK FOR DENSITY CLASS 1 AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED. |           |           |           |           |           |           |           |           |           |  |  |
|---|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|-----------|--|--|
| SECTOR  | 0-152     | 152-305   | 305-457   | 457-610   | 610-762   | 762-914   | 914-1067  | 1067-1219 | TOTAL PK  |  |  |
| 0-152   | 0.0003255 | 0.0012023 | 0.0002515 | 0.0002515 | 0.0002515 | 0.0002515 | 0.0002515 | 0.0002515 | 0.0002515 |  |  |
| 152-305   | 0.0004111 | 0.0004111 | 0.0004111 | 0.0004111 | 0.0004111 | 0.0004111 | 0.0004111 | 0.0004111 | 0.0004111 |  |  |
| 305-457   | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 |  |  |
| 457-610   | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 |  |  |
| 610-762   | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 |  |  |
| 762-914   | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 |  |  |
| 914-1067  | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 |  |  |
| 1067-1219   | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 | 0.0003374 |  |  |
| TOTALS  | 0.0001108 | 0.0047761 | 0.0075947 | 0.0079440 | 0.0180647 | 0.0435568 | 0.0       | 0.0       | 0.0902164 |  |  |

\*\*\*\* END OF JCB - - PC01 SCENARIO RUN COMPLETE \*\*\*\*



APPENDIX G

P001 ANALYSIS EXTENDED OUTPUT

MASK ANGLE FOR THIS RUN = 0.0 DEG.



PAGE --- 1 ---

~~AFAIL P-001-AAASIM " AIRCRAFT-COMBAT-SURVIVABILITY-SCENARIO~~

**AND**

CITYMA

## WEADYAG

20

YDOT

INDEX

**DEFD**

222

yy

XXX

TIME

WILEY-INTERSCIENCE

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PAGE 2

AFATL P-001 AASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

| TIME   | XXX   | YYY   | ZZZ   | SPEED  | XOCT  | YDOT  | ZDOT  | HEADING | CLIMB | ROLL  |
|--------|-------|-------|-------|--------|-------|-------|-------|---------|-------|-------|
| 30.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 31.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 32.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 33.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 34.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 35.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 36.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 37.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 38.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 39.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 40.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 41.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 42.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 43.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 44.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 45.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 46.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 47.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 48.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 49.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 50.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 51.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 52.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 53.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 54.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 55.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 56.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 57.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 58.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 59.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |
| 60.000 | 5.000 | 7.000 | 1.000 | 30.000 | 0.000 | 0.000 | 0.000 | 0.000   | 0.000 | 0.000 |

XR= 0.0 YR= 0.0 ZR= 0.0 PS= 0.0 DT= 0.0



**PAGE 3**

~~AFATL P-001 AAASIN - AIRCRAFT CCMBAT-SURVIVABILITY SCENARIO~~

[illegible]

~~AFATL P-001 AAASIN " AIRCRAFT COMBAT SURVIVABILITY SCENARIO -~~

### 10 TIME INTERVALS FOR PK ACCUMULATION

|      |       |       |       |       |       |       |       |       |        |        |
|------|-------|-------|-------|-------|-------|-------|-------|-------|--------|--------|
| 0.00 | 11.47 | 22.95 | 34.42 | 45.89 | 57.37 | 68.84 | 80.31 | 91.78 | 103.26 | 599.99 |
|------|-------|-------|-------|-------|-------|-------|-------|-------|--------|--------|

UNVULNERABLE AREA (SQ METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW  
VULNERABLE AREA (TABLE VS TYPE 1 AND 2 WEAPONS)

[illegible]





**PAGE 6**

AFAIL P-001 AAASIN - AIRCRAFT - COMBAT - SURVIVABILITY - SCENARIO -

[illegible]

PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED. LOC 2 GT 1 EM 1 X 162CO. Y 82CC. Z 40. RADIUS 0.0

[illegible]

|        | 0.0 | 0.0 | 0.0 | 0.0 | 0.0       |
|--------|-----|-----|-----|-----|-----------|
| TOTALS | 0.0 | 0.0 | 0.0 | 0.0 | 0.0010945 |





| PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED. |          | LOC   | 3       | GT 2    | EM 1     | X 13600.  | Y 7200. | Z        | 20.       | RADIUS    | O.M |
|---|----------|-------|---------|---------|----------|-----------|---------|----------|-----------|-----------|-----|
| SECTOR  | ALTIMUTH | 0-152 | 152-305 | 305-457 | 457-610  | 610-762   | 762-914 | 914-1067 | 1067-1219 | TOTAL PK  |     |
| 28  | 180-225  | 0.0   | 0.0     | 0.0     | 0.0      | 0.0       | 0.0     | 0.0      | 0.0       | 0.0       |     |
| 29  | 225-270  | 0.0   | 0.0     | 0.0     | 0.0      | 0.0       | 0.0     | 0.0      | 0.0       | 0.0       |     |
| 30  | 270-315  | 0.0   | 0.0     | 0.0     | 0.0      | 0.0       | 0.0     | 0.0      | 0.0       | 0.0       |     |
| 31  | 315-360  | 0.0   | 0.0     | 0.0     | 0.0      | 0.0       | 0.0     | 0.0      | 0.0       | 0.0       |     |
| TOTALS  |          | 0.0   | 0.0     | 0.0     | 0.000267 | 0.0003423 | 0.0     | 0.0      | 0.0       | 0.0003690 |     |

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AFATL P-001 AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

ECM INPUTS (INITIAL OR CHANGED)

IP = 5  
 IJ = 0  
 GAIN(JDB) = 1.00  
 PJW(L) = 500.00  
 PLEN(S) = 0.0  
 IX = 1  
 SPEC(SOM) = 0.0  
 CALX = 1.00  
 SJTRAX(DB) = 1.00

JAMMER ANTENNA GAIN 1.000 DB

AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX= 1.00

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TABLE DATA

19 ELEMENTS FROM 0.0 10 100.00 BY 10.00 AZ  
2 ELEMENTS FROM 100.00 10 100.00 BY 10.00 AZ  
ELSEWHERE TABLE IS 1000.00



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| ECN   | MATRIX | 130    | 140    | 150    | 160    | 170    | 180    |
|-------|--------|--------|--------|--------|--------|--------|--------|
| 120.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 130.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 140.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 150.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 160.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 170.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |
| 180.0 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 | 100.00 |

VULNERABLE AREA (SQ. METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW  
VULNERABLE AREA TABLE VS TYPE 3 WEAPONS

[illegible]

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**PAGE 12**

~~AFATL P-001 AAASIM # AIRCRAFT CCMBAT SURVIVABILITY SCENARIO~~

|                    |              |              |               |                |              |              |               |              |               |                      |              |                |
|--------------------|--------------|--------------|---------------|----------------|--------------|--------------|---------------|--------------|---------------|----------------------|--------------|----------------|
| LOCATION           | 5            | GUN TYPE     | 3             | ERRR MODE      | 4            | POSITION=(   | 11300.0,      | 9700.0,      | 50.0)         | RADIUS=              | 0.C M        |                |
| FIRE<br>L OEN TIME | FLY2<br>TIME | INCP<br>TIME | FIRE<br>RANGE | INCIP<br>RANGE | SIG1<br>SIG2 | BIA1<br>BIA2 | CLOSE<br>VEL. | AZIM<br>RATE | ELEV.<br>RATE | MEAN<br>AZERR ELSEPR | VULM<br>AREA | SHOT PK CUM.PK |

**PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.**

| LOC | 5 | GT 3 | EM 4 | X 113CG. | Y 57CG. | Z 50. | RADIUS 0.6M |
|-----|---|------|------|----------|---------|-------|-------------|
|     |   |      |      |          |         |       |             |

[illegible]



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PAGE 13

AFATL P-001 AAASLM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

MULTIPATH INPUTS INITIAL OR CHANGED

IRMP = 1  
RESC = 0.350



AFATL P-001 AAASIN# AIRCRAFT CCMBAT-SURVIVABILITY-SCENARIO-

[illegible]

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**PAGE 15**

AFATL P-001 AAASIM-4 AIRCRAFT COMBAT SURVIVABILITY SCENARIO-

[illegible]



**PAGE 16**

~~AFATL P-001 AAASIN # AIRCRAFT-CCP0AT-SURVIVABILITY-SCENARIO-~~

[illegible]



**PAGE 17**

AFATL P-001 AAASIM - AIRCRAFT-COMBAT-SURVIVABILITY-SCENARIO-

[illegible]

**PAGE 10**

AFATL P-001-AAASIN " AIKRAFT COMBAT SURVIVABILITY SCENARIO--

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~~AFATL P-001 AAASIN " AIRCRAFT COMBAT SURVIVABILITY SCENARIO~~

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**PAGE 21**

~~AFAIL P-001 AAASIN " AIRCRAFT COMBAT SURVIVABILITY SCENARIO~~

| LOCATION | 6 | GUN TYPE | 3 | INCR RANGE | INCR TIME | FREE RANGE | FREE TIME | INCR RANGE | INCR TIME | ERRC MODE | 3 | SIG1 | SIG2 | BIAS1 | BIAS2 | CLOSE VLE | ATIME | ATIME RATE | ELV RATE | AZERR | MEAN ELERR | BEAN | WUM AREA | SHOT PK | CUP PK |
|----------|---|----------|---|------------|-----------|------------|-----------|------------|-----------|-----------|---|------|------|-------|-------|-----------|-------|------------|----------|-------|------------|------|----------|---------|--------|
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |
|          |   |          |   |            |           |            |           |            |           |           |   |      |      |       |       |           |       |            |          |       |            |      |          |         |        |



~~AFATL P-001 AAASIM - AIRCRAFT-COMBAT-SURVIVABILITY-SCENARIO~~

[illegible]



| PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED. |         | LOC     | 6     | GT 3      | EM 3      | X 15600. | Y 15600. | Z       | 90.      | RADIUS    | 0.0       |
|---|---------|---------|-------|-----------|-----------|----------|----------|---------|----------|-----------|-----------|
| SECTOR  | AZIMUTH | ELEV    | 0-152 | 152-305   | 305-457   | 457-610  | 610-762  | 762-914 | 914-1067 | 1067-1219 | TOTAL PK  |
| 32  | 315-360 | 135-180 | 0.0   | 0.0       | 0.0       | 0.0      | 0.0      | 0.0     | 0.0      | 0.0       | 0.0       |
| TOTALS  |         |         | 0.0   | 0.0039258 | 0.0073024 | 0.0      | 0.0      | 0.0     | 0.0      | 0.0       | 0.0111996 |

## MULTIPATH INPUTS INITIAL OR CHANGED

IRMP = 2  
REFC = 0.350

## ECM INPUTS (INITIAL OR CHANGED)

IP = 5  
IJ = 0  
GAINJ(08) = 1.00  
P(08) = 300.00  
PL(08) = 0.0  
XSEC(SOW) = 1.0  
CALX = 1.000  
TREC = 1.000  
SUTRAX(08) = 17.00

JAMMER ANTENNA GAIN 1.000 DB

AIRCRAFT CROSS SECTION TABLE SPECIFIED. PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX= 1.00

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TABLE DATA  
19 ELEMENTS FROM 0-0 TO 180-00 BY 30-00 AZ  
7 ELEMENTS FROM 180-00 TO 30-00 BY 30-00 AZ  
ELSEWHERE TABLE IS 1000.00

[illegible]



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| ECV MATRIX | 130     | 140     | 150     | 160     | 170     | 180     |
|------------|---------|---------|---------|---------|---------|---------|
| -120.0     | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 |
| -150.0     | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 |
| -300.0     | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 |
| -400.0     | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 |
| -120.0     | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 | 1000.00 |

AFATL P-001 AAASIM "AIRCRAFT COMBAT SURVIVABILITY SCENARIO"

VULNERABLE AREA (SQ. METERS) AS A FUNCTION OF IMPACT SPEED (METERS/SEC) AND ASPECT VIEW  
 VULNERABLE AREA TABLE VS TYPE 3 WEAPON

[illegible]

AFATL P-001 AAASIN - AIRCRAFT-COMBAT-SURVIVABILITY-SCENARIO-

[illegible]



AFATL P-001 AAASIN # AIRCRAFT COMBAT-SURVIVABILITY-SCENARIO-

[illegible]





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AFATL P-001 AASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

| LOC | PKILL     | RCUNDS | FIRE TIME | XCUM     | YCUM     | RADUS GL | T REACT | T TRACK | GT | EM | CR | SR | LOC |
|-----|-----------|--------|-----------|----------|----------|----------|---------|---------|----|----|----|----|-----|
| 1   | 0.0010546 | 17     | 5.25      | 14200.00 | 5000.00  | 0.00     | 0.00    | 2.5000  | 1  | 1  | 1  | 1  | 1   |
| 2   | 0.0003651 | 0      | 6.80      | 13600.00 | 8200.00  | 0.00     | 0.00    | 2.5000  | 1  | 1  | 1  | 1  | 2   |
| 3   | 0.0       | 0      | 0.00      | 13400.00 | 8000.00  | 0.00     | 0.00    | 2.5000  | 1  | 1  | 1  | 1  | 3   |
| 4   | 0.0       | 0      | 0.00      | 13000.00 | 9700.00  | 0.00     | 0.00    | 2.5000  | 1  | 1  | 1  | 1  | 4   |
| 5   | 0.0112059 | 422    | 21.10     | 15600.00 | 10900.00 | 0.00     | 0.00    | 6.0000  | 1  | 1  | 1  | 1  | 5   |
| 6   | 0.0884375 | 134    | 51.42     | 12800.00 | 7500.00  | 0.00     | 0.00    | 6.0000  | 1  | 1  | 1  | 1  | 6   |

| LOC | TIME 1 | TIME 2 | TIME 3 | TIME 4 | TIME 5 | TIME 6 | TIME 7 | TIME 8 | TIME 9 | TIME 10 |
|-----|--------|--------|--------|--------|--------|--------|--------|--------|--------|---------|
| 1   | 1.47   | 2.95   | 34.42  | 45.89  | 57.37  | 68.84  | 80.31  | 91.78  | 103.26 | 599.99  |
| 2   | 1.47   | 2.95   | 34.42  | 45.89  | 57.37  | 68.84  | 80.31  | 91.78  | 103.26 | 599.99  |
| 3   | 1.47   | 2.95   | 34.42  | 45.89  | 57.37  | 68.84  | 80.31  | 91.78  | 103.26 | 599.99  |
| 4   | 1.47   | 2.95   | 34.42  | 45.89  | 57.37  | 68.84  | 80.31  | 91.78  | 103.26 | 599.99  |
| 5   | 1.47   | 2.95   | 34.42  | 45.89  | 57.37  | 68.84  | 80.31  | 91.78  | 103.26 | 599.99  |
| 6   | 1.47   | 2.95   | 34.42  | 45.89  | 57.37  | 68.84  | 80.31  | 91.78  | 103.26 | 599.99  |

| LOC | PK(1F 1) | PK(1F 2) | PK(1F 3) | PK(1F 4) | PK(1F 5) | PK(1F 6) | PK(1F 7) | PK(1F 8) | PK(1F 9) | PK(1F 10) |
|-----|----------|----------|----------|----------|----------|----------|----------|----------|----------|-----------|
| 1   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 2   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 3   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 4   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 5   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 6   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 7   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 8   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 9   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 10  | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |

| LOC | PK(1I 1) | PK(1I 2) | PK(1I 3) | PK(1I 4) | PK(1I 5) | PK(1I 6) | PK(1I 7) | PK(1I 8) | PK(1I 9) | PK(1I 10) |
|-----|----------|----------|----------|----------|----------|----------|----------|----------|----------|-----------|
| 1   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 2   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 3   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 4   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 5   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 6   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 7   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 8   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 9   | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |
| 10  | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0       |

AFATL P-001-AAASIM - AIRCRAFT COMBAT SURVIVABILITY SCENARIO

| LOC  | PK(TI 1) | PK(TI 2) | PK(TI 3) | PK(TI 4) | PK(TI 5) | PK(TI 6)  | PK(TI 7)  | PK(TI 8)  | PK(TI 9)  | PK(TI 10) |
|------|----------|----------|----------|----------|----------|-----------|-----------|-----------|-----------|-----------|
| 7    | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0025023 | 0.0166661 | 0.0491013 | 0.0005465 | 0.0003037 |
| 7CUM | 0.0      | 0.0      | 0.0      | 0.0      | 0.0      | 0.0025023 | 0.0191064 | 0.0672699 | 0.0281327 | 0.0684356 |

| LOC | RHO 1  | RHO 2 | RHO 3 | RHO 4 | RHO 5 | RHO 6 | RHO 7 | RHO 8 | RHO 9 |
|-----|--------|-------|-------|-------|-------|-------|-------|-------|-------|
| 1   | 0.0000 |       |       |       |       |       |       |       |       |
| 2   | 0.0000 |       |       |       |       |       |       |       |       |
| 3   | 0.0000 |       |       |       |       |       |       |       |       |
| 4   | 0.0000 |       |       |       |       |       |       |       |       |
| 5   | 0.0000 |       |       |       |       |       |       |       |       |
| 6   | 0.0000 |       |       |       |       |       |       |       |       |

ATTRITION ACCRUED AS A FUNCTION OF TIME OF FIRE

| TIME<br>SEC. | CUM FOR<br>CLASS 1 | DENSITY<br>CLASS 1 | DENSITY<br>CLASS 2 | DENSITY<br>CLASS 3 | DENSITY<br>CLASS 4 | DENSITY<br>CLASS 5 | DENSITY<br>CLASS 6 | DENSITY<br>CLASS 7 | DENSITY<br>CLASS 8 | DENSITY<br>CLASS 9 |
|--------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|
| 1            | 0.0                | 0.0                |                    |                    |                    |                    |                    |                    |                    |                    |
| 2            | 0.0                | 0.0                |                    |                    |                    |                    |                    |                    |                    |                    |
| 3            | 0.0                | 0.0                |                    |                    |                    |                    |                    |                    |                    |                    |
| 4            | 0.0024345          | 0.0024345          |                    |                    |                    |                    |                    |                    |                    |                    |
| 5            | 0.0025303          | 0.0049648          |                    |                    |                    |                    |                    |                    |                    |                    |
| 6            | 0.0220685          | 0.0270333          |                    |                    |                    |                    |                    |                    |                    |                    |
| 7            | 0.0653287          | 0.0923622          |                    |                    |                    |                    |                    |                    |                    |                    |
| 8            | 0.0766286          | 0.1689908          |                    |                    |                    |                    |                    |                    |                    |                    |
| 9            | 0.0802163          | 0.2456071          |                    |                    |                    |                    |                    |                    |                    |                    |
| 10           | 0.0802163          | 0.2456071          |                    |                    |                    |                    |                    |                    |                    |                    |

ATTRITION ACCRUED AS A FUNCTION OF TIME AT INTERCEPT

| TIME<br>SEC. | CUM FOR<br>CLASS 1 | DENSITY<br>CLASS 1 | DENSITY<br>CLASS 2 | DENSITY<br>CLASS 3 | DENSITY<br>CLASS 4 | DENSITY<br>CLASS 5 | DENSITY<br>CLASS 6 | DENSITY<br>CLASS 7 | DENSITY<br>CLASS 8 | DENSITY<br>CLASS 9 |
|--------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|--------------------|
| 1            | 0.0                | 0.0                |                    |                    |                    |                    |                    |                    |                    |                    |
| 2            | 0.0                | 0.0                |                    |                    |                    |                    |                    |                    |                    |                    |
| 3            | 0.0                | 0.0                |                    |                    |                    |                    |                    |                    |                    |                    |
| 4            | 0.0                | 0.0                |                    |                    |                    |                    |                    |                    |                    |                    |
| 5            | 0.0025023          | 0.0025023          |                    |                    |                    |                    |                    |                    |                    |                    |
| 6            | 0.0191067          | 0.0216090          |                    |                    |                    |                    |                    |                    |                    |                    |
| 7            | 0.0676140          | 0.0892237          |                    |                    |                    |                    |                    |                    |                    |                    |
| 8            | 0.0760475          | 0.1652712          |                    |                    |                    |                    |                    |                    |                    |                    |
| 9            | 0.0802164          | 0.2456071          |                    |                    |                    |                    |                    |                    |                    |                    |
| 10           | 0.0802164          | 0.2456071          |                    |                    |                    |                    |                    |                    |                    |                    |
| TOTALS       |                    | 0.0802243          |                    |                    |                    |                    |                    |                    |                    |                    |

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TOTAL PK FGR DENSITY CLASS 1 AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.

[illegible]

\*\*\*\*\* END OF JOB - - P001 SCENARIO RUN COMPLETE \*\*\*\*\*



# APPENDIX H

## P001 INPUT PROGRAM (PIP) LISTING

```

*****
THIS PROGRAM WILL PUNCH ALL REQUIRED CARDS FOR THE EXECUTION OF THE
POOL ANALYSIS WITH THE EXCEPTION OF THE FIRST CARD (THE GREEN JOB
CARD) AND THE LAST CARD (THE ORANGE END OF FILE CARD). THE MINIMAL
INPUT TO THIS PROGRAM IS THE X, Y AND Z COORDINATES FOR EACH OF THE
FLIGHT PATH MILESTONES. SIX GUN EMPLACEMENT LOCATIONS MAY BE
SPECIFIED IF THE SIX PRESET GUN LOCATIONS ARE NOT DESIRED. IF THE
PRESET GUN EMPLACEMENT LOCATIONS ARE USED, THE FINAL DATA CARD IS A
CONTROL CARD THAT SPECIFIES THE DESIRED INPUT, OUTPUT AND SCENARIO
OPTIONS. IF THE PRESET GUN LOCATIONS ARE NOT USED, THE FINAL DATA
CARDS WILL BE THE 6 INPUT GUN LOCATION CARDS.
*****
MILESTONE CARDS: THE X, Y, Z COORDINATES FOR UP TO 199 FLIGHT PATH
MILESTONES MUST BE ENTERED IN 3F10.0 FORMAT, ONE MILESTONE PER INPUT
CARD. THE VALUES (IN METERS) MUST BE ENTERED IN DECIMAL FORM WITH THE
X COORDINATE IN COLUMNS 1-10, THE Y COORDINATE IN COLUMNS 11-20 AND
Z COORDINATE IN COLUMNS 21-30. THE DECIMAL POINT MUST APPEAR IN EACH
COORDINATE VALUE.
*****
MILESTONE DELIMITER CARD: A CARD CONTAINING 99999. LEFT JUSTIFIED
(INPUT FORMAT F6.0) MUST BE PLACED AFTER THE FINAL MILESTONE CARD TO
MARK THE END OF THE MILESTONE DATA INPUT.
*****
CONTROL CARD: A CONTROL CARD SPECIFYING THE CRUISE SPEED, THE NUMBER
OF THE BOMB RELEASE MILESTONE (COUNT THE INITIAL POSITION AS MILE-
STONE 1), THE "99999." CARD. THE DATA ON THE OPTION CARD MUST BE
SPECIFIED IN F10.0, I2, 8I1, F10.0 FORMAT.
*****
F10.0: THE AIRCRAFT CRUISE SPEED IN METERS PER SECOND.
I2: THE NUMBER OF THE BOMB RELEASE MILESTONE.
I1: EW OPTION - 0 FOR NO EW; 1 FOR EW (WITH EW, AN AIRBORNE
JAMMER IS USED TO JAM TRACKING RADARS.
I1: ANTI-JAM OPTION - 0 FOR NO AJ; 1 FOR AJ (WITH AJ, RADAR
SYSTEMS WITH THE CAPABILITY USE AJ AGAINST THE JAMMER.)
I1: MULTIPATH OPTION - 0 FOR NO MULTIPATH; 1 FOR MULTIPATH (IF
SPECIFIED, THAT ARE AFFECTED BY RADAR MULTIPATH ARE INCLUDED IN THE RADAR
SYSTEMS THAT ARE AFFECTED BY MULTIPATH.)
I1: GUN EMPLACEMENT LOCATION INPUT OPTION - 0 FOR PRESET GUN
LOCATIONS; 1 FOR GUN EMPLACEMENT LOCATIONS INPUT AS DATA; DECK:
I1: LIST OF POOL INPUT DECK PROVIDED AS PART OF THE OUTPUT;
1 FOR LIST OF POOL INPUT DECK PROVIDED AS PART OF THE OUTPUT;
I1: PUNCH OPTION - 0 FOR NO POOL INPUT PUNCHED CARDS DESIRED;
1 FOR POOL INPUT DECK PROVIDED AS PART OF THE OUTPUT.
I1: PLOT OPTION - 0 FOR NO PLOT OF FLIGHT PATH AND GUN LOCATIONS
DESIRED; 1 FOR PLOT DESIRED.
*****

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C      DATA XGUN/14800.,16200.,13600.,13400.,11300.,15600.,12800./
490      DATA YGUN/9000.,8200.,7200.,8000.,9700.,10900.,7500./
495      DATA ZGUN/40.,40.,20.,20.,50.,90.,20./
500
505
510
515
520
525
530
535
540
545
550
555
560
565
570
575
580
585
590
595
600
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715
720

C      VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS
C
C      DATA VAT1N2/2*.4645,6*7.107,2*.6568,6*5.551,2*.6968,6*5.574,2*.656
18*7.357,2*.6968,6*5.574,2*.4645,6*7.7432,2*.6568,6*2.858,2*.4645,6*
26*7.357,2*.6568,6*2.858,2*.4645,6*7.7432,2*.6568,6*2.858,2*.4645,6*
33*298,2*.6568,6*2.858,2*.4645,6*7.7432,2*.6568,6*2.858,2*.4645,6*
42*58,2*.6568,6*2.858,2*.6568,6*5.551,2*.6968,6*5.574,2*.6568,6*
57*2*.6968,6*5.574,2*.6568,6*5.551,2*.6968,6*5.574,2*.6568,6*
62*.6968,6*5.574,2*.4645,6*7.107/
C
C      VULNERABLE AREA TABLE VS TYPE 3 WEAPONS
C
C      DATA VAT3/2*12.54,6*13.47,2*9.853,6*10.51,2*9.639,6*11.15,2*12.64,
16*14.78,2*9.639,6*11.15,2*9.853,6*10.51,2*9.639,6*11.15,2*12.64,
21*14.78,2*9.639,6*11.15,2*9.853,6*10.51,2*9.639,6*11.15,2*12.64,
37*62,6*6.240,8*1.394,2*4.762,6*6.240,2*5.342,6*7.762,6*2.24,
40*2*9.853,6*10.51,2*9.639,6*11.15,2*12.64,6*14.78,2*5.639,6*11.15,
52*9.853,6*10.51,2*9.639,6*11.15,2*12.64,6*14.78,2*5.639,6*11.15,
612.54,6*13.47/
C
C      VULNERABLE AREA TABLE VS TYPE 5 WEAPONS
C
C      DATA VAT5/8*55.37,8*43.22,8*47.10,8*62.53,8*47.10,8*43.22,8*47.10,
18*62.53,8*47.10,8*5.761,8*27.45,8*33.07,8*27.45,8*5.761,8*27.45,8*
233*07,8*27.45,8*5.761,8*47.10,8*62.53,8*47.10,8*43.22,8*47.10,8*62
3.53,8*47.10,8*55.37/
C
C      RADAR CROSS SECTION TABLE
C
C      DATA RCSTAB/19*1000.,19*100.,1.273,178.,053,023,156.,166.,063,1
173,235.,347,02,015,011,073,072,019,028,11,454,005,005,
2*007,011,093,093,039,068,112,032,015,015,009,035,032,0
3017,033,102,505,273,178,053,023,156,166,063,173,235,0
447,02,015,011,073,072,019,028,11,454,19*100.,19*1000./
C
C      CALL ERRSET SUPPRESSES ANY POSSIBLE UNDERFLOW PROBLEMS THAT MAY
C      RESULT FROM MANIPULATION OF SCENARIO PARAMETERS.
C
C      CALL ERRSET (208,50,-1,1,1)
C
C      PI = 3.14159
C      MNUM = 0
C      IBR = 0

```



```

C C C C C
      READ THE MILESTONE CARDS, COUNT THE NUMBER OF MILESTONES AND STOP
      INPUT UPON REACHING THE '99999'. DELIMITER CARC.
      DO 1 I=1,200
      READ (5,40) X(I),Y(I),Z(I)
      IF (X(I).EQ.99999.) GO TO 2
      MNUM = MNUM+1
      X10(I) = X(I)/2000.0
      Y10(I) = Y(I)/2000.0
      1 CONTINUE
C C C C C
      2 CONTINUE
      READ THE CRUISE SPEED, BOMB RELEASE MILESTONE, EW OPTION, ANTI-JAM
      OPTION, MULTIPATH OPTION, GUN LOCATION INPUT OPTION AND THE LIST
      PUNCH, PLOT AND EXTENDED OUTPUT OPTICS AND THE JAMMER POWER.
      READ (5,42) CVEL,MBR,IEW,IAJ,IMULT,IGUN,ILST,IPNCH,IPLT,TEXT,PJAM
      JAMMER ON BUT JAMMER POWER = 0 ??? THEN TURN JAMMER OFF.
      IF (IEW.EQ.1.AND.PJAM.LE.0.0) IPJAM1=1
      IF (IPJAM1.EQ.1) IEW=0
      IF (IPJAM1.EQ.1) WRITE (6,43)
      JAMMER POWER GREATER THAN 1000 WATTS ??? THEN LIMIT IT TO 1000 WATTS
      IF (PJAM.GT.1000) IPJAM2=1
      IF (IPJAM2.EQ.1) PJAM=1000
      IF (IPJAM2.EQ.1.AND.IEW.EQ.1) WRITE (6,44)
      AJ ON BUT JAMMER OFF ??? THEN TURN AJ OFF.
      IF (IAJ.EQ.1.AND.IEW.EQ.0) WRITE (6,45)
      IF (IEW.EQ.0) IAJ = 0
      OPTION TO INPUT THE 6 ADDITIONAL GUN EMPLACEMENT LOCATIONS
      IF (IGUN.NE.1) GO TO 3
      DC 3 I=1,6
C C C C C

```

```

3      READ (5,41,END=38) XGUN(I),YGUN(I),ZGUN(I)
3      CONTINUE
CCCCCCCC
      CALCULATE THE X, Y AND Z VELOCITIES, THE CLIMB ANGLE, HEADING AND
      ROLL ANGLE AND THE TIME AT EACH MILESTONE ALONG WITH VARIOUS OTHER
      PARAMETERS FOR LATER USE.
CCCCCCCC
      DO 9 I=1,MNUM
      IF (I.NE.MNUM) GO TO 4
      CALCULATIONS FOR THE FINAL MILESTONE
      HDG(I) = HDG(I-1)
      HCGDEG(I) = HDG(I)*57.29578
      IF (HCGDEG(I).LT.0) HDGDEG(I)=HDGDEG(I)+360
      RA(I) = 0
      CADEG(I) = 0
      ZCOT(I) = 0
      XDOT(I) = VEL(I)*COS(HDG(I))
      YDOT(I) = VEL(I)*SIN(HDG(I))
      TNRT(I) = 0
      GO TO 8
      GENERAL COORDINATE CALCULATIONS FOR LATER USE
      CC
      4      CCCONTINUE
      DX = X(I+1)-X(I)
      DX2 = DX**2
      CY = Y(I+1)-Y(I)
      DY2 = DY**2
      DZ = Z(I+1)-Z(I)
      DZ2 = DZ**2
      DIST = SQRT(DX2+DY2+DZ2)
      CLIMB ANGLE CALCULATIONS
      CC
      CA(I) = ATAN2(DZ,SQRT(DX2+DY2))
      IF (CA(I).GT.1.5533) CA(I)=1.5533
      CADEG(I) = CA(I)*57.29578
      HEADING CALCULATIONS
      CC
      IF (DX.NE.0.OR.DY.NE.0) GO TO 5
      HCG(I) = HDG(I-1)

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```

GC TO 6
5 HDG(I+1) = ATAN2(DY,DX)
6 HDG(I) = HDG(2)
6 HCGDEG(I) = HDG(I)*57.29578
IF (HCGDEG(I).LT.0) HCGDEG(I)=HCGCEG(I)+360

VELOCITY CALCULATIONS
VEL(I) = CVEL
VEL(I+1) = VEL(I)-TAN(CA(I))*DIST/100+(CVEL-VEL(I))*(DIST/VEL(I))/
130

BCMB RELEASE POINT CONSIDERATIONS
IF (I.EQ.MBR) IBR = 1

**** RESTRICTION: MAX VEL PRIOR TO BOMB RELEASE POINT IS 260 MPS.
IF (VEL(I+1).GT.260.AND.IBR.EQ.0) VEL(I+1)=260

**** RESTRICTION: MAX VEL AFTER BCMB RELEASE PCINT IS 310 MPS.
IF (VEL(I+1).GT.310.AND.IBR.NE.0) VEL(I+1)=310
VAVG = (VEL(I)+VEL(I+1))/2

VELOCITY COMPONENT CALCULATIONS
ZCOT(I) = VEL(I)*SIN(CA(I))
XYVEL = VEL(I)*COS(CA(I))
XCOT(I) = XYVEL*COS(HDG(I))
YDOT(I) = XYVEL*SIN(HDG(I))

MILESTONE TIME CALCULATIONS
T(I) = 0
T(I+1) = T(I)+DIST/VAVG
DT(I) = 1
IF (I.EQ.1) GO TO 7
DT(I) = T(I)-T(I-1)

TURN RATE AND ROLL ANGLE CALCULATIONS
7 TNANG = HDG(I+1)-HDG(I)
TNRT(I) = TNANG/DT(I)
RA(I) = ATAN(TNRT(I)*VAVG/9.81)*57.29578
8 CONTINUE
9 CONTINUE

```



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CC CC      CARD 2 TIME INCREMENT CALCULATION
CC CC      TINC = T(MNUM)/1000
CC CC      CARD 6 TIME INCREMENT CALCULATIONS
CC CC      TINKI = 0
CC CC      DO 10 I=1,9
CC CC      TINK(I) = TINKI+T(MNUM)/10
CC CC      TINKI = TINK(I)
CC CC      10 CONTINUE
CC CC      /////*****PRINT PROGRAM*****/////
CC CC      OPTION TO LIST THE POOL CARD DECK
CC CC      IF (ILST.EQ.0) GO TO 16
CC CC      COMMENCE PRINTED OUTPUT OF THE POOL CARD DECK.
CC CC      THE JCL CARDS.
CC CC      WRITE (6,47)
CC CC      WRITE (6,48)
CC CC      WRITE (6,49)
CC CC      WRITE (6,50)
CC CC      WRITE (6,51)
CC CC      WRITE (6,52)
CC CC      WRITE (6,53)
CC CC      WRITE (6,54)
CC CC      WRITE (6,55)
CC CC      WRITE (6,56)
CC CC      BLANK CARD FOR RADAR MASK ANGLE = 0.
CC CC

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```

      WRITE (6,57)
      THE OUTPUT TITLE CARD.
      WRITE (6,58)
      CARD 2
      WRITE (6,59) T(MNUM),TINC
      THE 2A CARDS (MILESTONES).

      DC 11 I=1,MNUM
      WRITE (6,60) T(I),X(I),Y(I),Z(I),XDOT(I),YDOT(I),ZDOT(I),PCGDEG(I)
      1,CADEG(I),RA(I)
      11 CONTINUE

      WRITE (6,61)
      WRITE (6,62)
      CARD 3 (GUN EMPLACEMENT CARD).
      WRITE (6,63) XGUN(1),YGUN(1),ZGUN(1)
      CARD 4 (GUN TYPE).
      WRITE (6,64)
      CARD 5
      WRITE (6,65)
      CARD 6
      WRITE (6,66) (TINK(I),I=1,9)
      CARD 7 (VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS)
      WRITE (6,67)
      WRITE (6,68) (VATIN2(I),I=1,208)
      CARD 12 (EXECUTE RUN).
      EXTENDED OUTPUT OPTION

```





|   |  |  |      |
|---|--|--|------|
| C | EXTENDED OUTPUT OPTION                           |  | 2165 |
| C |  | IF (IEXT.NE.1) WRITE (6,70)            | 2170 |
| C |  | IF (IEXT.EQ.1) WRITE (6,69)            | 2175 |
|   |  | WRITE (6,63) XGUN(6), YGUN(6), ZGUN(6) | 2180 |
|   |  | WRITE (6,75)                           | 2185 |
| C | MULTIPATH OPTICN                                 |  | 2190 |
| C |  | IF (IMULT.NE.1) GO TO 13               | 2200 |
| C | CARD 13 (MULTIPATH EFFECTS)                      |  | 2210 |
| C |  | IRMP = 1                               | 2220 |
| C |  | WRITE (6,76) IMULT,IRMP                | 2230 |
| C | 13 CONTINUE                                      |  | 2240 |
| C |  |  | 2250 |
| C | EXTENDED OUTPUT OPTION                           |  | 2260 |
| C |  | IF (IEXT.NE.1) WRITE (6,70)            | 2270 |
|   |  | IF (IEXT.EQ.1) WRITE (6,69)            | 2280 |
|   |  | WRITE (6,63) XGUN(7), YGUN(7), ZGUN(7) | 2285 |
|   |  | WRITE (6,77)                           | 2290 |
| C | CARD 7 (VULNERABLE AREA TABLE VS TYPE 5 WEAPONS) |  | 2300 |
| C |  | WRITE (6,148)                          | 2305 |
|   |  | WRITE (6,68) (VAT5(I), I=1,208)        | 2310 |
| C | MULTIPATH OPTICN                                 |  | 2315 |
| C |  | IF (IMULT.NE.1) GO TO 14               | 2320 |
| C | CARD 13 (MULTIPATH EFFECTS)                      |  | 2330 |
| C |  | IRMP = 2                               | 2335 |
|   |  | WRITE (6,76) IMULT,IRMP                | 2340 |
| C | 14 CCNTINUE                                      |  | 2350 |
| C |  |  | 2355 |
| C | EW (JAMMER) OPTION                               |  | 2360 |
| C |  | IF (IEW.NE.1) GO TO 15                 | 2370 |
| C |  | IRECM = 2                              | 2380 |
|   |  |  | 2390 |
|   |  |  | 2400 |

00150505  
44142423  
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C C C      WRITE (7,86)
C C C      WRITE (7,83)
C C C      WRITE (7,87)
C C C      WRITE (7,83)
C C C      WRITE (7,88)

      LEADING BLANK DATA CARD SIGNIFIES RADAR MASKING ANGLE OF ZERO.

      WRITE (7,89)

      THE OUTPUT TITLE CARD.

      WRITE (7,90)

      CARD 2

      WRITE (7,91) T(MNUM),TINC

      THE 2A CARDS (MILESTONES).

      DC 17 I=1,MNUM
      WRITE (7,92) T(1),X(1),Y(1),Z(1),XCOT(1),YDCT(1),ZDOT(1),HDGDEG(1)
      1,CADEG(1),RA(1)
      17 CCNTINUE

C C C

      WRITE (7,93)
      WRITE (7,94)

      CARD 3 (GUN EMPLACEMENT CARD).

      WRITE (7,95) XGUN(1),YGUN(1),ZGUN(1)

      CARD 4 (GUN TYPE).

      WRITE (7,96)

      CARD 5

      WRITE (7,97)

      CARD 6

      WRITE (7,98) (TINK(I),I=1,9)
C

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C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS)
      WRITE (7,99)
      WRITE (7,100) (VATIN2(I),I=1,208)
C C
C C CARD 12 (EXECUTE RUN).
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C THE REMAINDER OF THE CARDS INTRODUCE NEW GUN LOCATIONS, GUN TYPES
C C AND VULNERABLE AREA TABLES TO BE EXECUTED BY THE PROGRAM.
C C WRITE (7,95) XGUN(2),YGUN(2),ZGUN(2)
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(3),YGUN(3),ZGUN(3)
C C WRITE (7,103)
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(4),YGUN(4),ZGUN(4)
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(5),YGUN(5),ZGUN(5)
C C WRITE (7,104)
C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 3 WEAPONS)
C C WRITE (7,145)
C C WRITE (7,100) (VAT3(I),I=1,208)
C C EW (JAMMER) OPTION
C C IF (IEW.NE.1) GO TO 18
C C IRECM = 1
C C SJTMAX = 3
C

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3100
3105
3110
3115
3120

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C C CARD 14 (SPECIFIES EW OPTION AND JAMMER INFO.)
C C WRITE (7,105) IEW,PJAM,IREFM,SJTMAX
C C WRITE THE RADAR CROSS SECTION TABLE.
C C WRITE (7,106) (RCSTAB(I),I=1,133)
C C 18 CONTINUE
C C
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(6),YGUN(6),ZGUN(6)
C C WRITE (7,107)
C C
C C MULTIPATH OPTION
C C IF (IMULT.NE.1) GC TO 19
C C IRMP = 1
C C
C C CARD 13 (MULTIPATH EFFECTS)
C C WRITE (7,108) IMULT,IRMP
C C 15 CONTINUE
C C
C C EXTENDED OUTPUT OPTION
C C IF (IEXT.NE.1) WRITE (7,102)
C C IF (IEXT.EQ.1) WRITE (7,101)
C C WRITE (7,95) XGUN(7),YGUN(7),ZGUN(7)
C C WRITE (7,109)
C C
C C CARD 7 (VULNERABLE AREA TABLE VS TYPE 5 WEAPON)
C C WRITE (7,150)
C C WRITE (7,151) (VAT5(I),I=1,208)
C C
C C MULTIPATH OPTION
C C IF (IMULT.NE.1) GC TO 20
C C IRMP = 2
C C CARD 13 (MULTIPATH EFFECTS)

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C      WRITE (7,108) IMULT,IRMP
C      20 CCNTINUE
C      EW (JAMMER) OPTION
C      IF (IEW.NE.1) GO TO 21
C      IRECM = 2
C      ANTI-JAM OPTION
C      IF (IAJ.EQ.1) IRECM=3
C      IF (IRECM.EQ.2) SJTMAX=1.5
C      IF (IRECM.EQ.3) SJTMAX=17
C      CARD 14 (SPECIFIES EW OPTION AND JAMMER INFO.)
C      WRITE (7,105) IEW,PJAW,IRECM,SJTMAX
C      WRITE THE RADAR CROSS SECTION TABLE.
C      WRITE (7,106) (RCSTAB(I),I=1,133)
C      21 CONTINUE
C      EXTENDED OUTPUT OPTION
C      IF (IEXT.NE.1) WRITE (7,102)
C      IF (IEXT.EQ.1) WRITE (7,101)
C      WRITE (7,93)
C      WRITE (7,110)
C      WRITE (7,93)
C      ////*****PLOT SECTION*****////
C      OPTCN TC PLOT THE POOL SCENARIO
C      22 CCNTINUE
C      INITIALIZE PLOT
C      CALL PLOTS
C      IF (IPLOT.EQ.0) GO TO 25
C      ESTABLISH X AXIS

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165,3.6,3.5,3.35,3.3,3.15,3.3,3.5,4.0,4.15,4.25,4.2,4.0,3.8/
CALL LINE (RVX2,RVY2,27,1,1)
REAL RVX3(11)/6.85,6.9,6.85,7.0,7.35,7.5,7.7,8.0,8.4,8.8,9.0/
REAL RVY3(11)/3.6,3.35,3.0,2.6,2.3,2.25,2.2,1.65,1.3,1.0,0.8/
CALL LINE (RVX3,RVY3,11,1,1)
REAL RVX4(7)/7.23,7.3,7.3,7.5,8.0,8.5,9.0/
REAL RVY4(7)/3.9,3.7,3.35,3.0,2.6,2.1,1.7/
CALL LINE (RVX4,RVY4,7,1,1)

CC
PLOT THE BIG ISLAND
CC
REAL BIX1(12)/7.84,8.0,8.14,8.31,8.4,8.38,8.2,8.0,7.85,7.83,7.85,7
1.84/
REAL BIY1(12)/2.3,2.22,2.18,1.8,1.7,1.63,1.68,1.82,2.0,2.1,2.2,2.3
1/
CALL LINE (BIX1,BIY1,12,1,1)

CC
PLOT THE XY AXIS ARROWS
CC
REAL XYX1(9)/0.4,0.35,0.45,0.4,0.4,1.0,0.8,0.8,1.0/
REAL XYX1(9)/1.0,0.8,0.8,1.0,0.4,0.4,0.45,0.35,0.4/
CALL LINE (XYX1,XYX1,9,1,1)
REAL XCHAR(1)/X/
CALL SYMBOL (0.8,0.1,0.21,XCHAR,0.0,1)
REAL YCHAR(1)/Y/
CALL SYMBOL (0.1,0.8,0.21,YCHAR,0.0,1)

CC
PLOT THE TRUE NORTH ARROW
CC
REAL TNX1(5)/2.0,2.0,1.95,2.05,2.0/
REAL TNY1(5)/4.8,5.4,5.2,5.4/
CALL LINE (TNX1,TNY1,5,1,1)
CALL TNCHAR(1)/N/
CALL SYMBOL (1.9,5.5,0.21,TNCHAR,0.0,1)

CC
WRITE THE PLOT TITLE
CC
REAL PTITLE(6)/AE,3251,POC,1 SC,ENAR,10 /
CALL SYMBOL (2.3,6.5,0.21,PTITLE,0.0,24)

CC
PLOT THE MILESTONES AND FLIGHT PATH
CC
CALL LINE (X10,Y10,MNUM,1,7)

CC
PLOT THE BOMB RELEASE POINT
CC
RX = X10(MBR)
RY = Y10(MBR)

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CALL LINE (RX,RY,1,1,6)  
PLOT THE GUN LOCATIONS AND RANGE RADIUS

```

DC 24 I=1,7
TIMIX = XGUN(1)/2000.0
TIMIY = YGUN(1)/2000.0
IF (I-GE.3) MKR = 3
IF (I-GE.5) MKR = 4
IF (I-GE.7) MKR = 5
IF (I-GE.1) MKR = 2
IF (I-GE.3) RAD = 0.78
IF (I-GE.5) RAD = 0.91
IF (I-GE.7) RAD = 1.42
CALL LINE (TIMIX,TIMIY,1,1,MKR)

```

```

DO 23 J=1,25
XCIRC(J) = TIMIX+RAD*COS(J*PI/12)
YCIRC(J) = TIMIY+RAD*SIN(J*PI/12)
IF (XCIRC(J).LT.0) XCIRC(J)=0
IF (YCIRC(J).LT.0) YCIRC(J)=0
IF (XCIRC(J).GT.9) XCIRC(J)=9
IF (YCIRC(J).GT.6) YCIRC(J)=6
CCNTINUE

```

```

CALL LINE (XCIRC,YCIRC,25,1,MKR)
CCNTINUE

```

```

CALL PLOT (0.0,12.0,-3)
CALL PLOTE
CONTINUE

```

/////\*\*\*\*\* ERROR CHECK SECTION \*\*\*\*\*////

CPTICN SUMMARY

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WRITE (6,111) MNUM,T(MNUM),MBR
WRITE (6,112)
WRITE (6,113) CVEL

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• LR

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99 FCRMAT ('07 VULNERABLE AREA TABLE VS TYPE 1 AND 2 WEAPONS')
100 FCRMAT ('08F8.3)
101 FCRMAT ('1211111111')
102 FCRMAT ('12)
103 FCRMAT ('04 211111)
104 FCRMAT ('04 34411)
105 FCRMAT ('14.13.5 0 1.0.7X,F6:1 ,4X,1.0E-06',6X,1.1,11X,
1 1.0.6X,15,F10.2 ,/GEND,/,3X,19 ,4X,1.0E-06',6X,1.1,11X,
2 1.0.0)
106 FCRMAT ('08F10.3,/,8F10.3,/,3F10.3)
107 FCRMAT ('04 33411)
108 FCRMAT ('13,13,15, C.35')
109 FCRMAT ('04 53211)
110 FCRMAT ('/GO.FT05F003 DD *')
111 FCRMAT ('1.*** PATH CONSISTS OF ,I3,, MILESTONES WITH A TOTAL ,
1 THE FLIGHT TIME OF ,F6.1, SECONDS ,/1X,
2 FLIGHT RELEASE IS AT MILESTONE ,I3, ,/1X,
3 BOMB *** CRUISE SPEED IS ,F6.1, METERS PER SECOND.))
113 FCRMAT ('1X, CRUISE SPEED IS ,F6.1, METERS PER SECOND.))
114 FCRMAT ('1X, A POOL INPUT LISTING IS NOT SPECIFIED AS CPUT.))
115 FCRMAT ('1X, A PUNCHED DECK IS NOT SPECIFIED AS CPUT.))
116 FCRMAT ('1X, A PUNCHED DECK IS NOT SPECIFIED AS CPUT.))
117 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
118 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
119 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
120 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
121 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
122 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
123 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
124 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
125 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
126 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
127 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
128 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
129 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
130 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
131 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
132 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
133 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))
134 FCRMAT ('1X, A SCENARIO PLOT IS NOT SPECIFIED AS CPUT.))

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135 FORMAT (1X, 'MAX ALTITUDE DURING PCP UP WAS ', F6.1, ' METERS WHICH',
1    ' IS LESS THAN ', F6.1, ' METERS', //)
136 FORMAT (1X, 'THE AIRCRAFT HEADING INTO THE BCMB RELEASE PCINT IS ',
1    ' OF 1219 METERS', //)
137 FORMAT (1X, 'THE HEADING TO THE TARGET IS ', F5.1, ' WHICH IS GREATER THAN THE ',
1    ' THE DEGREE DIFFERENCE IS ', F5.1, ' WHICH IS GREATER THAN THE ',
1    ' 5 DEGREE MAXIMUM DIFFERENCE LIMIT ', //)
138 FORMAT (1X, 'THE LENGTH OF THE LEG IMMEDIATELY PRIOR TO THE BCMB ',
1    ' RELEASE PCINT IS ', F4.2, ' SECONDS', //)
139 FORMAT (1X, 'THE MINIMUM OF 2 SECONDS', //)
140 FORMAT (1X, 'THE BCMB RELEASE ALTITUDE IS ', F7.1, ' METERS WHICH IS',
1    ' NOT IN THE BOMB RELEASE ALTITUDE', //)
141 FORMAT (1X, 'THE BOMB WAS RELEASED AT A DISTANCE OF ', F7.1,
1    ' METERS FROM THE TARGET WHICH IS IN EXCESS ', //)
142 FORMAT (1X, 'THE BCMB RELEASE MAXIMUM RANGE ', //)
143 FORMAT (1X, 'MILESTONE ', //)
144 FORMAT (1X, 'STALL OCCURS AT 90 METERS PER SECOND. DECREASE ', //)
145 FORMAT (1X, 'THE CLIMB ANGLE PRIOR TO MILESTONE ', //)
146 FORMAT (1X, 'MILESTONE ', //)
147 FORMAT (1X, 'TURN RESULTS IN A TURN RATE OF ', F1.1, ' ',
1    ' TURN RESULTS IN A G LOADING ', //)
148 FORMAT (1X, 'THE 6 G MAX LOADING ', //)
149 FORMAT (1X, 'THE TURN ANGLE AT MILESTONE ', //)
150 FORMAT (1X, 'THE TURN ANGLE AT MILESTONE ', //)
151 FORMAT (1X, 'NO FLIGHT PATH ERRORS IN THIS RUN *****', //)
152 FORMAT (1X, 'NO FLIGHT PATH ERRORS *****', //)
153 FORMAT (1X, 'NO FURTHER FLIGHT PATH ERRORS *****', //)
154 FORMAT (1X, 'END OF POOL INPUT PROGRAM - SUMMARY COMPLETE', //)
155 FORMAT (1X, 'THE EXTENDED OUTPUT OPTION IS NOT SPECIFIED', //)
156 FORMAT (1X, 'EXTENDED PRINTOUT IS PROVIDED AS OUTPUT', //)
157 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 3 WEAPONS', //)
158 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 5 WEAPONS', //)
159 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 3 WEAPONS', //)
160 FORMAT (1X, '07 VULNERABLE AREA TABLE VS TYPE 5 WEAPONS', //)
161 END

```

# APPENDIX I

## P001 PROGRAM LISTING (IBM)

```

CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C      AAASIM --- MULTIPLE GUN ANTI-AIRCRAFT ARTILLERY SIMULATION
C      WRITTEN BY THOMAS D. MCMURCHIE AND JAMES G. SEVERSON
C      AIRFORCE ARMAMENT LABORATORY (AFATL-DLYS) EGLIN AFB, FLORIDA
C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C      PROGRAM RECEIVED 25 OCT 72
C      CONVERTED AND MAINTAINED BY ASD/XROA
C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
C      MODIFIED FOR USE ON IBM 360/370 AT NPS MONTEREY, CA., 7 MAR 78.
C      FOR INFORMATION CALL PROF R.E. BALL 408-646-2885, (AUTOVON 878).
C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C
COMMON/BLOCK1/ITITLE(20)
COMMON/BLOCK2/NEPA,TMIN,TMAX,DTFPA
COMMON/BLOCK3/XGUN,YGUN,ZGUN
COMMON/BLOCK4/IGT,IEM,IC8,ISB,IGL,CIRCLE
COMMON/BLOCK5/NRHOS,RHO(9)
COMMON/BLOCK6/NTINTS,TINTER(10)
COMMON/BLOCK7/IVACOM(20),VA(19,5,9)
COMMON/BLOCK8/TREACT,TRACK1,TRACK2
COMMON/BLOCK9/TROUND(6),THDMAX(6),PHDMAX(6),PHIMIN(6),PHIMAX(6),
1  VELMIN(6),VELMAX(6),RANMAX(6),RANMIN(6),RANMAX(6),
2  ATLAG(6),ETHMAX(6),EPHMAX(6),TFMAX1(6),TFMAX2(6),RVACON(6),RVBCON(6),VMUZEL(6)
COMMON/BLOCK10/CONSTS/DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
COMMON/HEADFO/LINE,NUMBER
COMMON/NFARM/XA,YA,ZA,RA2,RA,TIME
COMMON/IGXGYG/IG,XG(8),YG(8)
COMMON/MAGIC/FRACT,INDEX1,INDEX2
COMMON/VSASBS/VMUZZ,ASHCON,BSHCON,CQUAD
COMMON/BUDGET/BDA(6)
COMMON/CECM/IRECM,IJ,GAINJ,IX,XSEC,CALX,PJW,
* * *
COMMON/X,Y,Z,ROL,PIT,HOG,
COMMON/FIGT,FJAM,GJ,SJT,SN
COMMON/NEWA/VY1,VY2,VX,ALFA,J,IRMP,PLEN,O1,O8,D1,JJJ,XR,ZT,PZI,IERR
COMMON/NEWB/VY1,P2,VX,BETA,J,IF6,IJAM,ISL,O4,CPS,TIPERS,JMODE,YI,T2
COMMON/NEWC/CP,X2,P1,B2,JFILE,IJAM,ISL,O4,CPS,TIPERS,JMODE,YI,T2
COMMON/NEWD/T33,X1,Z2,B1,VX2,C3,CPK,NROUND,IOFF,I,XI,TM,SP,T1
COMMON/NEWF/Z1,A2,VX1,VZ2,PHI,IF5,IEOF,YR,IF2,T13,IFILE,Y2,A1
COMMON/NEWG/V2,VZ1,V,FUZZ,IF9,SD2RJ,SD2RJM,V1,VY2,F,J,P,IF7,REFC
COMMON/XFPA(1201),YFPA(1201),ZFPA(1201),BFPA(1201),AFPA(1201),

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RMIN =RANMIN(IGT)
RMAX =RANMAX(IGT)
VMIN=VELMIN(IGT)
VMAX=VELMAX(IGT)
IG=1
LINE=66
IF (IPRINT(6).LE.0) GO TO 63
CALL PAGES(5,0,JP)
WRITE (6,1013) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
TTTTT RRRR A A CCC K K
T RRRR A A C C K K K
T RRRR A A C C K K K
T R R A A C C C K K
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
COMPUTE ACTUAL AIRCRAFT PARAMETERS AT FIRE TIME FOR USE IN ERROR EQS
60 CALL INTERP(TIME/DTFPA)
X=GETVAL(XFPA)-XGUN
Y=GETVAL(YFPA)-YGUN
Z=GETVAL(ZFPA)-ZGUN
VX=GETVAL(VXFPA)
VY=GETVAL(VYFPA)
VZ=GETVAL(VZFPA)
ILOOP = ILOOP+1
RCL = GETVAL(PFPA)
PIT = GETVAL(AFPA)
HDG = GETVAL(BFPA)
G2=X*X+Y*Y
G=SQRT(G2)
TRET=ATAN2(Y,X)
PHIT=ATAN2(Z,G)
IF MULTIPATH HAS BEEN SPECIFIED(IMUL=1) AND THE ELEVATION
ANGLE IS MEASURED BY RADAR(IOEM=3), COMPUTE THE ELEVATION
TRACKING BIAS(PBMP),VARIANCE(SP2MP),AND APPARENT TARGET
ALTITUDE(Z).
SP2MP = 0.
PBMP = J.
IF(IMUL.NE.1.OR. IOEM.NE.3) GO TO 30
CALL MULTPTH(IRMP,REFC,PHIT,PBMP,SP2MP)
PHIT = PHIT+PBMP
Z = G*TAN(PHIT)
CCCCCCCC

```

```

30 CONTINUE
   R = SQRT(G2+Z*Z)
   RD = (X*VX+Y*VY+Z*VZ)/R
   TD = (X*VY-Y*VX)/G2
   PD = (VZ-Z*RD/R)/G
   IF(TIME.GT.TMIN) GO TO 58
   RDD = 0.0
   TLD = 0.0
   PCD = 0.0
   RANS=R
   THES=THET
   PHIS=PHIT
   ERAN2 = 0.0
   ERAN3 = 0.0
   ERAN4 = 0.0
   ETHE2 = 0.0
   ETHE3 = 0.0
   ETHE4 = 0.0
   EPHI2 = 0.0
   EPHI3 = 0.0
   EPHI4 = 0.0
   GO TO 59
58 RCD=(RD-RDS)/.064
   TOD=(TD-TDS)/.064
   PDD=(PD-PDS)/.064
   (STORE PREVIOUSLY OBSERVED MEAN TRACKING ERRORS FOR USE IN MEAN
    TRACKING ERROR EQUATIONS)
59 ERAN1=ERAN2
   ERAN2=ERAN3
   ERAN3=ERAN4
   ERAN4=ERAN1+.71875*(ERAN2-ERAN1)
   ETHE1=ETHE2
   ETHE2=ETHE3
   ETHE3=ETHE4
   ETHE4=(G/R)*ANGLIM(THET-THES)
   ETHE=ETHE1+.71875*(ETHE2-ETHE1)
   EPHI1=EPHI2
   EPHI2=EPHI3
   EPHI3=EPHI4
   EPHI4=EPHI1+.71875*(EPHI2-EPHI1)
   EPHI=EPHI1+MASK*ANGLE
   IF(PHIT.LE.AMASK) TFIRE= TIME+ TREACT+TTRACK
   (SKIP FIRE ATTEMPT IF INSUFFICIENT TRACKING TC FIRE)

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C

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C



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C      IF(TIME-LE.TFIRE)GO TO 62
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C      FFFF  IIIII  RRRR  EEEEE
C      F      I      R      E
C      FFFF  I      RRRR  EEEEE
C      F      I      R      E
C      F      IIIII  R      EEEEE
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C      VG=SQRT(VX*VX+VY*VY)
C      V=SQRT(VG*VG+VZ*VZ)
C      PSI=ANGLIM(PFPA(INDEX1))+FRACT*ANGLIM(PFPA(INDEX2))-PFPA(INDEX1))
C      (SKIP FIRE ATTEMPT IF MAX ALLOWED TRACKING ERROR IS EXCEEDED)
C
C      54 IF(ABS(ETHE4).GT.ETMAX)GO TO 64
C      IF(ABS(EPHI4).GT.EPMAX)GO TO 64
C      IF(IDEM.GT.1)GO TO 56
C      (LIMIT INPUT RANGE ESTIMATE)
C      RC=AMAX1(RMIN,AMIN1(RMAX,RANS-0.575*RD))
C      CCMPUTE MEAN ASSUMED TIME OF FIRE AIRCRAFT POSITION (MECHANICAL
C      COMPUTATION)
C      XF=RC*CTBCPB-XG(IG)
C      YF=RC*STBCPB-YG(IG)
C      ZF=RC*SPB
C      GF=SQRT(XF*XF+YF*YF)
C      RF=SQRT(GF*GF+ZF*ZF)
C      (SET UP MATRIX T, THE TRANSFORMATION BETWEEN THE LINE OF SIGHT
C      SYSTEM AND THE FALSE HORIZON SYSTEM)
C      T22 = X/G
C      CT  = X/G
C      ST  = Y/G
C      T33 = G/R
C      CP  = G/R
C      T13 = Z/R
C      SP  = Z/R
C      CS = COS(PSI)
C      SS = SIN(PSI)
C      CA = VG/V
C      SA = VZ/V

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```

CG=VX/VG
SG=VY/VG
T11=CT*CP
T12=ST*CP
T21=-ST
T31=-CT*SP
T32=-ST*SP
  (SET UP FALSE HORIZON SYSTEM VELOCITY COMPONENTS)
VXP=T11*VX+T12*VY+T13*VZ
VYP=T21*VX+T22*VY
VZP=T31*VX+T32*VY+T33*VZ
VGP=SQR T(VXP*VXP+VYP*VYP)
CAP=VGP/V
SAP=VZP/V
CBP=VXP/VGP
SBP=VYP/VGP
  (SET UP UNIT VECTOR OUT LEFT WING OF AIRCRAFT)
UX=-SA*CG*SS-SG*CS
UY= CG*CS-SA*SG*SS
UZ= CA*SS
UZP=T31*UX+T32*UY+T33*UZ
IF(VXP) 31,32,31
CSP=-T11*UX-T12*UY-T13*UZ
GO TO 33
32 CSP=(VGP*(T21*UX+T22*UY)+UZP*VZP*SBP)/VXP
33 SSP=UZP/CAP
  (COMPUTE FALSE HORIZON SYSTEM MEAN AND STANDARD DEVIATION OF
  ERROR IN DIVE AND COURSE ANGLE ESTIMATES)
EMAP=SAP*(.3196*ABS(CBP)-.1859*ABS(SBP))
ESAP=.04712+.08063*ABS(SAP)*(1.0+1.16*ABS(CBP))
EMBP=.4060*CAP*SBP*CBP
ESBP=(.1670-.08098*ABS((CBP*CBP-SBP*SBP)*CSP)+
1 SEMAP=.09006*ABS(SBP*SSP*CSPI))/CAP
SEMAP=SIN(EMAP)
CEMAP=COS(EMAP)
SEMBP=SIN(EMBP)
CEMBP=COS(EMBP)
THE NEXT CARD ELIMINATES AN EQUIVALENCE BY USING A DOUBLE
REPLACEMENT.
A33 = SAP*CEMAP + CAP*SEMAP
SABP= SAP*CEMAP + CAP*SEMAP
CABP=CAP*CEMAP-SAP*SEMAP

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      SBBP=SBP*CEMBP+CBP*SEMBP
      CBBP=CBP*CEMBP-SBP*SEMB
      (LIMIT VELOCITY ASSESSMENT (MECHANICAL COMPUTERS))
      VBP=AMIN1(VMAX,AMAX1(VMIN,V))
      ESVP=ESVPC*T*V
      SET UP THE ELEMENTS OF THE MATRIX A.
      A CONTAINS THE PARTIALS OF VXE,VYE,VZE W.R.T. ALPHA,BETA,SPEED

      A31=CABP*CBBP
      A32=CABP*SBBP
      A21=-VBP*A32
      A22=VBP*A31
      A11=-VBP*SABP*CBBP
      A12=-VBP*SABP*SBBP
      A13=VBP*CABP

      COMPUTE MEAN ESTIMATED VELOCITY COMPONENTS (MECHANICAL
      COMPUTATION)
      VXE=(A31*T11+A32*T21+A33*T31)*VBP
      VYE=(A31*T12+A32*T22+A33*T32)*VBP
      VZE=(A31*T13
      GO TO 63

      COMPUTE MEAN ASSUMED TIME OF FIRE AIRCRAFT POSITION (ELECTRONIC
      COMPUTATION)

      56 XF=RANS*CTBCPB
      YF=RANS*STBCPB
      ZF=RANS*SPB
      GF=RANS*CPB
      RF=RANS

      ITERATION TO DETERMINE MEAN THEORETICAL INTERCEPT POINT

      63 RS=0.0
      VS=VMUZZ
      T=0.0
      21 XE=XF+VXE*T
      YE=YF+VYE*T
      ZE=ZF+VZE*T
      XE2=XE*XE
      YE2=YE*YE
      ZE2=ZE*ZE
      GE2=XE2+YE2

```



```

RE2=GE2+ZE2
RE=SQRT(RE2)
IF(RC-LE.1.0)GO TO 22
VC=VS-(XE*VXE+YE*VYE+ZE*VZE)/RE
IF(VD-LE.1.0)GO TO 64
T=T+RC/VD
IF(T.GT.TTFMAX)GO TO 64
RS=RSHELL(T)
VS = VSHELL(T)
GC TO 21
T2=T*T
22
C
C
C
CHANGE 22 JAN 76
IF INTERCEPT POINT BELOW MASK*****SKIP FIRE
GE= SQRT(GE2)
IF(ATAN2(ZE,GE).LE. AMASK) GO TO 64
C
C
C
ITERATION TO DETERMINE ACTUAL INTERCEPT POSITION, RANGE, AND TIME
TU=AMIN1(TFMAX,TMAX-TIME)
23 CALL RPLANE(TU)
IF(RSHELL(TU).GT.RA)GO TO 24
C
C
C
(SKIP FIRE ATTEMPT IF SHELL CANNOT CATCH AIRCRAFT)
IF((XA*GETVAL(VXFPA)+YA*GETVAL(VYFPA)+ZA*GETVAL(VZFFPA))/RA .LT.
1 VSHELL(TU))GO TO 64
TU=TU-1.0
IF(TU)64,64,23
TL=0.0
24 T=0.5*(TL+TU)
25 CALL RPLANE(T)
RC=RA-RSHELL(T)
IF(RC.GT.1.0)GO TO 26
IF(RC.GT.-1.0)GO TO 37
TL=T
GC TO 25
26 TL=T
GC TO 25
27 GC=VS*RE-XE*VXE-YE*VYE-ZE*VZE
Q1=(VZE-VS*ZE/RE)/Q0
Q2=(XE*VXE-YE*VYE)/Q0
C
C
C
COMPUTE THE PARTIAL DERIVATIVES OF BIG THETA
DTDX=Q2*XE-YE

```

```

C
C
DITDY=Q2*YE+XE
DITDZ=Q2*ZE
DITDR=(ZF*DITDZ+ YF*DITDY+XF*DITDX)/RF
DITDI= XF*DITDY-YF*DITDX
DITDP= GF*DITDZ-(YF*DITDY+XF*DITDX)*ZF/GF
COMPUTE THE PARTIAL DERIVATIVES OF BIG PHI
C
C
DPPDX=Q1*XE
CPPDY=Q1*YE
DPPDZ=Q1*ZE+1.0
DPPDR=(ZF*DPPDZ+ YF*DPPDY+XF*DPPDX)/RF
DPPDI= XF*DPPDY-YF*DPPDX
DPPDP= GF*DPPDZ-(YF*DPPDY+XF*DPPDX)*ZF/GF
GE4=GE2*GE2
GC TO (210,220,230,240),IOEM
C
C
C COMPUTATIONS FOR MODE 1 OPERATION
C
210 GO TO (211,212,213,214,215,999),IGT
C
C TRACKING ERROR DISTRIBUTION SIZES (SPHERICAL COORDINATES)
C (FOR GT 1, 2, OR 3)
C
211 CONTINUE
212 CONTINUE
213 SR2=(123.0+0.0225*R)**2
ST2=(.0643*TD)**2
SP2=(.1320*PD)**2
GO TO 219
C
C (FOR GT 4 AND 5)
C
214 CONTINUE
215 SR2=(123.0+0.0225*R)**2
ST2=(0.0167-.000710/(.0517+ABS(TD)))**2
SP2=(0.0116-.000216/(.0235+ABS(PD)-4.0*PDD))**2
C
C SET UP THE ELEMENTS OF THE MATRIX B=AT
C
219 B11=A11*T11+A12*T21+A13*T31
B12=A11*T12+A12*T22+A13*T32
B13=A11*T13+A12*T23+A13*T33
B21=A21*T11+A22*T21
B22=A21*T12+A22*T22
B23=A21*T13+A22*T23
B31=A31*T11+A32*T21+A33*T31
B32=A31*T12+A32*T22+A33*T32

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```

      B33=A31*T13      +A33*T33
CC      COMPUTE THE PARTIAL DERIVATIVES OF BIG, THETA AND BIG PHI W.R.T.
CC      ON CARRIAGE INPUTS (ALPHA, BETA, SPEED)
CC
      DTIDAP=B11*CTTDX+B12*DTTDX+B13*DTTDX
      DTIDBP=B21*CTTDX+B22*DTTDX+B23*DTTDX
      DTIDVP=B31*CTTDX+B32*DTTDX+B33*DTTDX
      DPPDAP=B11*CPPDX+B12*DTTDX+B13*DTTDX
      DPPDBP=B21*CPPDX+B22*DTTDX+B23*DTTDX
      DPPDVP=B31*CPPDX+B32*DTTDX+B33*DTTDX
CC      COMPUTE THE VARIANCES OF BIG, THETA AND BIG PHI
CC
      STT2=((DTIDR**2)*SR2 + (DTIDI**2)*ST2 + (DTIDVP**2)*SP2
1      + ((DTIDAP**ESAP)**2 + (DTIDBP**ESBP)**2 + (DTIDVP**ESVP)**2)/GE4
      SPP2=((DPPDR**2)*SR2 + (DPPDI**2)*ST2 + (DPPDVP**2)*SP2
1      + ((DPPDAP**ESAP)**2 + (DPPDBP**ESBP)**2 + (DPPDVP**ESVP)**2)/GE2
      GO TO 29
CC
      COMPUTATIONS FOR MODE 2, 3, OR 4 OPERATION
CC
      220 GO TO (999,999,223,999,225,226),IGT
CC
      TRACKING ERROR DISTRIBUTIONS SIZES (SPHERICAL COORDINATES)
      (FOR MODE 2)
CC
      223 CONTINUE
      225 CONTINUE
      226 SR2=(41.0+0.0075*R)**2
      ST2=(.000982+.1681*TD*TD)**2
      SP2=(.000491+.033*ABS(ABS(PD)-4.0*PDD))**2
      GO TO 65
      230 GO TO (999,999,233,999,235,236),IGT
CC      (FCR MODE 3)
CC
      233 CONTINUE
      235 CONTINUE
      236 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SC2RJ
      ST2=(.00196+0.050*TD)**2
      SP2=(.000982+0.11*ABS(ABS(PO)-2.0*PDD))**2+SP2MP
      GO TO 65
      240 GO TO (999,999,243,999,245,246),IGT
CC      (FCR MODE 4)
CC
      243 CONTINUE

```



```

245 CONTINUE
246 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SC2RJ
    ST2=(0.000982+0.1681*TD*TD)**2
    SP2=(0.000491+0.033*ABS(ABS(PD)-4.0*POD))**2+SP2MP
C
65 R8TD=RANS*THESD
    R8PD=RANS*PHISD
C
    VELOCITY COMPONENT ERROR DISTRIBUTION SIZES
C
    SVX2=(SR2*(PHISD*CTBSPB+THESD*STBCPB)**2
1      +ST2*(R8PD*STBSPB-R8TD*CTBSPB-RANS*STBSPB)**2
2      +SP2*(R8PD*CTBSPB-R8TD*STBSPB+RANS*CTBSPB)**2)*ATLCON
    SVY2=(SR2*(PHISD*STBSPB-THESD*CTBSPB)**2
1      +ST2*(R8PD*CTBSPB+R8TD*STBSPB-RANS*CTBSPB)**2
2      +SP2*(R8PD*STBSPB-R8TD*CTBSPB+RANS*CTBSPB)**2)*ATLCON
    SVZ2=(SR2*(PHISD*CPB)**2+SP2*(R8PD*SPB-RANS*CPB)**2)*ATLCON
C
    COMPUTE THE VARIANCES OF BIG THETA AND BIG PHI
C
    STT2=((DTDR**2)*SR2 + (DTTD**2)*ST2 + (DTDD**2)*SP2
1      + (((DTDX**2)*SVX2 + (DTDY**2)*SVY2 + (DTDZ**2)*SVZ2)*T2)/GE4
1+SD2J
    SPP2=((DPPDR**2)*SR2 + (DPPD**2)*ST2 + (DPPD**2)*SP2
1      + (((DPPDX**2)*SVX2 + (DPPDY**2)*SVY2 + (DPPDZ**2)*SVZ2)*T2)/GE2
1+SD2J
C
    COMPUTATION OF VULNERABLE AREA OF AIRCRAFT AT INTERCEPT
C
29 VP=VSHELL(T)
    XU=XE/RE
    YU=YE/RE
    ZU=ZE/RE
    XE=XU*RA
    YE=YU*RA
    ZE=ZU*RA
    GE=SQRT(XE*XE+YE*YE)
    VXA=GETVAL(VXFPA)
    VYA=GETVAL(VYFPA)
    VZA=GETVAL(VZFPA)
    VGA=SQRT(VXA*VXA+VYA*VYA)
    VA=SQRT(VGA*VGA+VZA*VZA)
    VXI=VP*XU-VXA
    VYI=VP*YU-VYA
    VZI=VP*ZU-VZA
    VI=SQRT(VXI*VXI+VYI*VYI+VZI*VZI)
    ALFA=GETVAL(AFPA)
    CA=COS(ALFA)

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```

SA=SIN(ALFA)
BETA=BFPA(INDEX1)+FRACT*ANGLIM(BFPA(INDEX2)-BFPA(INDEX1))
CG=COS(BETA)
SG=SIN(BETA)
PZI=PFPA(INDEX1)+FRACT*ANGLIM(PFPA(INDEX2)-PFPA(INDEX1))
THE NEXT CARDS ELIMINATE AN EQUIVALENCE BY USING A DOUBLE
REPLACEMENT.
T33 = COS(PZI)
CP = COS(PZI)
T13 = SIN(PZI)
SP = SIN(PZI)
Q1=VXI*CG+VYI*SG
Q2=VZI*CA-QI*SA
Q3=VYI*CG-VXI*SG
VXF=Q1*CA+VZI*SA
VYF=Q2*SP+Q3*CP
VZF=Q2*CP-Q3*SP
      (SET UP INDICES FOR VULNERABLE AREA INTERPOLATION)
F1=ATAN2(VYF,VXF)/QTRPI
IF(F1.LT.0.0)F1=F1+8.0
I1=F1-FLOAT(I1)
I1=I1+1
F2=ARCOS(VZF/VI)/QTRPI
I2=F2
I2=I2+1
F3=AMIN1(7.599999999,VI/152.4)
I3=F3-FLOAT(I3)
I3=I3+1
C1=1.0-F1
D2=1.0-F2
D3=1.0-F3
J1=I1+1
J2=I2+1
J3=I3+1
      (PERFORM LINEAR THREE DIMENSIONAL INTERPOLATION)
AVT=D3*(D1*VAT(I1,I2,I3)+F1*VAT(J1,I2,I3))+
1      F2*(D1*VAT(I1,J2,I3)+F1*VAT(J1,J2,I3))+
2      F3*(D1*VAT(I1,I2,J3)+F1*VAT(J1,I2,J3))+
3      F2*(D1*VAT(I1,J2,J3)+F1*VAT(J1,J2,J3))
      SET UP DISTRIBUTION SIZES OF OTHER SOURCES OF RANDOM ERROR

```

C

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SVA=XU*VXA+YU*VYA+ZU*VZA
CVA2=VA*VA-SVA*SVA
VMQ=.99*VMUZZ/RA-ASHCON
DTI=(VMQ-SQT(VMQ*VMQ-4.0*BSSHCON))/(2.0*BSSHCON)-T
SLXMV2=CVA2*(DTI*VP/(VP-SVA/.99))**2
SLYFR2=(0.010*VA*T)**2
SLXFR2=SLYFR2*(1.0+CVA2/(VP-SVA)**2)
SAOAP2=(0.003*RA)**2
SAOGJ2=(0.003*RA)**2
SAOBD2=(BDACON(IGT)*RA)**2
IF(10EM.EQ.1)GO TO 68
SAOPE2=(0.002*RA)**2
GO TO 67
68 SAOPE2=(V*V-((X*VX+Y*VY+Z*VZ)/R)**2)*
RA2*(0.0001463-7.478E-11*(R-1386.0)**2)**2
67 CCIIST=SAOAP2+SAOGJ2+SAOPE2+SAOBD2

```

COMBINE ALL ERRORS INTO CNE DISTRIBUTION, COMPUTE BIAS

C  
C  
C

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SXA2=STT2*RA2
SYA2=SPP2*RA2
SXL2=CDIST+SLXMV2+SLXFR2
CTT=XE/GE
STT=YE/GE
CPP=GE/RE
SPP=ZE/RE
BXA=X*STT-YA*CTT
BYA=ZA*CPP-(YA*STT+XA*CTT)*SPP
BXA2=BXA*BXA
BYA2=BYA*BYA
VAM=VXA*STT-VYA*CTT
VAP=VZA*CPP-(VYA*STT+VXA*CTT)*SPP
VAM2=VAM*VAM
VAP2=VAP*VAP
VAI2=VAM2+VAP2
CD2=VAM2/VAI2
SD2=VAP2/VAI2
SXA22=SXA2+CD2*SXL2+SD2*SYL2
SYA22=SYA2+CD2*SYL2+SD2*SXL2
TWOCOV=2.0*VAM*VAP*(SXL2-SYL2)/VAI2
DIF=SXA22-SYA22
DEN=2.0*SQT(TWOCOV+DIF*DIF)
HC2Z=DIF/DEN
CZ2=0.5+HC2Z
SZ2=0.5-HC2Z
SZCZ=TWOCOV/DEN

```



```

33 STUFF=2.0*SZCZ*BXA*BYA
   SXF2=CZ2*BXA2+SZ2*BYA2+STUFF
   SYF2=CZ2*BYA2+SZ2*BXA2-STUFF
   SYF2=CZ2*SXA2+SZ2*SYA2+SZCZ*TWOCOV
   SYF2=CZ2*SYA2+SZ2*SXA2-SZCZ*TWOCOV
   AVTPI=AVT/PI2

   CCMPUTE PROBABILITY OF KILL

   STUFF=BXF2/(SXF2+AVTPI)+BYF2/(SYF2+AVTPI)
   IF(STUFF.LT.50.0)GO TO 75
   PK=0.0
   GO TO 78
75 PK=AMIN1(1.0,EXP(-.5*STUFF)*AVTPI/SQRT((SXF2+AVTPI)*(SYF2+AVTPI)))
78 PS=(1.0-PK)**1SB
   PK=1.0-PS
   TI=TIME+T

   CCACCUMULATE PK AS A FUNCTION OF INPUT TIME INTERVALS

   I=0
   I=I+1
50 IF(TIME-GE.TINTER(I))GO TO 50
   J=I
51 IF(TI.LT.TINTER(J))GO TO 52
   J=J+1
   GO TO 51
52 PTOTTF(I)=PK+PS*PTOTTF(I)
   PTOTTI(J)=PK+PS*PTOTTI(J)
   CPS=CPS+PS

   CCACCUMULATE PK FOR EACH SPHERICAL SECTOR

   SPKT(11,12,13)=PK+PS*SPKT(11,12,13)

   CCMPUTE QUANTITIES FOR EXTENDED OUPUT, WHEN DESIRED

   IF(IPRINT(6).LE.0)GO TO 20
   O1=THESD*DEGREE
   O2=PHISD*DEGREE
   O3=ETHE4*DEGREE
   C4=EPHI4*DEGREE
   ***** ABS(BXF2,BYF2,SXF2,SYF2) WERE ADDED BELCW AS A TEMPORARY
   ***** MEASURE TO PRECLUDE NEGATIVE ARGUMENTS FOR A SQRT FUNCTION.
   ***** CAUSE OF NEGATIVE ARGUMENT: POSSIBLE ACCURACY DIFFERENCES
   ***** BETWEEN THE CDC AND IBM TYPE COMPUTERS.
   ***** CHANGE MADE BY LCDR C. SWENSON ON 3 MAR 1978.
   BXF2 = ABS(BXF2)

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UUU

UUU

FIRE ADDITIONAL GUNS IN COMPLEX, IF ANY

УУУ

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62 TIME=TIME+0.064
   IF (TIME.GT.TMAX) GO TO 69

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U U





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C 340 GO TO (999,999,343,999,345,346),IGT
343 CONTINUE
345 CONTINUE
346 THESD=0.910*TD+0.45*TDD+6.0*ETHE
PHISD=0.75*PD-0.25*PDD+6.0*EPHI
RANS=0.804*RD + 3.0*ERAN

C (LIMIT SLEW RATES AND ELEVATION ANGLE TO WEAPON MAXIMUMS)
C
73 THESD=SIGN(AMIN1(TDMAX,ABS(THESD)),THESD)
PHISD=SIGN(AMIN1(PDMAX,ABS(PHISD)),PHISD)
THES=ANGLIM(THES+0.064*THESD)
PHIS=AMAX1(PHMIN,AMIN1(PHMAX,PHIS+0.064*PHISD))
RANS=AMAX1(0.0,RANS+0.064*RANS)
CTB=COS(THES)
STB=SIN(THES)
CPB=COS(PHIS)
SPB=SIN(PHIS)
CTBCPB=CTB*CPB
CTBSPB=CTB*SPB
STBCPB=STB*CPB
STBSPB=STB*SPB
RCS=RD
TCS=TD
PCS=PD
IF(ITEM.LT.2)GO TO 60

C COMPUTE MEAN (SMOOTHED) VELOCITY COMPONENTS (ELECTRONIC
C COMPUTATION)
VXE=RANS*CTBCPB-RANS*(STBCPB*THESC+CTBSPB*PHISD)
VYE=RANS*STBCPB+RANS*(CTBCPB*THESC-STBSPB*PHISD)
VZE=RANS*SPB+RANS*CPB*PHISD
VXE=VXE+EMDTA*(VXES-VXE)
VYE=VYE+EMDTA*(VYES-VYE)
VZE=VZE+EMDTA*(VZES-VZE)
VXES=VXE
VYES=VYE
VZES=VZE
GO TO 60

1013 FORMAT(LOCATION',15,6X,GUN TYPE',12,6X,ERROR MODE',12,6X,
1 POSITION=(F8.1,F8.1,F8.1,F8.1,F8.1,F8.1,M./OL
FIRE FLT. INTCP,INTCP,INTCP,INTCP,INTCP,INTCP,AZIM,
2 MEAN MEAN VULN',N OEM VEL. TIME RATE RATE AZ.ERR EL.ERR
3 SIG2 MEAN BIAS1 BIAS2
4 AREA SIG2 CUM.PK./)
5 AREA SHOT PK
1014 FORMAT(I2,I3,2F7.2,F8.2,2F7.0,1X,2F6.1,1X,2F7.1,F8.1,1X,2F7.2,1X,

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```

1044 2 3F7.2,2F9.5)
15X,12,3X,E10.4,5X,E10.4,5X,F7.2,5X,F7.2,5X,F7.2,
15 CALL $15
69 CALL $69
999 CALL $999
8888 CALL $8888
9994 CALL $9994
9995 CALL $9995
9996 CALL $9996
9997 CALL $9997
7777 CONTINUE
END

SUBROUTINE TOOBIG WAS CONSTRUCTED FROM A PORTION OF THE ORIGINAL POOL
MAIN PROGRAM IN ORDER TO ENABLE PROPER COMPILATION OF THE MAIN.

SUBROUTINE TOOBIG (TEMP, SPKT, SPKT2)
COMMON/BLOCK1/ITITLE(20)
COMMON/BLOCK2/NFPA,TMIN,TMAX,DTFPA
COMMON/BLOCK3/XGUN,YGUN,ZGUN
COMMON/BLOCK4/IGT,IEM,ICB,ISB,IGL,CIRCLE
COMMON/BLOCK5/NRHOS,RHQ(9)
COMMON/BLOCK6/NTINTS,TINTER(10)
COMMON/BLOCK7/IVACCM(20),VAT(9,5,5)
COMMON/BLOCK8/TREACT,TRACK1,TRACK2
COMMON/BLOCK9/TROUND(6),THDMAX(6),PHDMAX(6),PHIMIN(6),PHIMAX(6),
VELMIN(6),VELMAX(6),RANMIN(6),RANMAX(6),
ATLAG(6),ETHMAX(6),EPHMAX(6),RMODES(6),
COMMON/BLOCKA/TFMAX1(6),TFMAX2(6),RVACON(6),RVBCON(6),VMUZEL(6)
COMMON/CONSTS/DEGREE,RADIAN,PI,P12,QTRPI,SQRT2
COMMON/HEADFO/LINE,NUMBER
COMMON/NFPA/PA,XA,YA,ZA,RA2,RA,TIME
COMMON/IGXGYG/IG,XG(8),YG(8)
COMMON/MAGIC/FRACT,INDEX1,INDEX2
COMMON/VSASBS/VMUZZ,ASHCON,BSHCON,CQUAD
COMMON/BUDGET/BDACON(6)
COMMON/CECM1/IRECM,IJ,GAINJ,IX,XSEC,CALX,PJW,
X,Y,Z,ROL,PIT,PDG,
X,FIGT,FJAM,GJ,SJT,SN
COMMON/NEWA/VY1,P2,VX,BETA,IRMP,PLEN,O1,O8,D1,JJJ,XR,ZT,PZI,IERR
COMMON/NEWB/VY1,P2,VX,BETA,IF6,IMUL,IP,N,O7,D,TTRACK,KMODE,PSI,K
COMMON/NEWC/CP,X2,P1,B2,JFILE,IJAM,ISL,O4,CPS,TPERS,JMCDE,YT,T2
COMMON/NEWD/T33,X1,Z2,B1,VX2,C3,CPK,NROUND,IOFF,I,Xf,tm,SP,T1
COMMON/NEWE/Z1,A2,VX1,VZ2,PHI,IF5,IEOF,YR,IF2,T13,IFILE,Y2,A1
COMMON/NEWF/V2,VZ1,V,FUZZ,SD2RJ,SD2RJM,V1,VY2,F,JP,IF7,REFC
COMMON/NEWG/SJIMAX,O2,PK,D2,ISW

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```

PZI=PSI*RADIAN
T13 = SIN(PZI)
SP = SIN(PZI)
T33 = COS(PZI)
CP = COS(PZI)
LINE=66
IF(JMODE)5,5,1
1 ASSIGN 4 TO IEOF
2 ASSIGN 3 TO IERR
IF(JMODE.GT.IFILE)GO TO 2
REWIND 9
IFILE=1
GO TO 4
IFILE=1 FILE+1
3 READ(9,ERR=9992,END=9993)
GO TO 3
4 IF(JMODE.GT.IFILE)GO TO 2
ASSIGN 9999 TO IECF
AREAC 28 TO IERR
28 ASSIGN 9996 TO IERR
ASSIGN=6
5 T1=T2
X1=X2
Y1=Y2
Z1=Z2
P1=P2
A1=A2
B1=B2
V1=V2
VX1=VX2
VY1=VY2
VZ1=VZ2
IF(JMODE)6,6,7
6 READ(INUNIT,1000,ERR=9992,END=9993)T2,X,Y,Z,VX,VY,VZ2,B2,A2,P2
B2=B2*RADIAN
A2=A2*RADIAN
P2=P2*RADIAN
V2=SQRT(VX*VX+VY*VY+VZ2*VZ2)
GO TO 9
7 JMODE=JMODE+1
IF(JMODE.LE.6)GO TO 8
JMODE=1
8 READ(9,ERR=9992,END=9993)TEMP
T2 =TEMP( 1,JMODE)
X =TEMP( 2,JMODE)
Y =TEMP( 3,JMODE)
Z =TEMP( 4,JMODE)

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```

VX = TEMP( 5, JMODE )
VY = TEMP( 6, JMODE )
VZ2 = TEMP( 7, JMODE )
VZ = TEMP( 11, JMODE )
BZ = TEMP( 13, JMODE )
A2 = TEMP( 14, JMODE )
P2 = TEMP( 15, JMODE )
9 X2 = XT + (X - XR) * CP + (Y - YR) * SP
  Y2 = YT + (Y - YR) * CP - (X - XR) * SP
  Z2 = ZT + Z
  VX2 = VX * CP + VY * SP
  VY2 = VY * CP - VY * SP
  B2 = ANGL IM(B2 - PZ1)
  T2 = T2 - TMIN
10 IF (TM - T1) / (T2 - T1)
  F = (TM - T1) / (T2 - T1)
  NFPA = NFPA + 1
  XFPA(NFPA) = X1 + F * (X2 - X1)
  YFPA(NFPA) = Y1 + F * (Y2 - Y1)
  ZFPA(NFPA) = Z1 + F * (Z2 - Z1)
  VXFPA(NFPA) = VX1 + F * (VX2 - VX1)
  VYFPA(NFPA) = VY1 + F * (VY2 - VY1)
  VZFPA(NFPA) = VZ1 + F * (VZ2 - VZ1)
  V = V1 + F * (V2 - V1)
  BFPA(NFPA) = ANGL IM(B1 + F * ANGL IM(B2 - B1))
  AFPA(NFPA) = A1 + F * (A2 - A1)
  PFPA(NFPA) = ANGL IM(P1 + F * ANGL IM(P2 - P1))
  K = K - 1
  IF (K) 34, 34, 35
34 K = KMODE
  PHI = DEGREE * PFPA(NFPA)
  BETA = DEGREE * BFPA(NFPA)
  ALFA = DEGREE * AFPA(NFPA)
  CALL PAGES(1, 2, JP)
  IF (JP.EQ.0) WRITE (6, 1003)
  WRITE(6, 1004) TM, XFPA(NFPA), YFPA(NFPA), VYFPA(NFPA), VZFPA(NFPA), BETA, ALFA, PHI
1  TM = OTFPA * FLOAT(NFPA)
  IF (TM - LE.TMAX) GO TO 10
  TMAX = TM - OTFPA - FUZZ
  CALL PAGES(4, 0, JP)
  WRITE(6, 1005) XR, YR, XT, YT, PSI, ZT
  CALL TPLOT(NFPA)
  IF (JMODE.GT.0) OR. INUNIT.EQ.8) GO TO 12
  ASSIGN 12 IC IEOF
13 READ(INUNIT, 1000, ERR=9992, END=9993)
  GC TO 13

```

C

```

C      DECODE DATA BLOCK 03 -- WEAPON LOCATION
C      103  READ (99,1008)      XGUN,YGUN,ZGUN
C           GO TO 12
C
C      DECODE DATA BLOCK 04 -- WEAPON TYPE, MODE, NUMBER OF BARRELS (CYCLIC
C      AND SIMULTANEOUS), NUMBER OF WEAPONS PER
C      LOCATION, RADIUS OF CIRCLE OF WEAPON COMPLEX
C      104  READ (99,1009)      IGT,IEM,ICB,ISB,IGL,CIRCLE
C           IF(IGL-1)9998,61,66
C           61  XG(1)=0.0
C              YG(1)=0.0
C              GO TO 12
C           66  DC 16 I=1,IGL
C              F=PI2*FLOAT(I)/FLOAT(IGL)
C              XG(I)=CIRCLE*COS(F)
C              YG(I)=CIRCLE*SIN(F)
C              16  GO TO 12
C
C      DECODE DATA BLOCK 05 -- WEAPON DENSITY FACTORS
C      105  READ (99,1015)      IF5,NRHOS,(RHO(I),I=1,NRHOS)
C           GO TO 12
C
C      DECODE DATA BLOCK 06 -- PK ACCRUAL TIME INTERVALS
C      106  READ (99,1015)      IF6,NTINTS,(TINTER(I),I=1,NTINTS)
C           NTINTS=NTINTS+1
C           TINTER(NTINTS)=999.99
C           GO TO 12
C
C      DECODE DATA BLOCK 07 -- AIRCRAFT VULNERABLE AREAS
C      107  DC 11 I=1,20
C           11  IVACOM(I)=ICARD(I)
C              READ(INUNIT,1000,ERR=9992,END=9993)(VAT(1,1,K),K=2,9)
C              DC 88 J=2,4
C              DO 89 I=1,8
C              READ(INUNIT,1000,ERR=9992,END=9993)(VAT(I,J,K),K=2,9)
C              CONTINUE
C              DC 88 K=2,9
C              88  VAT(9,J,K)=VAT(1,J,K)
C              READ(INUNIT,1000,ERR=9992,END=9993)(VAT(1,5,K),K=2,9)
C              DC 18 K=2,9
C              DO 18 I=2,9
C              VAT(1,5,K)=VAT(1,5,K)
C              18  VAT(1,1,K)=VAT(1,1,K)

```



```

IF7=1
GO TO 12
C DECODE DATA BLOCK 08 -- WEAPON REACTION AND TRACK TIMES
C
108 READ (99,1008) TREACT,TRACK1,TRACK2
C GO TO 12
C DECODE DATA BLOCK 09 -- WEAPON PARAMETERS
C
109 READ (99,1008) TROUND(IGT),TFDMAX(IGT),PHDMAX(IGT),
1 PHIMIN(IGT),PHIMAX(IGT),VELMIN(IGT),VELMAX(IGT),
2 RANMIN(IGT),RANMAX(IGT)
1 READ(INUNIT,1000,ERR=9992,END=9993)ATLAG(IGT),ETHMAX(IGT),
1 EPHMAX(IGT),RMODES(IGT)
1 IF9=1
GC TO 12
C DECODE DATA BLOCK 10 -- SHELL PARAMETERS
C
110 READ (99,1008) TFMAX1(IGT),TFMAX2(IGT),RVACON(IGT),
1 RVBCON(IGT),VMUZEL(IGT)
1 IF9=1
GO TO 12
C DECODE DATA BLOCK 11 -- INPUT OPTION (CARD/TAPE)
C
111 READ (99,1028) I
81 IF(I)81,81,82
81 INUNIT=5
82 GO TO 15
82 INUNIT=8
ASSIGN 83 TO IEQF
IF(I.GT.JFILE)GO TO 84
REWIND 8
JFILE=1
83 IF(I-JFILE)12,12,84
84 JFILE=JFILE+1
85 READ(8,1000,ERR=9992,END=83)
GC TO 85
C DECODE DATA BLOCK 13 --- LOW ALTITUDE RADAR MULTIPATH EFFECT
C
113 READ (99,1038) IMUL,IRMP,REFC
IF(IMUL.EQ.0) GO TO 12
CALL PAGES(12,0,JP)
WRITE(6,1098) IRMP,REFC
1098 FORMAT('///', MULTIPATH INPUTS (INITIAL OR CHANGED '.,//,

```









```

D=RHO(I)
PK=D*CPK
PKTTDC(I)=PK+(1.0-PK)*PKTTDC(I)
DC 55 J=1,NTINTS
D1=D*PTOTTF(J)
PKTFDC(J,I)=D1+(1.0-D1)*PKTFDC(J,I)
D2=D*PTOTTI(J)
55 PKTTDC(J,I)=D2+(1.0-D2)*PKTTDC(J,I)
C
C COMPUTE, STORE, AND WRITE TOTAL PKS FOR ENTIRE ARRAY OF WEAPONS
C
F=FLOAT(NROUND/ISB)*TPERS
WRITE(7)ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTTI,IPRINT
ISL=ISL+1
ASSIGN 70 TO IEOF
GO TO 15 7
70 ENDFILE 7
C
C REMIND 7
C LINE=66
PUNCH2 CARDS WRITE ONLY ONE CARD IMAGE ON TAPE4 WHEN THERE IS
ONLY ONE DEFENSE PER EMPLOYMENT (I.E., ONLY ONE "12" CARD
BEFORE 7/8/9 END-OF-RECORD).
K=ISL
CALL PAGES(4,0,JP)
WRITE(6,1033)
ASSIGN 40 TO IEOF
79 READ(7,ERR=9992,END=9993)ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,
1 TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTTI,IPRINT
IF(JP.EQ.0) WRITE(6,1033)
WRITE(6,1034)ISL,CPK,NROUND,F,XGUN,YGUN,ZGUN,CIRCLE,IGL,
1 TREACT,TTRACK,IGT,IEM,ICB,ISB,ISL
1 WRITE(4,1060) ITITLE(9),ITITLE(10),ISL,IGT,IEM,ICB,ISB,IGL,CPK,
1 NROUND,XGUN,YGUN,ZGUN,F,NUMBER
GO TO 79
40 IF(K.NE.2) WRITE(4,1064) ITITLE(9),ITITLE(10),PKTTDC(1),NUMBER
IF(IFLAGS(2).LE.0) GO TO 76
REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1031) (I,I=1,10)
77 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTTF,PTOTTI,IPRINT
IF(IPRINT(2).LE.0) GO TO 77
CALL PAGES(1,4,JP)
IF(JP.EQ.0) WRITE(6,1031) (I,I=1,10)
WRITE(6,1032)ISL,(TINTER(I),I=1,NTINTS)

```

AD-A057 907

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF  
ADAPTATION OF THE IMPROVED ANTI-AIRCRAFT ARTILLERY SIMULATION CO--ETC(U)  
MAR 78 C F SWENSON

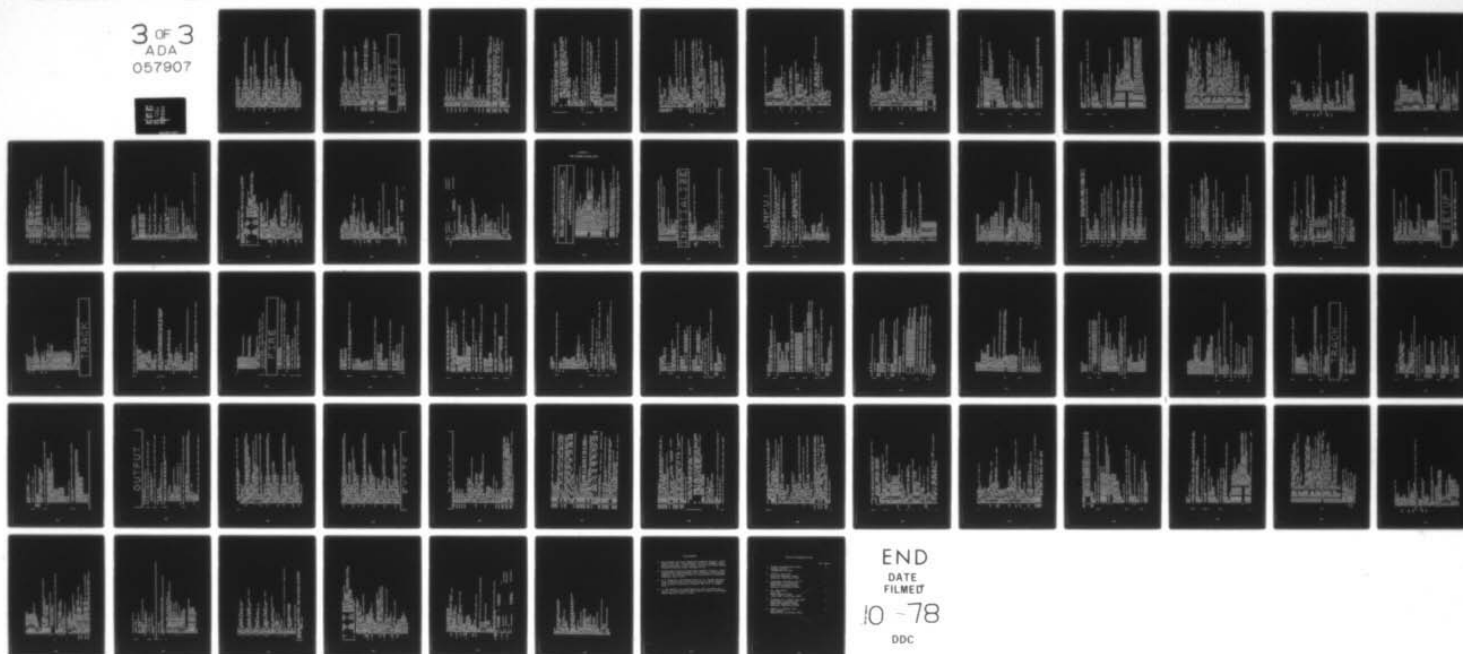
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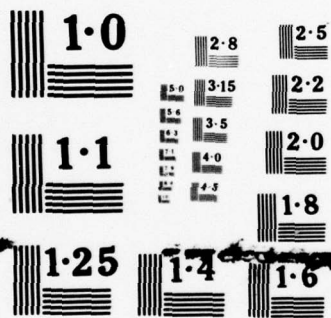
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NATIONAL BUREAU OF STANDARDS  
MICROCOPY RESOLUTION TEST CHART

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IFLAGS(2)=IFLAGS(2)-IPRINT(2)
IF(IFLAGS(2).GT.0)GO TO 77
IF(IFLAGS(3).LE.0)GO TO 42
76 REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1035) (I,I=1,10)
41 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF(IPRINT(3).LE.0)GO TO 41
CALL PAGES(2,4,JP)
IF(JP.EQ.0)WRITE(6,1035) (I,I=1,10)
WRITE(6,1018) ISL, (PLOTF(I),I=1,NTINTS)
DO 410 I=2,NTINTS
410 PLOTF(I)=PLOTF(I-1) + (1.0-PLOTF(I-1))*PLOTF(I)
WRITE(6,1058) ISL, (PLOTF(I),I=2,NTINTS)
IFLAGS(3)=IFLAGS(3)-IPRINT(3)
IF(IFLAGS(3).GT.0)GO TO 41
42 REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1036) (I,I=1,10)
43 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF(IPRINT(4).LE.0)GO TO 43
CALL PAGES(2,4,JP)
IF(JP.EQ.0)WRITE(6,1036) (I,I=1,10)
WRITE(6,1018) ISL, (PLOTTI(I),I=1,NTINTS)
DO 430 I=2,NTINTS
430 PLOTTI(I)=PLOTTI(I-1) + (1.0-PLOTTI(I-1))*PLOTTI(I)
WRITE(6,1058) ISL, (PLOTTI(I),I=2,NTINTS)
IFLAGS(4)=IFLAGS(4)-IPRINT(4)
IF(IFLAGS(4).GT.0)GO TO 43
71 REWIND 7
CALL PAGES(4,0,JP)
WRITE(6,1017) (I,I=1,9)
74 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTF,PLOTTI,IPRINT
IF(IPRINT(1).LE.0)GO TO 74
CALL PAGES(1,4,JP)
IF(JP.EQ.0)WRITE(6,1017) (I,I=1,9)
WRITE(6,1059) ISL, (RHO(I),I=1,NRHCS)
IFLAGS(1)=IFLAGS(1)-IPRINT(1)
IF(IFLAGS(1).GT.0)GO TO 74
44 CALL PAGES(7,NTINTS,0,JP)
WRITE(6,1015)
WRITE(6,1021) (I,I=1,9)
PLOTF(1) = PKTFDC(1,1)

```





```

CALL EXIT
ENTRY S9995 (TEMP, SPKT, SPKT2, *, *, *)
WRITE(6, 1039)
CALL EXIT
ENTRY S9996 (TEMP, SPKT, SPKT2, *, *, *)
WRITE(6, 1037)
CALL EXIT
ENTRY S9997 (TEMP, SPKT, SPKT2, *, *, *)
REWIND 4
LINE = 66
WRITE(6, 6060)
6060 FORMAT(//, 1X, '***** END OF JOB - - POOL SCENARIO RUN COMPLETE',
1 *****
RETURN 3
CALL PAGES(3, 0, JP)
6061 WRITE(6, 1061)
READ(4, 1062, END=991) ICARD, IPRINT, I
990 CALL PAGES(1, 3, JP)
IF (JP.EQ.0) WRITE(6, 1061)
WRITE(6, 1063) ICARD, IPRINT, I
IF (ISW.NE.1006) CALL AVG(ICARD(3), IPRINT, ISW)
GO TO 990
991 REWIND 4
CALL EXIT
RETURN 3
9998 WRITE(6, 1006) I, ICARD
CALL EXIT
9999 WRITE(6, 1002)
CALL EXIT
RETURN 3
1002 FORMAT(//, 'UNEXPECTED END-OF-RECORD/FILE ENCOUNTERED.')
1006 FORMAT(//, 'IMPROPER INPUT CARD ENCOUNTERED. ', I2, 19A4, A2, ' ')
1027 FORMAT(//, 'GUN TYPE', I2, ' ERROR MODE', I2, ' COMBINATION' INVAL
1 IC)
1037 FCMT(//, 'UNRECOVERABLE PARITY ERROR DETECTED. CALL EXIT.')
1039 FCMT(//, 'ECM SHOULD NOT BE SPECIFIED WHEN IEM IS 1 OR 2')
1040 FCMT(//, 'MULTIPATH TRACKING ERROR SPECIFIED WITH IEM=3 ONLY')
1061 FCMT(1H0, 4X, DATE, 7X, TIME, 4X, LCC, GT, EM, CB, SB, GL, 5X, P(K',
1 ILL) RDS XGUN YGUN ZGUN F TIME PAGE,/)
1062 FCMT(2A9, A3, 5A2, A10, A5, 2A7, 3A6)
1063 FCMT(2(2X, A9), 2X, A3, 5(2X, A2), 2X, A10, 1X, A5, 2(1X, A7), 3(2X, A6))
87 RETURN 1
55 RETURN 2
9992 GO TO IERR(3, 28, 9996)
9993 GO TO IEOF(4, 12, 40, 70, 83, 9997, 9999)
END

```

```

SUBROUTINE PAGES(N,NT,JP)
  KEEPS NUMBER OF LINES PER PAGE LESS THAN 59, PRINTS HEADER,
  AND GETS TIME INFORMATION FROM SYSTEM. REPLACES
  NPAGE(MAX) AND HEADER IN AFATL PROGRAM.

  N      NUMBER OF LINES TO BE PRINTED BEFORE NEXT CALL TO PAGES.
  NT     NUMBER OF LINES IN TITLE OR HEADER CF DATA BEING PRINTED.
  IF CALL TO PAGES IS TO PRINT HEADER ONLY, "N" SHOULD
  BE NUMBER OF LINES AND "NT" SHOULD BE ZERO.
  JP     FLAG FROM PAGES, SET TO ZERO WHEN A NEW PAGE IS STARTED
  INDICATING NECESSITY TO PRINT HEADER.

  COMMON /BLOCK1/ ITITLE(20)
  COMMON /BLOCK2/ LINE,NUMBER
  JP = 2
  LINE = LINE + N
  IF (LINE .LT. 59) RETURN
  NUMBER = NUMBER + 1
  WRITE (6,1000) ITITLE,NUMBER
  LINE = 2 + N + NT
  JP = 0
  RETURN
1000 FORMAT('1AFATL P-001 AAASIM ','17A4,',A10,2(1XA9),2X'PAGE',14/)
END
SUBROUTINE PRSEGS(P,ISL)
  PRINTS THE PK AS A FUNCTION OF ASPECT AND IMPACT SPEED TABLES

  THIS SUBROUTINE EXTENSIVELY MODIFIED TO PRINT ASPECT SECTOR
  ANGLES AND PROPERLY LABEL THE TWO CASES FOR WHICH IT
  PRINTS TABLES.
  COMMON /BLOCK3/ XGUN,YGUN,ZGUN
  COMMON /BLOCK4/ IGT,IEM,ICB,ISB,IGL,CIRCLE
  DIMENSION P(32,8),PT(8)
  REAL*8 ANG(8),315-360,000-045,045-090,090-135,135-180,
  180-225,225-270,270-315,
  8888 CONTINUE
  PK = 0.0
  PT(1) = 0.0
  PT(2) = 0.0
  PT(3) = 0.0
  PT(4) = 0.0
  PT(5) = 0.0
  PT(6) = 0.0
  PT(7) = 0.0
  PT(8) = 0.0
  111 CALL PAGES(60,JP)
  222 IF (ISL.GT.0) WRITE (6,1001) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE

```





```

C      MIN-MAX VALUES OF X, Y, AND Z (ZMIN = 0. BY DEFN.)
      FIND MIN-MAX VALUES OF X, Y, AND Z (ZMIN = 0. BY DEFN.)
      XMIN = 1.E99
      YMIN = 1.E99
      ZMIN = 1.E99
      XMAX = -1.E99
      YMAX = -1.E99
      ZMAX = -1.E99
      DO 100 I=1,NFPA,IFPA,IFPA
        XMIN = AMIN1(XMIN,XFPA(I))
        YMIN = AMIN1(YMIN,YFPA(I))
        ZMIN = AMIN1(ZMIN,ZFPA(I))
        XMAX = AMAX1(XMAX,XFPA(I))
        YMAX = AMAX1(YMAX,YFPA(I))
        ZMAX = AMAX1(ZMAX,ZFPA(I))
      C 100 CHECK X RANGE TO AVOID FUNNY X-Z PLOT.
      XCRNG = XMAX - XMIN
      IF (XCRNG.GT. 49.999) GO TO 110
      CALL PAGES(3,0,1)
      WRITE (6,210)
      RETURN
      C 110 SCALE X,Y TC METERS PER CHARACTER.
      D = AMAX1(XCRNG/11.,(YMAX-YMIN)/7.25)
      DX = D/10.
      DY = D/8.
      C 120 PLOT BY RE-CALCULATING X-Y MIN-MAX.
      XMIN = (XMAX+XMIN)/2. - 55.*DX
      YMIN = (YMAX+YMIN)/2. - 29.*DY
      XMAX = XMIN + 110.*DX
      YMAX = YMIN + 58.*DY
      C 130 SCALE Z (NOT TO SAME SCALE AS X-Y).
      DO 120 I=1,100
      D = FLOAT(I*500)
      IF (D.GT. ZMAX) GO TO 130
      CCNTINUE
      C 140 SEE IF AXES SHOULD APPEAR ON PLOT.
      XA = .FALSE.
      NYA = -10
      IF (XMIN.LE. 0. .AND. XMAX.GE. 0.) XA = .TRUE.
      IF (YMIN.LE. 0. .AND. YMAX.GE. 0.) NYA = .TRUE.
      IF (XA) NYA = 1 + INT(0.5-XMIN/DX)
      SET 8 LINES PER INCH SPACING ON PRINTER.
      WRITE (6,280)
      SET PAGE CONTROL, PLOT X-Z, THEN X-Y.
      CALL PAGES(56,0,1)
      WRITE (6,250)
      IFPA = 2 * IFPA
      DC 160 K=1,11
      DO 140 I=1,111

```

198

```

END
SUBROUTINE RPLANE(T)
  COMPUTES INFORMATION ABOUT THE POSITION OF THE AIRCRAFT AT TIME =T
  COMMON/BLOCK2/NFPA,TMIN,TMAX,DTFPA
  COMMON/BLOCK3/XGUN,YGUN,ZGUN
  COMMON/NFARM/XA,YA,ZA,RA2,RA,TIME
  COMMON/IGXGYG/IG,XG(8),YG(8)
  COMMON/XFPA(1201),YFPA(1201),ZFPA(1201),BFPA(1201),AFPA(1201),
  COMMON/PFPA(1201),VFPA(1201),VZFPA(1201)
1  COMMON/CECM1/IRECM,I,J,GAINJ,IX,XSEC,CALX,PJW,
  * X,Y,Z,ROL,PIT,PDG,
  * FGT,FJAM,GJ,SJT,SN
  CALL INTERP((T+TIME)/DTFPA)
  XA=GETVAL(XFPA)-XGUN-XG(IG)
  YA=GETVAL(YFPA)-YGUN-YG(IG)
  ZA=GETVAL(ZFPA)-ZGUN
  RA2=XA*XA+YA*YA+ZA*ZA
  RA=SQRT(RA2)
  RETURN
END
FUNCTION ANGLIM(X)
  LIMITS ANGLES TO PRINCIPAL ANGLES BETWEEN -PI AND +PI
  COMMON /CONSTS/ DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
  IF(ABS(X)-PI)1,1,2
1  ANGLIM=X
  2  ANGLIM=X-PI2*FLOAT(IFIX((X+SIGN(PI,X))/PI2))
  RETURN
END
FUNCTION RSHELL(T)
  COMPUTES RANGE TO SHELL AT TIME = T
  COMMON/VSASBS/VMUZZ,ASHCON,BSHCON,DQUAD
  DQUAD=1.0+T*(ASHCON+T*BSHCON)
  RSHELL=T*VMUZZ/DQUAD
  RETURN
END
FUNCTION VSHELL(T)
  COMPUTES SPEED OF SHELL AT TIME=T. CAN ONLY BE USED AFTER A CALL
  TO RSHELL AT THE SAME TIME, SINCE RSHELL COMPUTES DQUAD FOR VSHELL
  COMMON/VSASBS/VMUZZ,ASHCON,BSHCON,DQUAD

```



```

VSHELL=VMUZZ*(1.0-BSHCON*T*T)/(DQUAD*DQUAD)
RETURN
END
SUBROUTINE INTERP(FINT)
SETS CONSTANTS (FRACT, INDEX1, AND INDEX2) FOR TWO PCINT INTER-
POLATION
COMMON/MAGIC/FRACT,INDEX1,INDEX2
INDEX1=FINT
FRACT=FINT-FLOAT(INDEX1)
INDEX1=INDEX1+1
INDEX2=INDEX1+1
RETURN
END
FUNCTION GETVAL(ARRAY)
PERFORMS TWO POINT INTERPOLATION
COMMON/MAGIC/FRACT,INDEX1,INDEX2
DIMENSION ARRAY(1201)
GETVAL=ARRAY(INDEX1)+FRACT*(ARRAY(INDEX2)-ARRAY(INDEX1))
RETURN
END
BLOCK DATA
COMMON /BLOCK1/ ITITLE
COMMON /BLOCK3/ XGUN,YGUN,ZGUN
COMMON /BLOCK4/ IGT,IEM,IC8,ISB,IGL,CIRCLE
COMMON /BLOCK5/ NRHOS,RHO(9)
COMMON /BLOCK6/ NTINTS,TINTER(10)
COMMON /BLOCK7/ IVACOM,VAT(9,5,9)
COMMON /BLOCK8/ TREAT,TRACK1,TRACK2
COMMON /BLOCK9/ TROUND(6),THDMAX(6),PHDMAX(6),PHIMIN(6),PHIMAX(6),
VELMIN(6),VELMAX(6),RANMIN(6),RANMAX(6),ATLAG(6),
ETHMAX(6),EPHMAX(6),RMODES(6)
COMMON /BLOCKA/ TFMAX1(6),TFMAX2(6),RVACON(6),RVBCCN(6),VMUZEL(6)
COMMON /CONSTS/ DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
COMMON /HEADFO/ LINE,NUMBER
COMMON /BUDGET/ BCACON(6)
COMMON /NEWA/ VY1,VY,ALFA,J,IRMP,PLEN,O1,O8,D1,JJJ,XR,ZT,PZI,IERR
COMMON /NEWB/ Y1,P2,VX,BETA,IF6,IMUL,IP,N,O,C,C,TTRACK,KMODE,PSI,K
COMMON /NEWC/ CP,X2,P1,B2,FILE,IJAM,ISL,C4,CPS,TPERS,JMODE,YT,I2
COMMON /NEWD/ T33,X1,Z2,B1,VX2,C3,CPK,NROUND,IOFF,I,XI,TM,SP,TI
COMMON /NEWE/ Z1,A2,VX1,VZ2,PHI,IF5,IEOF,YR,IF2,TI3,IFILE,Y2,A1
COMMON /NEWF/ V2,VZ1,V,FUZZ,IF9,SC2RJ,SD2RJ,V1,VY2,F,J,P,IF7,REFC
COMMON /NEWG/ SJTMAX,O2,PK,D2,ISM
DATA IJAM/O,SD2RJ/O,
DATA IMUL/O,

```

DATA IOFF/2HOF/ IF2, IF5, IF6, IF7, IF9/1,1,1,1,1,1/  
 DATA VX2/0,0,0, VY2/0,0,0, VZ2/0,0,0, V2/0,0,0, P2/0,0,0, A2/0,0,0, B2/0,0,0/  
 DATA X2/0,0,0, Y2/0,0,0, Z2/0,0,0, FUZZ/0,0,0,000001/  
 DATA IFILE/9999/ JFILE/9999/ (BEE)  
 VALUES ON "ASIDOC" CARDS FROM A.S.I. DOCUMENTATION DRAFT.  
 DATA ATLAG /2\*9999.99,1.33,9999.99,2\*1.33/  
 DATA BDACON /002651,005011,0031,00697,00113,00113,00113/  
 DATA DEGREE /57.295779513082/ RADIANT/0.01745329251994/  
 DATA EPHMAX /6\*1/ ETHMAX /6\*1/  
 DATA IGT /3/ IEM/4/ ICB/1/ ISB/4/ IGL/1/ CIRCLE/0.0/  
 INTEGER ITITLE(20)/7\* . . . ASD/1, XROA, 11\* . . .  
 INTEGER IVACOM(20)/20\* . . .  
 DATA LINE /66/ NUMBER/0/  
 DATA NRHOS /9/  
 DATA NTINTS /10/  
 DATA PHDMAX /43633, 43633, 78540, 34910, 31416, 3491/  
 DATA PHIMAX /1.48353, 1.57079, 1.48353, 1.48353, 1.51844, 1.43117/  
 DATA PHIMIN /-1.17453, -1.14835, -1.17453, -0.8727, -0.6981, -0.5236/  
 DATA PI RP1 /3.1415926535898/ PI2/6.2831853071796/  
 DATA RANMAX /3000, 3000, 3300, 5000, 5500, 59999.9/  
 DATA RANMIN /0.0, 400, 500, 0.0, 0.59999.9/  
 DATA RHO /1.5, 3.333333333, 25.2, 1666666666, 1.02, .01/  
 DATA RMODES /-1.80209, 251499, 22958, 089321, 07845, 050694/  
 DATA RVACON /0.12392, -0.06259, -0.06889, 0.04262, -0.00421, -0.00357/  
 DATA SQR2 /1.4142135623731/  
 DATA SQR2 /1.4142135623731/  
 TFMAX CASE, BUT NOT EXACT VALUES FOR MAX RANGE. 13 FEB 73 (BEE)  
 DATA TFMAX1 /1.6, 2.2, 3.8, 4.1, 6.2, 59.99/  
 DATA TFMAX2 /99.99, 99.99, 7.5, 99.99, 11.6, 19.2/  
 DATA THDMAX /0.5236, 0.5236, 1.39626, 0.5236, 0.5236, 0.5236/  
 DATA TINTER /10, 20, 30, 40, 50, 60, 70, 80, 90, 599.99/  
 DATA TREAT /0.75, 4, 2, 75, 857, 4.7/  
 DATA TROUND /405\*0/  
 DATA VELMAX /300, 300, 350, 250, 300, 999.9/  
 DATA VELMIN /0, 0, 0, 0, 0, 999.9/  
 DATA VMUZEL /840, 1000, 930, 880, 960, 0, 800.0/  
 DATA XGUN /0.0/  
 DATA YGUN /0.0/  
 DATA ZGUN /0.0/  
 END  
 SUBROUTINE AVG(ITST, IPRINT, ISW)  
 DIMENSION IPRINT(6)  
 DATA IBLNK, SUM, CNT/2H , 0.0, 0.0, 0.0/  
 DATA IALL/2HAL/

```

          CALL REREAD
          READ (99,900)      PK
          FORMAT(F10.7)
          READ (99,910)      YGUN
          FORMAT(F7.0)
          IF(I$M.EQ.IALL) GOTO 600
          IF(IT$T.EQ.IBLNK) GOTO 200
          IF(YGUN.NE.0.0) GOTO 150
          CNT=CNT+0.5
          SUM=SUM+PK/2.
          POLD=0.0
          GOTO 500
150      CGNT INUE
          CNT=CNT+1.0
          SUM=SUM+PK
          POLD=PK
          GOTO 500
200      IF(CNT.LE.0.5) GOTO 500
          CNT=CNT-0.5
          SUM=SUM-POLD/2.0
          AVERG=SUM/CNT
          WRITE(6,800) CNT,AVERG
          FORMAT(15X,18H AVERAGE P(KILL) ON,F6.1,20H OFFSET LOCATIONS IS,
          X F10.7/)
          SUM=0.0
          CNT=0.0
          RETURN
500      IF(IT$T.NE.IBLNK) GOTO 610
          IF(CNT.NE.0.0) GOTO 201
          GOTO 500
610      SUM=SUM+PK
          CNT=CNT+1.0
          GOTO 500
          END
          SUBROUTINE JAMER1(PLEN,SDSQ)
          SDR = PLEN*0.6826/2.*2.998E8
          SDSQ = SDR*SDR
          RETURN
          END
          SUBROUTINE MULPTH(I,REFC,EL,BIAS,SD2)
          DIMENSION C(3),S(3),B(3)
          DATA
          * AK/-0.6931471806/
          * ,SQR2/1.414213562/
          * ,B/0.0244346,0.0314159,0.0785398/
          * ,C/0.0132557,0.0185345,0.0478558/
          * ,S/0.00872665,0.0104720,0.0244346/
          * BM = B(I)

```



```

CAL = C(I)
SQ = S(I)
DIR = EXP(AK*(SQ/BW)**2)
EL2 = 2*EL*(EL2+SQ)/BW)**2)
RL = EXP(AK*(EL2-SQ)/BW)**2)
DRUIS = (DIR+REFC*RU)**2
DRU2S = (DIR+REFC*RU)**2
DRL1S = (DIR+REFC*RL)**2
DRL2S = (DIR+REFC*RL)**2
DIF1 = (DRUIS-DRL1S)
SUM1 = (DRUIS+DRL1S)
DIF2 = (DRU2S-DRL2S)
SUM2 = (DRU2S+DRL2S)
SIGER1 = DIF1/SUM1
SIGER2 = DIF2/SUM2
ANGER1 = CAL*SIGER1
ANGER2 = CAL*SIGER2
PPBY2 = ABS(ANGER2-ANGER1)/2.
BIAS = ANGER1+PPBY2
SD = PPBY2/SQRT2
SD2 = SD*SD
RETURN
END
SUBROUTINE ECM1
DIMENSION RGDB(3), PRW(3), FREQ(3), IRTYP(4), RNOISE(3)
DIMENSION TABJ(37), TABX(37,37)
COMMON /BLOCK1/ ITITLE(10)
COMMON /HEADFO/ LINE, NUMBER
COMMON
COMMON /CECM1/ IREC, IJ, GAINJ, IX, XSEC, CALX, PJW,
X, Y, Z, ROL, PIT, HDG,
FTGT, FJAM, GJ, SJT, SN
**
**
NAMELIST /NAML/ RGDB, PRW, FREQ, IRTYP, I, RG, WL, FTGT, FJAM, PJW
DATA
RGDB/40.1, 38.5, 28./
PRW/10500., 17500., 25000./
FREQ/15.1E9, 9.3805E9, 2.838E9/
IRTYP/1, 2, 2, 3/
PI4/12.56637061/
RNOISE/-123.0, -130.6, -132.2/
C
I = IRTYP(IREC)
RG = 10.**((RGDB(I))/10.)
WL = 2.998E8/FREQ(I)
RN = RNOISE(I)
FTGT = PRW(I)*RG*RG*WL/PI4/PI4/PI4

```

```

FJAM = PJW*RG*WL*WL/PI4/PI4
CALL PAGES(3,0,JJP)
IF(IJ.EQ.0) WRITE(6,9003) GAINJ
9003 FORMAT(//, JAMMER ANTENNA GAIN',F7.3,' DB')
IF(IJ.NE.0) WRITE(6,9004)
9004 FORMAT(//, JAMMER TABLE SPECIFIED')
CALL PAGES(3,0,JJP)
IF(IX.EQ.0) WRITE(6,9005) XSEC
9005 FORMAT(//, AIRCRAFT CROSS SECTION ',F9.2,' SQ.METERS')
IF(IX.NE.0) WRITE(6,9006) CALX
9006 FORMAT(//, AIRCRAFT CROSS SECTION TABLE SPECIFIED.',F9.2)
* PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX=,F9.2)
IF(IJ.EQ.0) GO TO 1
C
C
C JAMMER TABLE
CALL TABLR(TABJ,37)
DO 3 I=1,37
DO 3 J=1,37
TABJ(I,J) = 10.**(TABJ(I,J)/10.)
3 CONTINUE
1 GJ = 10.**(GAINJ/10.)
IF(IX.EQ.0) GO TO 2
C
C
C X-SECTION TABLE
CALL TABLR(TABX,37)
2 RETURN
C
C
C
C
C ENTRY ECM2
NAMELIST/NAM2/ X,Y,Z,ROL,PIT,HDG,
*CX1,CY1,CZ1,CX2,CY2,CZ2,AZ,EL,GAINJ,XSEC,D2,SJ,ST,SJT
*,GJ
C
IF(IX.EQ.0 .AND. IJ.EQ.0) GO TO 5
CALL DIRCOS(X,Y,Z,0.,0.,CX1,CY1,CZ1)
CALL CARROT(CX1,CY1,CZ1,ROL,PIT,HDG,CX2,CY2,CZ2)
CALL RECSPH(CX2,CY2,CZ2,AZ,EL)
IF(IJ.EQ.0) GO TO 6
CALL INTRP(TABJ,AZ,EL,37,GJ)
6 IF(IX.EQ.0) GO TO 5
CALL INTRP(TABX,AZ,EL,37,XSEC)
XSEC = XSEC*CALX
5 D2 = DIST2(X,Y,Z,0.,0.,0.)
SJ = FJAM*GJ/D2

```

```

ST = FTGT*XSEC/D2/D2
SN = 10.*ALOG10(ST)-RN
SJT = 10.*ALOG10(SJ/ST)
RETURN
END
SUBROUTINE JAMER2(IRAD,AJS,SDSQ)
DIMENSION AJS1(3),SD1(3)
DIMENSION AJS2(4),SD2(4)
DIMENSION AJS3(2),SD3(2)
DIMENSION AJS4(4),SD4(4)
DATA
* N1/3/ -5.16, 9./
* AJS1/ -5.16, 9./
* SD1/ 0.002963, 0.01185, 0.1374 /
DATA
* N2/4/ -2.16, 10.16, 16./
* AJS2/ -2.16, 10.16, 16./
* SD2/ 0.002963, 0.009877, 0.08278, 0.1374 /
DATA
* N3/2/ 15.30./
* AJS3/ 15.30./
* SD3/ 0.0018, 0.01441 /
DATA
* N4/4/ -2.16, 10.16, 16./
* AJS4/ -2.16, 10.16, 16./
* SD4/ 0.006914, 0.02173, 0.1024, 0.1374 /
GO TO (1,2,3,4),IRAD
1 CALL INT2(N1,AJS1,SD1,AJS,SD)
GO TO 5
2 CALL INT2(N2,AJS2,SD2,AJS,SD)
GO TO 5
3 CALL INT2(N3,AJS3,SD3,AJS,SD)
GO TO 5
4 CALL INT2(N4,AJS4,SD4,AJS,SD)
5 SDSQ = SD*SD
RETURN
END
SUBROUTINE INT2(NVAL,X,Y,XVAL,YVAL)
DIMENSION X(NVAL),Y(NVAL)
YVAL = Y(1)
IF(X(1)-XVAL) 4,4,3
4 DC 1 I=1,NVAL
IF(X(I)-XVAL) 1,1,2
1 CONTINUE
YVAL = Y(NVAL)
GO TO 3
2 YVAL = Y(I-1) + (Y(I)-Y(I-1)) / (X(I)-X(I-1)) * (XVAL-X(I-1))
3 RETURN

```



```

C SUBROUTINE TABLR(TABX,IDIM)
SUBROUTINE TO READ AND PRINT A TABLE CF UP TO 37 X 37 ELEMENTS
THE PROGRAM PROVIDES A DEFAULT VALUE FOR ELEMENTS OUTSIDE THE
DEFINED TABLE.
CARD VARIABLE FORMAT DEFINITION IDENTIFICATION
1 INAME 8A10 NO. OF AZ ELEMENTS
2 NAME I5 (ASSUMING ELEMENTS 1 CORRESPONDS TO AZ=0)
NEL I5 NO. OF EL ELEMENTS, EL GCES-ELEND TO +ELEND
ELENF F10.4 MAXIMUM ENTRY EL(DEG)
AZEND F10.4 MAXIMUM ENTRY AZ(DEG)
DEFAULT F10.4 DEFAULT VALUE
3+ TABX(IDIM,1) 8F10.0 DATA TABLE
COMMON/TABLES/ELO,DELAZ,DELEL,JEL
DIMENSION TABX(IDIM,IDIM),INAME(8)
DATA LE,LZ/2HEL,2HAZ/
DATA AZO,COTR/O.,.0174533/
READ(5,98) INAME
READ(5,99) NAZ,NEL,ELEND,AZENF,DEFAULT
98 FCFORMAT(8A10)
99 FORMATT(2I5,3F10.4)
NOTE IMPLIED INCREMENT
DELAZ=AZENF/(NAZ-1)
DELEL=(2.*ELEND)/(NEL-1)
LOCATE FIRST ELEVATION ENTRY ETC
JEL=(IDIM-NEL)/2 +1
MEL=(JEL+NEL-1)
ELO=-ELEND
WRITE(6,IOL) INAME,NAZ,AZO,AZENF,DELAZ,
1 LZ,NEL,ELO,ELEND,DELEL,LE,DEFAULT
101 FCFORMAT(I,1) TABLE DATA,/,IX,8A10,/,2(IX,I5;
1,1) ELEMENTS FROM ,F10.2, TO ,F10.2, BY ;
2F10.2,2XA2,/,, ELSEWHERE TABLE IS ,F10.2}
C INSERT DEFAULT
MAZ=(180./DELAZ)+1
IF ((MAZ.GT.37).OR.(MEL.GT.37)) GO TO 999
DC 8 IAZ=1,MAZ
DO 8 IEL=1,IDIM
8 TABX(I,IAZ)=DEFAULT
C READ TABLE
DC 14 I=JEL,MEL
READ(5,I02) (TABX(I,J),J=1,NAZ)
14 CONTINUE
102 FORMAT(8F10.0)

```

```

ICELAZ=DELAZ
NPAGE=NAZ/13+1
DO 20 LP=1,NPAGE
  JH1=(LP-1)*13+IDELAZ
  JH2=JH1+12*IDELAZ
  KH1=(LP-1)*13+1
  KP2=KH1+12
  IF(LP.EQ.NPAGE)KH2=MAZ
  IF(LP.EC.NPAGE)JH2=180
  WRITE(6,106)((JH,JH=JH1,JH2,DELAZ),,%,ELEV',1319)
106 FORMAT(11,RCS MATRIX',,%,ELEV',1319)
  JELO=MAXO(JEL-1,1)
  MELO=MINO(MEL+1,1DIM)
  ELPT=ELO-DELEL#{JEL-JELO)
  ELPT=TABLE
  C PRINT TABLE
  DO 15 J=JEL,MELO
    WRITE(6,104)ELPT,(TABX(J,K),K=KF1,KH2)
104 FORMAT(1X,F7.1,2X,13F5.2)
    ELPT=ELPT+DELEL
  15 CONTINUE
  20 CONTINUE
  C CONVERT TO RADIANS
  DELEL=DELEL*CDTR
  DELAZ=DELAZ*CDTR
  ELG=ELO*CDTR
  RETURN
  999 CONTINUE
  105 WRITE(6,105)
  105 FORMAT(10X,105)
  STOP
  ERRCR IN INPUT <<<<'>
  END
  SUBROUTINE DIRCOS(X1,Y1,Z1,X2,Y2,Z2,COSA,COSB,COSG)
  XC = X2-X1
  YD = Y2-Y1
  ZD = Z2-Z1
  D = SQRT(XD*XD+YD*YD+ZD*ZD)
  COSA = XD/D
  COSB = YD/D
  COSG = ZD/D
  RETURN
  END
  SUBROUTINE CARROT(X1,Y1,Z1,ROL,PIT,HDG,X2,Y2,Z2)
  C CARROT
  C CARROT CARTESIAN ROTATION
  X1=COS(HDG) + Y1*SIN(PDG)
  Y1=-X1*SIN(HDG) + Y1*COS(HDG)
  Z1 =
  C PITCH

```

```

      XX = X*COS(PIT)
      YY = Y
      ZZ = X*SIN(PIT) + Y
      C ROLL
      X2 = XX
      Y2 = -YY*SIN(ROL) + YY*COS(ROL)
      Z2 = ZZ*SIN(ROL) + ZZ*COS(ROL)
      RETURN
    END
  SUBROUTINE RECSPH(X,Y,Z,PHI,THE)
    THE = ARCOS(Z/SQRT(X*X+Y*Y+Z*Z))
    THE=THE-1.5708
    PHI=0.0
    SB=SQRT(X*X+Y*Y)
    IF(SB.NE.C.0)PHI=X/SB
    PHI=ARCOS(PHI)
    PHI=ABS(PHI)
    RETURN
  END
  SUBROUTINE INTRP(TAB,AZ,EL,NVAL,VALUE)
    COMMON/TABLES/ELO,DELAZ,DELEL,JEL
    DIMENSION TAB (NVAL,NVAL)
    A=ABS(AZ)
    E=EL
    AAZ=A/DELAZ+1.
    IAZ=AAZ
    IEL=(E-ELO)/DELEL+JEL
    IEL=EEL
    IAZ=MIN0(MAX0(IAZ,1),36)
    IEL=MIN0(MAX0(IEI,1),36)
    V1=TAB(IEI,IAZ)
    V2=TAB(IEI+1,IAZ)
    V3=TAB(IEI+1,IAZ+1)
    V4=TAB(IEI+1,IAZ+1)
    S=AAZ-IAZ
    V12=V1+(V2-V1)*S
    V34=V3+(V4-V3)*S
    S=EEL-IEI
    VALUE=V12+(V34-V12)*S
    RETURN
  END
  FUNCTION DIST2(X1,Y1,Z1,X2,Y2,Z2)
    XD = X2-X1
    YD = Y2-Y1
    ZD = Z2-Z1
    DIST2 = XD*XD+YD*YD+ZD*ZD
    RETURN
  END

```



# APPENDIX J

## P001 PROGRAM LISTING (CDC)

```

PROGRAM P7022(INPUT,OUTPUT,TAPE5=INPUT,TAPE6=OUTPUT,TAPE7,TAPE8,
1 TAPE9,TAPE4,TAPE11)
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
AAASIM -- MULTIPLE GUN ANTI-AIRCRAFT ARTILLERY SIMULATION
WRITTEN BY THOMAS D. MCMURCHIE AND JAMES O. SEVERSON
AIRFORCE ARMAMENT LABORATORY (AFATL-DLYS) EGLIN AFB, FLORIDA
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
PROGRAM RECEIVED 25 OCT 72
CONVERTED AND MAINTAINED BY ASD/XROA
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
COMMON/BLOCK1/ITITLE(10)
COMMON/BLOCK2/NFPA,TMIN,TMAX,DTFPA
COMMON/BLOCK3/XGUN,YGUN,ZGUN
COMMON/BLOCK4/IGT,IEM,ICB,ISB,IGL,CIRCLE
COMMON/BLOCK5/NRHCS,RHO(9)
COMMON/BLOCK6/NTINTS,TINTER(10)
COMMON/BLOCK7/IVACOM(8),VAT(9,5,9)
COMMON/BLOCK8/TREACT,TRACK1,TRACK2
COMMON/BLOCK9/TROUND(6),THDMAX(6),PHIMIN(6),PHIMAX(6),
1 VELMIN(6),VELMAX(6),RANMIN(6),RANMAX(6),
2 ATLAG(6),ETHMAX(6),EPHMAX(6),RMODES(6)
COMMON/BLOCK10/TFMAX1(6),TFMAX2(6),RVACON(6),RVBCON(6),VMUZEL(6)
COMMON/CONSTS/DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
COMMON/HEADFO/LINE,NUMBER
COMMON/NFPARM/XA,YA,ZA,RA2,RA,TIME
COMMON/IGXGYG/IG,XG(8),YG(8)
COMMON/MAGIC /FRACT,INDEX1,INDEX2
COMMON/VASASBS/VMUZZ,ASHCON,BSHCON,DQUAD
COMMON/BUDGET/BDACCN(6)
COMMON XFPA(1201),YFPA(1201),ZFPA(1201),BFPA(1201),AFPA(1201),
1 PFPA(1201),VXFPA(1201),VYFPA(1201),VZFPA(1201)
COMMON/CECM1/IRECM,I,J,GAINJ,IX,XSEC,CALX,PJW,
* X,Y,Z,ROL,PIT,HOG,
* FIGT,FJAN,GJ,SJT,SN
DIMENSION ICARD(8),TEMP(16,6),PTOTIE(10),PTCTTI(10)
DIMENSION PKTTDC(9),PKTIDC(10,9),PKTFDC(10,9)
DIMENSION SPKT(8,4,8),SPKT2(32,8),SPKTOT(32,8),IPRINT(6),IFLAGS(4)
EQUIVALENCE (SPKT2(1,1),SPKT(1,1,1),TEMP(1,1))
SCME EQUIVALENCES REMOVED BY B.E.E. BY CARRYING TWO VARIABLES
THROUGH PROGRAM WITH SAME VALUES FOR CLARITY.

```

C

```
DATA IFILE/9999/,JFILE/9999/,INUNIT/5/,FUZZ/0.000000001/
DATA ESVPCT/0.10/
DATA X2/0.0/,Y2/0.0/,Z2/0.0/,V2/0.0/,P2/0.0/,A2/0.0/,B2/0.0/
DATA VX2/0.0/,VY2/0.0/,VZ2/0.0/
DATA IF2,IF5,IF6,IF7,IF9,IFLAGS/1,1,1,1,1,0,0,0,0/
DATA IOFF/2HOF/
DATA IMUL/0/
DATA IJAM/0/,SD2RJ/0./,SD2J/0./
DATA ILOOP/-1/
CALL FTNBN(1,0,CPU)
CALL SECOND(CPU)
CALL TIMREM(CIO)
CALL DATE(TITLE(9))
ASSIGN 9996 TO IERR
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C IIIII N N IIIII TTTT IIIII A A L L IIII ZZZZ EEEE C
C I I I N N N I I I T T T A A L L I I ZZZZ EEE C
C I I I N N N I I I T T T A A A A L L I I ZZZZ EEE C
C IIIII N N IIIII T T T T T T T T T L L L L I I I I ZZZZ EEEE C
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
READ(INUNIT,1050) ISW,AMASK
AMASK1=AMASK*57.3
PRINT 38,AMASK1
FORMAT(5X,25HMASK ANGLE FOR THIS RUN =,F6.3,5H DEG.)
38 1050
95 ISL=1
I PAUL = 0
I JAM = 0
SD2RJ = 0
SD2J = 0
DC 96 J=1,9
PKTIDC(J)=0.0
DC 96 I=1,10
PKTIDC(I,J)=0.0
96 DC 94 I=1,32
PKTFDC(I,J)=0.0
94 DC 94 J=1,8
SPKTOT(I,J)=0.0
REWIND 7
ASSIGN 9997 TO IEOF
GU TO 15
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
C IIIII N N PPPP U U TTTT C
```





```

IF (EOF(9).NE.0.) GO TO IEOF,(4,12,40,70,83,5557,9999)
IF (IOCHEC(9).NE.0.) GO TO IERR,(3,28,9996)
GO TO 3
4 IF(JMODE.GT.IFILE)GO TO 2
  ASSIGN 28 TO IERR
  READ (9)
IF (EOF(9).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
IF (IOCHEC(9).NE.0.) GO TO IERR,(3,28,9996)
28 ASSIGN 9996 TO IERR
5 JMODE=6
  T1=T2
  X1=X2
  Y1=Y2
  Z1=Z2
  P1=P2
  A1=A2
  B1=B2
  V1=V2
  VX1=VX2
  VY1=VY2
  VZ1=VZ2
  IF(JMODE)6,6,7
6 READ(INUNIT,1000)T2,X,Y,Z,VX,VY,VZ2,B2,A2,P2
  IF (EOF(INUNIT).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
  IF (IOCHEC(INUNIT).NE.0.) GO TO IERR,(3,28,9996)
  B2=B2*RADIAN
  A2=A2*RADIAN
  P2=P2*RADIAN
  VZ=SQRT(VX*VX+VY*VY+VZ2*VZ2)
  GO TO 9
7 JMODE=JMODE+1
  IF(JMODE.LE.6)GO TO 8
  JMODE=1
  READ(9) TEMP
  IF (EOF(9).NE.0.) GO TO IEOF,(4,12,40,70,83,9597,9999)
  IF (IOCHEC(9).NE.0.) GO TO IERR,(3,28,9996)
8 T2=TEMP(1,JMODE)
  X=TEMP(2,JMODE)
  Y=TEMP(3,JMODE)
  Z=TEMP(4,JMODE)
  VX=TEMP(5,JMODE)
  VY=TEMP(6,JMODE)
  VZ2=TEMP(7,JMODE)
  B2=TEMP(11,JMODE)
  A2=TEMP(13,JMODE)
  P2=TEMP(15,JMODE)

```

```

9  X2=XT+(X-XR)*CP+(Y-YR)*SP
   Y2=YT+(Y-YR)*CP-(X-XR)*SP
   Z2=ZT+Z
   VX2=VX*CP+VY*SP
   VY2=VY*CP-VX*SP
   B2=ANGLIM(B2-PZ1)
   T2=T2-TMIN
10  IF(TM.GT.T2)GO TO 5
   F=(TM-T1)/(T2-T1)
   NFPA=NFPA+1
   XFPA(NFPA)=X1+F*(X2-X1)
   YFPA(NFPA)=Y1+F*(Y2-Y1)
   ZFPA(NFPA)=Z1+F*(Z2-Z1)
   VXFPA(NFPA)=VX1+F*(VX2-VX1)
   VYFPA(NFPA)=VY1+F*(VY2-VY1)
   VZFPA(NFPA)=VZ1+F*(VZ2-VZ1)
   V=V1+F*(V2-V1)
   BFPA(NFPA)=ANGLIM(B1+F*ANGLIM(B2-B1))
   AFPA(NFPA)=A1+F*(A2-A1)
   PFPA(NFPA)=ANGLIM(P1+F*ANGLIM(P2-P1))
   K=K-1
   IF(K)34,34,35
34  K=KMODE
   PHI=DEGREE*PFPA(NFPA)
   BETA=DEGREE*BFPA(NFPA)
   ALFA=DEGREE*AFPA(NFPA)
   CALL PAGES(1,2,JP)
   IF(JP.EQ.0)WRITE(6,1003)
   WRITE(6,1004)TM,XFPA(NFPA),YFPA(NFPA),ZFPA(NFPA),
1  V,VXFPA(NFPA),VYFPA(NFPA),VZFPA(NFPA),BETA,ALFA,PHI
35  TM=DTFPA*FLCAT(NFPA)
   IF(TM.LE.TMAX)GO TO 10
   TMAX=TM-DTFPA-FUZZ
   CALL PAGES(4,0,JP)
   WRITE(6,1005)XR,YR,XT,YT,PSI,ZT
   CALL TPLOT(NFPA)
   IF(JMODE.GT.0.OR. INUNIT.EQ.8)GO TO 12
   ASSIGN 12 TO IEOF
13  READ(INUNIT,1000)
   IF(EOF(INUNIT).NE.0.)GO TO IEOF,(4,12,40,70,83,9997,9999)
   IF(IOCHEC(INUNIT).NE.0.)GO TO IEAR,(3,28,5556)
   GO TO 13
C  DECODE DATA BLOCK 03 -- WEAPON LOCATION
C
C  103 DECODE(30,1008,ICARD)XGUN,YGUN,ZGUN
C  GO TO 12
C

```

```

C      CECODE DATA BLOCK 04 -- WEAPON TYPE, MODE, NUMBER OF BARRELS (CYCLIC
C      AND SIMULTANEOUS), NUMBER OF WEAPONS PER
C      LOCATION, RADIUS OF CIRCLE OF WEAPON COMPLEX
104  DECODE(14,1009,ICARD)IGT,IEM,ICB,ISB,IGL,CIRCLE
61  IF(IGL-1)9958,61,66
    XG(1)=0.0
    YG(1)=0.0
    GO TO 12
66  CO 16 I=1,IGL
    F=PI2*FLOAT(I)/FLOAT(IGL)
    XG(I)=CIRCLE*COS(F)
    YG(I)=CIRCLE*SIN(F)
    GO TO 12

C      CECODE DATA BLOCK 05 -- WEAPON DENSITY FACTORS
105  DECODE(78,1015,ICARD)IF5,NRHOS,(RPO(I),I=1,NRHOS)
    GO TO 12

C      CECODE DATA BLOCK 06 -- PK ACCRUAL TIME INTERVALS
106  DECODE(78,1015,ICARD)IF6,NTINTS,(TINTER(I),I=1,NTINTS)
    NTINTS=NTINTS+1
    TINTER(NTINTS)=999.99
    GO TO 12

C      CECODE DATA BLOCK 07 -- AIRCRAFT VULNERABLE AREAS
107  DC 11 I=1,8
11  IVACOM(I)=ICARD(I)
    READ(INUNIT,1000)(VAT(1,1,K),K=2,9)
    IF (EOF(INUNIT)).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
    IF (IOCHEC(INUNIT)).NE.0.) GO TO IERR,(3,28,5996)
    DC 88 J=2,4
    DO 89 I=1,8
    READ (INUNIT,1000) (VAT(I,J,K),K=2,9)
    IF (EOF(INUNIT)).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
    IF (IOCHEC(INUNIT)).NE.0.) GO TO IERR,(3,28,5996)
    CONTINUE
89  DC 88 K=2,9
88  VAT(9,J,K)=VAT(1,J,K)
    READ(INUNIT,1000)(VAT(1,5,K),K=2,9)
    IF (EOF(INUNIT)).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
    IF (IOCHEC(INUNIT)).NE.0.) GO TO IERR,(3,28,5996)
    DC 18 K=2,9
    DO 18 I=2,9
    VAT(1,5,K)=VAT(1,5,K)

```



```

18 VAT(I,1,K)=VAT(1,1,K)
   IF7=1
   GO TO 12
C
C   DECODE DATA BLOCK 08 -- WEAPON REACTION AND TRACK TIMES
C
108 DECODE(30,1008,ICARD)TREAT,TRACK1,TRACK2
   GO TO 12
C
C   DECODE DATA BLOCK 09 -- WEAPON PARAMETERS
C
109 DECODE(78,1008,ICARD)TROUND(IGT),THDMAX(IGT),PHDMAX(IGT),
      PHIMIN(IGT),PHIMAX(IGT),VELMIN(IGT),VELMAX(IGT),
      1  RANMIN(IGT),RANMAX(IGT)
      2  READ(INUNIT,1000)ATLAG(IGT),ETHMAX(IGT),EPHMAX(IGT),RMODES(IGT)
      IF (EOF(INUNIT).NE.0.) GO TO IEOF,(4,12,40,70,83,9997,9999)
      IF (IOCHEC(INUNIT).NE.0.) GO TO IERR,(3,28,5956)
      IF9=1
      GO TO 12
C
C   DECODE DATA BLOCK 10 -- SHELL PARAMETERS
C
110 DECODE(46,1008,ICARD)TFMAX1(IGT),TFMAX2(IGT),RVACON(IGT),
      1  RVBCON(IGT),VMUZEL(IGT)
      IF9=1
      GO TO 12
C
C   DECODE DATA BLOCK 11 -- INPUT OPTION (CARD/TAPE)
C
111 DECODE(4,1028,ICARD)I
   IF(1)81,81,82
   81 INUNIT=5
   82 INUNIT=8
   ASSIGN 83 TC IEOF
   IF(1.GT.JFILE)GO TO 84
   REWIND 8
   JFILE=1
   83 IF(1-JFILE)12,12,84
   84 JFILE=JFILE+1
   85 READ(8,1000)
   IF (EOF(8).NE.0.) GO TO 83
   IF (IOCHEC(8).NE.0.) GO TO IERR,(3,28,9996)
   GO TO 85
C
C   DECODE DATA BLOCK 13 --- LOW ALTITUDE RADAR MULTIPATH EFFECT
C
113 DECODE(18,1038,ICARD) IMUL,IRMP,REFC

```

```

IF(IMUL.EQ.0) GO TO 12
CALL PAGES(12,0,JP)
WRITE(6,1098) IRMP,REFC
1098 FORMAT(///,*,MULTIPATH INPUTS (INITIAL OR CHANGED *,//,
Q* IRMP = *,I5,/,
I* REFC = *,F6.3,///)
GO TO 12

C
C DECODE DATA BLOCK 14 --- ECM
C
114 SD2RJ = 0.
SC2RJ = 0.
DECODE(78,1041,ICARD) IJAM,IP,IJ,GAINJ,PJW,PLEN,IX,XSEC,CALX,
IRECM,SJTMAX
*
IF(IJAM.EQ.0) GO TO 12
CALL PAGES(28,0,JP)
WRITE(6,1096) IP,IJ,GAINJ,PJW,PLEN,IX,XSEC,CALX,IRECM,SJTMAX
1096 FORMAT(///,*,ECM INPUTS (INITIAL OR CHANGED) *,//,
Q* IP = *,I5,/,
I* IJ = *,I5,/,
1* GAINJ(DB) = *,F7.2,/,
3* PJW(W) = *,F9.2,/,
4* PLEN(S) = *,E12.6,/,
5* IX = *,I5,/,
6* XSEC(SQM) = *,F10.3,/,
7* CALX = *,F10.3,/,
8* IRECM = *,I5,/,
9* SJTMAX(DB) = *,F7.2,///)
CALL ECM1
IF(IRECM.EQ.3) CALL JAMER1(PLEN,SC2RJ,M)
IF(IEM.EQ.4) CALL JAMER1(PLEN,SD2RJ,M)
GO TO 12

C
C DECODE DATA BLOCK 12 -- PRINT OPTIONS FOR OUTPUT FORMAT - ALSO
C SIGNALS FOR RUN TO BEGIN
C
112 DECODE(6,1016,ICARD)IPRINT

C
C PRINT DATA BLOCKS 6, 7, 9, AND 10 (IF THEY CHANGE)
C "IF2" IS USED TO SET LINE COUNT TO PROPER VALUE. INPUT AND
C OUTPUT PRINT CUT START A NEW PAGE FOR EACH "12" CARD.
C
IF (IF2.EQ.0) LINE = 66
IF2=0
IF(IJAM.EQ.0) GO TO 48
IF(IP.EQ.0) GO TO 48
WRITE(11,1043) ITITLE
WRITE(11,1042)

```













```

XF=RC*CTBCPB-XG(IG)
YF=RC*STBCPB-YG(IG)
ZF=RC*SPB
GF=SQRT(XF*XF+YF*YF)
RF=SQRT(GF*GF+ZF*ZF)

      (SET UP MATRIX T, THE TRANSFORMATION BETWEEN THE LINE OF SIGHT
      SYSTEM AND THE FALSE HORIZON SYSTEM)

T22=CT=X/G
T1=Y/G
T33=CP=G/R
T13=SP=Z/R
CS=COS(PSI)
SS=SIN(PSI)
CA=VG/V
SA=VZ/V
CG=VX/VG
SG=VY/VG
T11=CT*CP
T12=ST*CP
T21=-ST
T31=-CT*SP
T32=-ST*SP

      (SET UP FALSE HORIZON SYSTEM VELOCITY COMPONENTS)

VXP=T11*VX+T12*VY+T13*VZ
VYP=T21*VX+T22*VY
VZP=T31*VX+T32*VY+T33*VZ
VGP=SQRT(VXP*VXP+VYP*VYP)
CAP=VGP/V
SAP=VZP/V
CBP=VXP/VGP
SBP=VYP/VGP

      (SET UP UNIT VECTOR OUT LEFT WING OF AIRCRAFT)

UX=-SA*CG*SS-SG*CS
UY=CG*CS-SA*SG*SS
UZ=CA*SS
UZP=T31*UX+T32*UY+T33*UZ
IF(VXP) 31, 32, 31
CSP=-T11*UX-T12*UY-T13*UZ
GC TO 33
31 CSP=(VGP*(T21*UX+T22*UY)+UZP*VZP*SBP)/VXP
33 SSP=UZP/CAP

```

```

C C
C C      ( COMPUTE FALSE HORIZON SYSTEM MEAN AND STANCRD DEVIATION CF
C C      ERROR IN DIVE AND COURSE ANGLE ESTIMATES)
C C
C C      EMAP=SAP*(.3196*ABS(CBP)-.1859*ABS(SBP))
C C      ESAP=.04712+.08063*ABS(SAP)*(1.0+1.16*ABS(CBP))
C C      EMBP=.4060*CAP*SBP*CBP
C C      ESBP=(.1670-.08098*ABS((CBP*CBP-SBP*SBP)*CSP)+
C C      1  .09006*ABS(SBP*SSP*CSP))/CAP
C C      SEMAP=$IN(EMAP)
C C      CEMAP=COS(EMAP)
C C      SEMBP=$IN(EMBP)
C C      CEMBP=COS(EMBP)
C C      THE NEXT CARD ELIMINATES AN EQUIVALENCE BY USING A DOUBLE
C C      THE REPLACEMENT.
C C      A33=SABP=SAP*CEMAP+CAP*SEMAP
C C      CABP=CAP*CEMAP-SAP*SEMAP
C C      SBBP=SBP*CEMBP+CBP*SEMBP
C C      CBBP=CBP*CEMBP-SBP*SEMB
C C
C C      (LIMIT VELOCITY ASSESSMENT (MECHANICAL COMPUTERS))
C C
C C      VBP=AMIN1(VMAX,AMAX1(VMIN,V))
C C      ESVP=ESVPCT*V
C C
C C      SET UP THE ELEMENTS CF THE MATRIX A.
C C      A CONTAINS THE PARTIALS OF VXE,VYE,VZE W.R.T. ALPHA,EETA,SPEED
C C
C C      A31=CABP*CBBP
C C      A32=CABP*SBBP
C C      A21=-VBP*A32
C C      A22=VBP*A31
C C      A11=-VBP*SABP*CBBP
C C      A12=-VBP*SABP*SBBP
C C      A13=VBP*CABP
C C
C C      CCMPUTE MEAN ESTIMATED VELOCITY CCMPONENTS (MECHANICAL
C C      COMPUTATION)
C C
C C      VXE=(A31*T11+A32*T21+A33*T31)*VBP
C C      VYE=(A31*T12+A32*T22+A33*T32)*VBP
C C      VZE=(A31*T13
C C      +A33*T33)*VBP
C C      GO TO 63
C C
C C      CCMPUTE MEAN ASSUMED TIME OF FIRE AIRCRAFT POSITION (ELECTRONIC
C C      COMPUTATION)
C C
C C      56 XF=RANS*CTBCPB
C C      YF=RANS*STBCPB

```

```

C C C C
ITERATION TO DETERMINE MEAN THEORETICAL INTERCEPT POINT
63 RS=0.0
VS=VMUZZ
T=0.0
21 XE=XF+VXE*T
YE=YF+VYE*T
ZE=ZF+VZE*T
XE2=XE+XE
YE2=YE+YE
ZE2=ZE+ZE
GE2=GE2+YE2
RE2=GE2+ZE2
RE=SQRT(RE2)
RC=RE-RS
IF(RC.LT.1.C)GO TO 22
VD=VS-(XE+VXE+YE+VYE+ZE+VZE)/RE
IF(VD.LE.1.C)GO TO 64
T=T+RC/VD
IF(T.GT.TTFMAX)GO TO 64
RS=RSHELL(T)
VS=VSHELL(T)
GC TO 21
22 T2=T*T
CHANGE 22 JAN 76
IF INTERCEPT POINT BELOW MASK*****SKIP FIRE
GE=SQRT(GE2)
IF(ATAN2(ZE,GE).LE.AMASK) GO TO 64
ITERATION TO DETERMINE ACTUAL INTERCEPT POSITION, RANGE
TU=AMIN1(TFMAX,TMAX-TIME)
23 CALL RPLANE(TU)
IF(RSHELL(TU).GT.RA)GO TO 24
(SKIP FIRE ATTEMPT IF SHELL CANNCT CATCH AIRCRAFT)
IF((XA*GETVAL(VXFPA))+YA*GETVAL(VYFPA)+ZA*GETVAL(VZFP
1 VSHELL(TU))GO TO 64
TU=TU-1.0
IF(TU)64,64,23
24 TL=0.0
C C C C
C C C C
C C C C
C C C C

```



```

25 T=0.5*(TL+TU)
   CALL RPLANE(T)
   RC=RA-RSHELL(T)
   IF(RC.GT.1.0)GO TO 26
   TU=T
   GC TO 25
26 TL=T
   GC TO 25
37 Q0=VS*RE-XE*VXE-YE*VYE-ZE*VZE
   Q1=(VZE-VS*ZE/RE)/Q0
   Q2=(XE*VXE-YE*VYE)/Q0
   COMPUTE THE PARTIAL DERIVATIVES OF BIG THETA
   DTIDX=Q2*XE-YE
   DTIDY=Q2*YE+XE
   DTIDZ=Q2*ZE
   DIIDR=(ZF*DTIDZ+ YF*DTIDY+XF*DTIDX)/RF
   DIIDT= XE*DTIDY-YF*DTIDX
   DIIDP= GF*DTIDZ-(YF*DTIDY+XF*DTIDX)*ZF/GF
   COMPUTE THE PARTIAL DERIVATIVES OF BIG PHI
   DPPDX=Q1*XE
   DPPDY=Q1*YE
   DPPDZ=Q1*ZE+1.0
   DPPDR=(ZF*DPPDZ+ YF*DPPDY+XF*DPPDX)/RF
   DPPDT= XE*DPPDY-YF*DPPDX
   DPPDP= GF*DPPDZ-(YF*DPPDY+XF*DPPDX)*ZF/GF
   GE4=GE2*GE2
   GC TO (210,220,230,240),IOEM
   COMPUTATIONS FOR MODE 1 OPERATION
210 GC TO (211,212,213,214,215,999),IGT
   TRACKING ERROR DISTRIBUTION SIZES (SPHERICAL COORDINATES)
   (FOR GT 1, 2, OR 3)
211 CCNTINUE
212 CCNTINUE
213 SR2=(123.0+0.0225*R)**2
   ST2=(.0643*TD)**2
   SP2=(.1320*PD)**2
   GC TO 219
   (FOR GT 4 AND 5)

```

```

C 214 CONTINUE
215 SR2=(123.0+0.0225*R)**2
    ST2=(0.0167-.000710/(.0517+ABS(TC1)))*2
    SP2=(0.0116-.000216/(.0235+ABS(ABS(PD)-4.0*PDD)))*2
C
C SET UP THE ELEMENTS OF THE MATRIX E=AT
C
C 219 B11=A11*T11+A12*T21+A13*T31
    B12=A11*T12+A12*T22+A13*T32
    B13=A11*T13+A12*T23+A13*T33
    B21=A21*T11+A22*T21
    B22=A21*T12+A22*T22
    B23=A21*T13+A22*T23
    B31=A31*T11+A32*T21+A33*T31
    B32=A31*T12+A32*T22+A33*T32
    B33=A31*T13+A32*T23+A33*T33
C
C COMPUTE THE PARTIAL DERIVATIVES OF BIG, THETA AND BIG PHI W.R.T.
C ON CARRIAGE INPUTS (ALPHA, BETA, SPEED)
C
C DTDAP=B11*DTDX+B12*DTDY+B13*DTIDZ
C DTDBP=B21*DTDX+B22*DTDY+B23*DTIDZ
C CTTDVP=B31*DTDX+B32*DTDY+B33*DTIDZ
C DFPDAP=B11*CPPDX+B12*CPPDY+B13*CPPDZ
C DFPDBP=B21*CPPDX+B22*CPPDY+B23*CPPDZ
C DPPDVP=B31*CPPDX+B32*CPPDY+B33*CPPDZ
C
C COMPUTE THE VARIANCES OF BIG, THETA AND BIG PHI
C
C STT2=((DTIDR**2)*SR2+((DTIDT**2)*ST2+((DTIDP**2)*SP2
1  +((DTIDAP**2)*SAP)**2+((DTIDBP**2)*SBP)**2+((DTIDVP**2)*SVP)**2)/GE4
C SPP2=((DPPDR**2)*SR2+((DPPDT**2)*ST2+((DPPDP**2)*SP2
1  +((DPPDAP**2)*SAP)**2+((DPPDBP**2)*SBP)**2+((DPPDVP**2)*SVP)**2)/GE2
C GO TO 29
C
C COMPUTATIONS FOR MODE 2, 3, OR 4 OPERATION
C
C 220 GO TO (999,999,223,999,225,226),IGT
C
C TRACKING ERROR DISTRIBUTIONS SIZES (SPHERICAL COORDINATES)
C (FOR MODE 2)
C
C 223 CONTINUE
225 CCNTINUE
226 SR2=(41.0+0.0075*R)**2
    ST2=(.000982+.1681*TD*TD)**2
    SP2=(.000491+.033*ABS(ABS(PD)-4.0*PDD))**2

```

```

230 GG TO 65
231 GG TO (999,999,233,999,235,236),IGT
232 (FOR MODE 3)
233 CONTINUE
234 CCNTINUE
235 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SD2RJ
236 ST2=(0.00196+0.050*TD)**2
SP2=(0.000982+0.11*ABS(ABS(PD)-2.0*PDD))**2+SP2MP
GG TO 65
240 GG TO (999,999,243,999,245,246),IGT
241 (FOR MODE 4)
242 CCNTINUE
243 CONTINUE
244 SR2=(17.0+0.24*ABS(RDD)+0.018*RDD*RDD)**2+SD2RJ
245 ST2=(0.000982+0.1681*TD*TD)**2
SP2=(0.000491+0.033*ABS(ABS(PD)-4.0*PDD))**2+SP2MP
65 R8TD=RANS*THESD
R8PD=RANS*PHISD
VELOCITY COMPONENT ERROR DISTRIBUTION SIZES
SVX2=(SR2*(PHISD*CTBSPB+THESD*STBCPB)**2
1 +ST2*(R8PD*STBSPB-R8TD*CTBSPB-RANS*STBCPB)**2
2 +SP2*(R8PD*CTBSPB-R8TD*STBSPB+RANS*CTBSPB)**2)*ATLCON
SVY2=(SR2*(PHISD*CTBSPB-THESD*CTBCPB)**2
1 +ST2*(R8PD*CTBSPB-R8TD*STBCPB-RANS*CTBCPB)**2
2 +SP2*(R8PD*STBCPB-R8TD*CTBSPB+RANS*CTBCPB)**2)*ATLCON
SVZ2=(SR2*(PHISD*CPB)**2+SP2*(R8PD*SPB-RANS*CPB)**2)*ATLCON
CCOMPUTE THE VARIANCES OF BIG THETA AND BIG PHI
STT2=((DTTOR**2)*SR2 + (DTTD**2)*ST2 + (DTTDP**2)*SP2
1 + ((DTDX**2)*SVX2 + (DTDY**2)*SVY2 + (DTDZ**2)*SVZ2)*T2)/GE4
1+SD2J
SPP2=((DPPDR**2)*SR2 + (DPPDT**2)*ST2 + (DPPDP**2)*SP2
1 + ((DPPDX**2)*SVX2 + (DPPDY**2)*SVY2 + (DPPDZ**2)*SVZ2)*T2)/GE2
1+SD2J
CCOMPUTATION OF VULNERABLE AREA OF AIRCRAFT AT INTERCEPT
29 VP=VSHELL(T)
XU=XE/RE
YU=YE/RE

```



```

ZU=ZE/RE
XE=XU*RA
YE=YU*RA
ZE=ZU*RA
GE=SQRT(XE*XE+YE*YE)
VXA=GETVAL(VXFPA)
VYA=GETVAL(VYFPA)
VZA=GETVAL(VZFPA)
VGA=SQRT(VXA*VXA+VYA*VYA)
VA=SQRT(VGA*VGA+VZA*VZA)
VXI=VP*XU-VYA
VYI=VP*YU-VZA
VI=SQRT(VXI*VXI+VYI*VYI+VZI*VZI)
ALFA=GETVAL(AFPA)
CA=COS(ALFA)
SA=SIN(ALFA)
BETA=BFPA(INDEX1)+FRACT*ANGLIM(BFPA(INDEX2)-BFPA(INDEX1))
CG=COS(BETA)
SG=SIN(BETA)
PZI=PFPA(INDEX1)+FRACT*ANGLIM(PFPA(INDEX2)-PFPA(INDEX1))
THE NEXT CARDS ELIMINATE AN EQUIVALENCE BY USING A DOUBLE
    THE REPLACEMENT.
T33=CP=COS(PZI)
T13=SP=SIN(PZI)
Q1=VXI*CG+VYI*SG
Q2=VZI*CG-Q1*SA
Q3=VYI*CG-VXI*SG
VXF=Q1*CA+VZI*SA
VYF=Q2*SP+Q3*CP
VZF=Q2*CP-Q3*SP
(SET UP INDICES FOR VULNERABLE AREA INTERPOLATION)
F1=ATAN2(VYF,VXF)/QTRPI
IF(F1-LT.0.0)F1=F1+8.0
I1=F1
F1=F1-FLOAT(I1)
I1=I1+1
F2=ACOS(VZF/VI)/QTRPI
I2=F2
F2=F2-FLOAT(I2)
I2=I2+1
F3=AMIN1(7.599999999,VI/152.4)
I3=F3
F3=F3-FLOAT(I3)
I3=I3+1
DI=1.0-F1

```

C C

C C C



```

VAM2=VAM*VAM
VAP2=VAP*VAP
VAI2=VAM2+VAP2
CD2=VAM2/VAI2
SXA12=SXA2+CD2*SXL2+SD2*SYL2
SYAT2=SYA2+CD2*SYL2+SD2*SXL2
TWOCOV=2.0*VAM*VAP*(SXL2-SYL2)/VAI2
DIF=SXA12-SYAT2
DEN=2.0*SQRT(TWOCOV*TWOCOV+DIF*DIF)
HC2Z=DIF/DEN
CZ2=0.5+HC2Z
SZ2=0.5-HC2Z
SZCZ=TWOCOV/DEN
STUFF=2.0*SZCZ*BXA*BYA
BXF2=CZ2*BXA2+SZ2*BYA2+STUFF
BYF2=CZ2*BYA2+SZ2*BXA2-STUFF
SYF2=CZ2*SXA2+SZ2*SYA2+SZCZ*TWOCOV
AVTPI=AVT/PI2

C
C
C      COMPUTE PROBABILITY OF KILL
      STUFF=BXF2/(SXF2+AVTPI)+BYF2/(SYF2+AVTPI)
      IF(STUFF.LT.50.0)GO TO 75
      PK=0.0
      GO TO 78
75  PK=AMINI(1.0,EXP(-.5*STUFF)*AVTPI/SQRT((SXF2+AVTPI)*(SYF2+AVTPI)))
78  PS=(1.0-PK)*ISB
      PK=1.0-PS
      TI=TIME+T

C
C
C      ACCUMULATE PK AS A FUNCTION OF INPUT TIME INTERVALS
      I=0
      I=I+1
50  IF(TIME-GE.TINTER(I))GO TO 50
      J=1
51  IF(TI-LT.TINTER(J))GC TO 52
      J=J+1
      GO TO 51
52  PTOITF(I)=PK+PS*PTOITF(I)
      PTOITI(J)=PK+PS*PTOITI(J)
      CPS=CPS*PS

C
C
C      ACCUMULATE PK FOR EACH SPHERICAL SECTOR
      SPKT(I1,I2,I3)=PK+PS*SPKT(I1,I2,I3)

```



```

CC
CC      CCMPUTE QUANTITIES FOR EXTENDED OUPUT,  WHEN DESIRED
      IF(IPRINT(6).LE.0)GO TO 20
      C1=THESD*DEGREE
      C2=PHISD*DEGREE
      C3=ETHI4*DEGREE
      C4=EPHI4*DEGREE
      C5=SQRT(BXF2+FUZZ)
      C6=SQRT(BYF2+FUZZ)
      C7=SQRT(SXF2)
      C8=SQRT(SYF2)
      C9=1.0-CPS

      WRITE EXTENDED OUTPUT

      CALL PAGES(1,5,JP)
      IF (JP.EQ.0) WRITE (6,1013) ISL,IGT,IEM,XGUN,YGUN,ZGUN,CIRCLE
      WRITE(6,1014)IG,IEM,TIME,T,II,R,RA,07,08,05,06,VI,01,02,03,04,
      1  AVI,PK,09
      20 NROUND=NROUND+ISB

      FIRE ADDITIONAL GUNS IN COMPLEX, IF ANY

      64 IG=IG+1
      IF(IG.GT.IGL)IG=1
      IF(FIRE.TFIRE+TPERS
      IF(TIME.GT.TFIRE)GO TO 54
      CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
      TTTT  RRRR  R  A  A  A  C  C  C  K  K  K
      T  F  RRRR  A  A  A  C  C  C  K  K  K
      T  R  R  R  A  A  A  C  C  C  K  K  K
      T  R  R  A  A  A  C  C  C  K  K  K
      CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
      62 TIME=TIME+0.064
      62 IF(TIME.GT.TMAX)GO TO 69

      SWITCH TO MCDE 1 TRACKING IF JAMMING IS ABOVE THRESHOLD OR IF
      RANGE IS TOO CLOSE

      IDEM = IEM
      SD2J = 0.
      SC2RJ = 0.
      IF(R.LT.RSMCDE) GO TO 501
      IF(IJAM.EQ.0) GO TO 502

```



```

PHISD=PD+6.*C*EPIH
RANS D = 0.804*RD + 3.0*ERAN
GO TO 73

(MODE 4, GT 3, 5, AND 6)

GO TO (999,999,343,959,345,346),IGT
CONTINUE
CONTINUE
THESD=0.910*TD+0.45*TD+6.0*EITH
PHISD=0.75*PD-0.25*PDD+6.0*EPIH
RANS D = 0.804*RD + 3.0*ERAN

(LIMIT SLEW RATES AND ELEVATION ANGLE TO WEAPCN MAXIMUMS)

73 THESD=SIGN(AMIN1(TDMAX,ABS(THESD)),THESD)
PHISD=SIGN(AMIN1(PDMAX,ABS(PHISD)),PHISD)
THES=ANGLIM(THES+0.064*THESD)
PHIS=AMAX1(PHMIN,AMIN1(PHMAX,PHIS+0.064*PHISD))
RANS=AMAX1(0.0,RANS+0.064*RANSQ)
CTB=COS(THES)
STB=SIN(THES)
CPB=COS(PHIS)
SPB=SIN(PHIS)
CTBCPB=CTB*CPB
CTBSPB=CTB*SPB
STBCPB=STB*CPB
STBSPB=STB*SPB
RCS=RD
TCS=TD
PLCS=PD
IF(ITEM.LT.2)GO TO 60

CCOMPUTE MEAN (SMOOTHED) VELOCITY CCMPONENTS (ELECTRONIC
COMPUTATION)

VXE=RANS D*CTBCPB-RANS*(STBCPB*THESD+CTBSPB*PHISD)
VYE=RANS D*STBCPB+RANS*(CTBCPB*THESD-STBSPB*PHISD)
VZE=RANS D*SPB+RANS*CPB*PHISD
VXE=VXE+EMDTA*(VYES-VXE)
VYE=VYE+EMDTA*(VYES-VYE)
VZE=VZE+EMDTA*(VZES-VZE)
VXES=VXE
VYES=VYE
VZES=VZE
GO TO 60

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CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
65 DC 92 I=1,4
92 IFLAG(I)=IFLAGS(I)+IPRINT(I)
C
PRINT PK AS A FUNCTION OF AIRCRAFT ASPECT AND IMPACT SPEED
C
IF(IPRINT(5).EQ.0)GO TO 80
CALL PRSEGS(SPKT2,ISL)
C
COMPUTE PK AS A FUNCTION OF ASPECT AND IMPACT SPEED FOR ALL GUNS
C
80 DC 36 I=1,32
DC 36 J=1,8
PK=RHO(I)*SPKT2(I,J)
36 SPKTOT(I,J)=PK+(1.0-PK)*SPKTOT(I,J)
C
STORAGE OF PK VS DENSITY FACTOR AND TIME INTERVALS (AT FIRE AND
INTERCEPT) PER WEAPON OR WEAPON COMPLEX
C
CPK=1.0-CPS
DC 55 I=1,NRHOS
D=RHO(I)
PK=D*CPK
PKTIDC(I)=PK+(1.0-PK)*PKTTDC(I)
DC 55 J=1,NTINTS
D1=D*PTOTIF(J)
PKTFDC(J,I)=D1+(1.0-D1)*PKTFDC(J,I)
D2=D*PTOTTI(J)
55 PKTIDC(J,I)=D2+(1.0-D2)*PKTIDC(J,I)
C
COMPUTE, STORE, AND WRITE TOTAL PKS FOR ENTIRE ARRAY OF WEAPCNS
C
F=FLOAT(NROUND/ISB)*IPERS
WRITE(7)ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PTOTIF,PTOTTI,IPRINT
ASSIGN 70 TC IE0F
GC TO 15
70 ENDFILE 7
LINE=66
REWIND 7
PUNCH2 CARDS WRITE ONLY ONE CARD IMAGE ON TAPE4 WHEN THERE IS

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ONLY ONE DEFENSE PER EMPLOYMENT (I.E., ONLY ONE "12" CARD
BEFORE 7/8/9 END-OF-RECORD).

K = ISL
CALL PAGES(4,0,JP)
WRITE (6,1033)
ASSIGN 40 TC IE OF
79 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTTI,IPRINT
1 IF (EOF(7).NE.0.) GO TO IE OF (4,12,40,70,83,9997,9999)
IF (IOCHEC(7).NE.0.) GO TO IERR,(3,28,9996)
CALL PAGES(1,4,JP)
IF (JP.EQ.0) WRITE (6,1033)
WRITE (6,1034) ISL,CPK,NROUND,F,XGUN,YGUN,ZGUN,CIRCLE,IGL,
1 TREAT,TTRACK,IGT,IEM,ICB,ISB,ISL
1 WRITE (4,1060) ITITLE(9),ITITLE(10),ISL,IGT,IEM,ICB,ISB,IGL,CPK,
1 NROUND,XGUN,YGUN,ZGUN,F,NUMBER
GO TO 79
40 IF (K.NE.2) WRITE (4,1064) ITITLE(9),ITITLE(10),PKTDC(1),NUMBER
IF (IFLAGS(2).LE.0) GO TO 76
REWIND 7
CALL PAGES(4,0,JP)
WRITE (6,1031) (I,I=1,10)
77 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTTI,IPRINT
1 IF (IPRINT(2).LE.0) GO TO 77
CALL PAGES(1,4,JP)
IF (JP.EQ.0) WRITE (6,1031) (I,I=1,10)
WRITE (6,1032) ISL,(TINTER(1),I=1,NTINTS)
IFLAGS(2)=IFLAGS(2)-IPRINT(2)
IF (IFLAGS(2).GT.0) GO TO 77
IF (IFLAGS(3).LE.0) GO TO 42
REWIND 7
CALL PAGES(4,0,JP)
WRITE (6,1035) (I,I=1,10)
41 READ(7) ISL,IGT,IEM,ICB,ISB,IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,PLOTTI,IPRINT
1 IF (IPRINT(3).LE.0) GO TO 41
CALL PAGES(2,4,JP)
IF (JP.EQ.0) WRITE (6,1035) (I,I=1,10)
WRITE (6,1018) ISL,(PLOTTF(I),I=1,NTINTS)
DO 410 I=2,NTINTS
PLOTTF(I) = PLOTTF(I-1) + (1.0-PLOTTF(I-1))*PLOTTF(I)
410 WRITE (6,1058) ISL,(PLOTTF(I),I=2,NTINTS)
IFLAGS(3)=IFLAGS(3)-IPRINT(3)
IF (IFLAGS(3).GT.0) GO TO 41
IF (IFLAGS(4).LE.0) GO TO 71
REWIND 7
CALL PAGES(4,0,JP)

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WRITE (6,1036) (I,I=1,10)
43 READ(7) ISL,IGT, IEM,ICB,ISB, IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,POTTF,POTTI,IPRINT
IF(IPRINT(4).LE.O) GO TO 43
CALL PAGES(2,4,JP)
IF(JP.EQ.O) WRITE (6,1036) (I,I=1,10)
WRITE(6,1018) ISL, (POTTI(I),I=1,NTINTS)
DC 430 I=2,NTINTS
POTTI(I) = POTTI(I-1) + (1,0-POTTI(I-1))*POTTI(I)
430 WRITE (6,1058) ISL,(POTTI(I),I=2,NTINTS)
IFLAGS(4)=IFLAGS(4)-IPRINT(4)
IF(IFLAGS(4).GT.O) GO TO 43
71 IF(IFLAGS(1).LE.O) GO TO 44
CALL REWIND 7
CALL PAGES(4,0,JP)
WRITE (6,1017) (I,I=1,9)
74 READ(7) ISL,IGT, IEM,ICB,ISB, IGL,XGUN,YGUN,ZGUN,TREACT,TTRACK,
1 CIRCLE,NROUND,F,CPK,NRHOS,RHO,NTINTS,TINTER,POTTF,POTTI,IPRINT
IF(IPRINT(1).LE.O) GO TO 74
CALL PAGES(1,4,JP)
IF(JP.EQ.O) WRITE (6,1017) (I,I=1,9)
WRITE(6,1055) ISL,(RHO(I),I=1,NRHOS)
IFLAGS(1)=IFLAGS(1)-IPRINT(1)
IF(IFLAGS(1).GT.O) GO TO 74
44 CALL PAGES(7,NTINTS,0,JP)
WRITE (6,1019)
WRITE (6,1021) (I,I=1,9)
POTTF(I) = PKTFDC(1,I)
DC 440 I=1,NTINTS
IF (I.EQ.1) GO TO 440
POTTF(I) = POTTF(I-1) + (1,0-POTTF(I-1))*PKTFDC(1,I)
440 WRITE (6,1018) I,POTTF(I),(PKTFDC(I,J),J=1,NRHOS)
CALL PAGES(9,NTINTS,0,JP)
WRITE(6,1020)
WRITE (6,1021) (I,I=1,9)
POTTI(I) = PKTIDC(1,I)
DC 441 I=1,NTINTS
IF (I.EQ.1) GO TO 441
POTTI(I) = POTTI(I-1) + (1,0-POTTI(I-1))*PKTIDC(1,I)
441 WRITE (6,1018) I,POTTI(I),(PKTIDC(I,J),J=1,NRHOS)
CALL PRSEGS(5PKTOT,0)
GO TO 95
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
EEEE X X X I I I I T T T T S S S S
EEE X X X I I I S S S
C C C C

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1006 FORMAT(/////*-IMPRCPR INPUT CARD ENCOUNTEREC.  **,I2,7A10,A8,***)
1007 FORMAT(I2,7A10,A8)
1008 FORMAT(6X,9E8.0)
1009 FORMAT(1X,5I1,E8.0)
1010 FORMAT(*-VULNERABLE AREA (SQ. METERS) AS A FUNCTION OF IMPACT *,
1 152 305 457 613 762 914 1067 1219*/)
2 *SPEED (METERS/SEC) AND ASPECT VIEW*/5X,8A10//* VIEW*,7X*0
1011 FORMAT(1H-,I2,* TIME INTERVALS FOR PK ACCUMULATION*/4X*0.00*,
1 10F8.2)
1012 FORMAT(I2,3F7.2,3F6.2,F6.0,2F12.7,2F6.1,2F7.0,F8.2,2F8.3)
1013 FORMAT(* LOCATION*,I5,6X,*GUN TYPE*,I2,6X,*ERRCR CODE*,I2,6X,*OL
1 *POSITION=(*,F8.1,**,F8.1,*),RADCLS=*,F6.1,* M*/*OL
2 FIRE MEAN FLT. INTCP * FIRE TIME *CLCSE * AZIM. ELEV.
3 MEAN SIG2 MEAN BIAS1 BIAS2 VULN*/* N DEM VEL. RATE RANGE
4 AREA SHOT PK CUM. PK*/)
5
1014 FORMAT(I2,13,2F7.2,F8.2,2F7.0,1X,2F6.1,1X,2F7.1,F8.1,1X,2F7.2,1X,
3F7.2,2F10.6)
1015 FCRMAT(3X,11,1X,11,9E8.0)
1016 FCRMAT(6I1)
1017 FCRMAT(*- LCC*,12X,9(7X*RRHO*,I2)/)
1018 FCRMAT(I5,10F12.7)
1019 FCRMAT(*-ATTITION ACCRUED AS A FUNCTION OF TIME OF FIRE*)
1020 FCRMAT(*-ATTITION ACCRUED AS A FUNCTION OF TIME AT INTERCEPT*)
1021 FCRMAT(*OTIME*,5X*CUM FOR*,9(5X*DENSTY*)/* SEG.*,5X*CLASS 1*,
9(5X*CLASS*,I2)/)
1023 FCRMAT(IH0,10X,*TOTALS*,9F12.7)
1024 FCRMAT(///*-END OF JOB. *,9X*EXECUTION CPU TIME *,F7.2,9X*EXEC*,
*UTION (*OG TIME/ MIN MAX *F7.2,9X*TOTAL CPU TIME *,F7.2)
1025 FCRMAT(*OG TIME/ MIN MAX *F7.2,9X*TOTAL CPU TIME *,F7.2)
2 ALLISTIC BALLISTIC VEL MAX AZIM ELEV MAX MUZZ
3X.EL*/* T ROUND ELEV VEL RATE RANGE SMOOTH MAX.AZ
4N.EL*/* 1 CONSTANT 2 MIN MAX RATE TOF1 TOF2 VEL
5RROR*/)
1026 FCRMAT(I5,4X,*0.00*,8F8.2)
1027 FCRMAT(///*-GUN TYPE*,I2,**, ERRCR CODE*,I2,**. COMBINATION INVAL
1IC.*)
1028 FCRMAT(I4,I2,9E8.0)
1029 FCRMAT(1H-,I1,* DENSTY CLASSES FOR PK ACCUMULATION*/9F12.5)
1031 FCRMAT(*- LCC*,10(6X,*TIME*,I2)/)
1032 FCRMAT(I5,10F12.2)
1033 FCRMAT(*- LCC*,5X*P(KILL)*,6X*ROUNDS*,3X*FIRE TIME*,8X*XGUN*,
1 8X*YGUN*,8X*ZGUN
2 *GT EM CB SB LOC*)
1034 FCRMAT(I5,F12.7,I12,4F12.2,I3,2F12.2,4I3,15)
1035 FCRMAT(*- LCC*,10(4X,*PK(TI*,I2,**)//)
1036 FCRMAT(*- LCC*,10(4X,*PK(TI*,I2,**)//)
1037 FCRMAT(///*-UNRECOVERABLE PARITY ERROR DETECTED. CALL EXIT.*)

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1038 FCRMAT(13,15,F10.0)
1039 FCRMAT(* ECM SHOULD NOT BE SPECIFIED WHEN IEM IS 1 OR 2 *)
1040 FCRMAT(* MULTIPATH TRACKING ERROR SPECIFIED WITH IEM=3 ONLY *)
1041 FCRMAT(13,13,12,3F10.0,15,2F10.0,15,F10.0)
1042 FCRMAT(//,* ECM VARIABLES*)
1043 FCRMAT(*1AAASIM-----*,10A10)
1044 FCRMAT(1X,F7.2,4X,F8.3,2X,F9.2,5X,F7.2,5X,F7.2,
15X,12,3X,E10.4,5X,E10.4,5X,F7.2)
1045 FCRMAT(//,TIME RANGE(M) X-SEC(SQM) JAM-GAIN(DB) J/S(DB)*,
1* FIRE ADD. TRK. VAR. ADD. RNG. VAR. S/N(DB)*)
2* MODE
1058 FCRMAT(15,*CUM*,9X,9F12.7)
1059 FCRMAT(15,12X,9F12.5)
1060 FCRMAT(2A9,13,512,F10.7,15,2F7.0,F6.0,F6.1,16)
1061 FCRMAT(1H0,4X*DATE*,7X*TIME*,4X*LCC GT EM CB SB GL*,5X*P(K*,
*ILL) RDS XGUN YGUN ZGUN F TIME PAGE*/)
1062 FCRMAT(2A9,A3,5A2,A10,A5,2A7,3A6)
1063 FCRMAT(2(2X,A9),2X,A3,5(2X,A2),2X,A10,1X,A5,2(1X,A7),3(2X,A6))
1064 FCRMAT(2A9,13X,F10.7,31X,16)
END
SUBROUTINE PAGES(N,NT,JP)
KEEPS NUMBER OF LINES PER PAGE LESS THAN 59, PRINTS HEADER,
AND GETS TIME INFORMATION FROM SYSTEM. REPLACES
NPAGE(MAX) AND HEADER IN AFATL PROGRAM.
N NUMBER OF LINES TO BE PRINTED BEFORE NEXT CALL TO PAGES.
NT NUMBER OF LINES IN TITLE OR HEADER OF DATA BEING PRINTED.
IF CALL TO PAGES IS TO PRINT HEADER ONLY, "N" SHOULD
BE NUMBER OF LINES AND "NT" SHOULD BE ZERO.
JP FLAG FROM PAGES, SET TO ZERO WHEN A NEW PAGE IS STARTED
INDICATING NECESSITY TO PRINT HEADER.
COMMON /BLOCK1/ ITITLE(10)
COMMON /HEADFO/ LINE,NUMBER
JP = 2
LINE = LINE + N
IF (LINE .LT. 59) RETURN
NUMBER = NUMBER + 1
CALL TIME(ITITLE(10))
WRITE(6,1000) ITITLE,NUMBER
LINE = 2 + N + NT
JP = 0
RETURN
1000 FCRMAT(*1AFATL P-COI AAASIM **,*7A10,*",*A10,2(1XA9),4X*PAGE*,14/)
END
SUBROUTINE PRSEGS(P,ISL)
PRINTS THE PK AS A FUNCTION OF ASPECT AND IMPACT SPEED TABLES

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C C C
THIS SUBROUTINE EXTENSIVELY MODIFIED TO PRINT ASPECT SECTOR
  ANGLES AND PROPERLY LABEL THE TWO CASES FOR WHICH IT
  PRINTS TABLES.
COMMON /BLOCK3/ XGUN, YGUN, ZGUN
COMMON /BLOCK4/ IGT, IEM, ICB, ISB, IGL, CIRCLE
DIMENSION P(32,8), PT(8), ANG(8)
DATA ANG/7H315-360,7H000-045,7H045-090,7H090-135,7H135-180,
7H180-225,7H225-270,7H270-315/
1 PK=PT(1)=PT(2)=PT(3)=PT(4)=PT(5)=PT(6)=PT(7)=PT(8)=0.0
CALL PAGES(6,0,JP)
IF (ISL.GT.0) WRITE (6,1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6,1030)
WRITE(6,1040)
DC 2 I=1,32
IAZ = 1 + MCD(I,8)
IEL = 2 + (I-1)/8
PP=0.0
DC 3 J=1,8
PT(J)=PT(I,J)+(1.0-PT(J))*P(I,J)
PK = PK + (1.0-PK) * P(I,J)
3 PP=PP+(1.0-PP)*P(I,J)
CALL PAGES(1,6,JP)
IF (JP.NE.0) GO TO 2
IF (ISL.GT.0) WRITE (6,1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6,1030)
WRITE (6,1040)
2 WRITE (6,1041) I, ANG(IAZ), ANG(IEL), (P(I,J), J=1,8), PP
CALL PAGES(2,6,JP)
IF (JP.NE.0) GO TO 4
IF (ISL.GT.0) WRITE (6,1001) ISL, IGT, IEM, XGUN, YGUN, ZGUN, CIRCLE
IF (ISL.EQ.0) WRITE (6,1030)
4 WRITE (6,1042) PT, PK
RETURN
1001 FCFORMAT(*,PK AS A FUNCTION OF ASPECT SECTOR AND IMPACT SPEED.*,
1 5X*LOC*,14,5X*GT*,12,5X*EM*,12,5X*X*,F7.0,3X*Y*,F7.0,3X*Z*,
1030 F7.0,5X*RADIUS*,F5.0,*M*)
2 FORMAT(*-TOTAL PK FOR DENSITY CLASS 1 AS A FUNCTION OF ASPECT*,
1 * SECTOR AND IMPACT SPEED.*)
1040 1 FORMAT(*3SECTOR AZIMUTH ELEV.*,8X*0-152 152-305 *?,
1 3X*1067-1219 610-762 762-914 914-1067*,
1041 FCFORMAT(17,2X,A7,8F12.7,F14.7)
1042 FCFORMAT(1H0,18X*TOTALS*,8F12.7,F14.7)
END
SUBROUTINE TPLOT(NFPA)

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```

C
C
C      PLOTS X VS. Y AND X VS. Z ON PRINTER FOR EACH FLIGHT PATH.
      NO PLOT WHEN XMAX-XMIN IS LESS THAN 50.

      COMMON XEPA(1201), YFPA(1201), ZFPA(1201),
             BFPA(1201), AFPA(1201), PFPA(1201),
             VXFPA(1201), VYFPA(1201), VZFPA(1201)
      DIMENSION PLOT(11)
      LOGICAL XA
      DATA BLANK, YAXIS, XAXIS, POINT/1H, 1H:, 1H-, 1H+/
      PLOT = MAXO(NFPA/50, 1)
      IFPA = MIN-MAX VALUES OF X, Y, AND Z (ZMIN = 0. BY DEFN.)
      XMIN = YMIN = 1.E99
      XMAX = YMAX = ZMAX = -1.E99
      DO 100 I=1, NFPA, IFPA
      XMIN = AMIN1(XMIN, XEPA(I))
      YMIN = AMIN1(YMIN, YFPA(I))
      XMAX = AMAX1(XMAX, XEPA(I))
      YMAX = AMAX1(YMAX, YFPA(I))
      ZMAX = AMAX1(ZMAX, ZFPA(I))
      CHECK X RANGE TO AVOID FUNNY X-Z PLOT.
      X RNG = XMAX - XMIN
      IF (X RNG .GT. 49.999) GO TO 110
      CALL PAGES(3, 0, 1)
      WRITE (6, 210)
      RETURN
      SCALE X, Y TE METERS PER CHARACTER.
      D = AMAX1(X RNG/11., (YMAX-YMIN)/7.25)
      DX = D/10.
      DY = D/8.
      CENTER = ((XMAX+XMIN)/2. - 55.*DX,
                (YMAX+YMIN)/2. - 29.*DY)
      XMIN = (XMAX+XMIN)/2.
      YMIN = (YMAX+YMIN)/2.
      XMAX = XMIN + 110.*DX
      YMAX = YMIN + 58.*DY
      SCALE Z (NOT TO SAME SCALE AS X-Y).
      DO 120 I=1, 100
      D = FLOAT(I*500)
      IF (D .GT. ZMAX) GO TO 130
      CONTINUE
      120 DZ = D/10.
      SEE IF AXES SHOULD APPEAR ON PLOT.
      XA = .FALSE.
      NYA = .TRUE.
      IF (XMIN .LE. 0.) AND. XMAX .GE. 0.) XA = .TRUE.
      IF (YMIN .LE. 0.) AND. YMAX .GE. 0.) NYA = .TRUE.
      IF (XA) NYA = 1 + INT(0.5-XMIN/DX)
      IF (NYA) XA = 1 + INT(0.5-YMIN/DY)
      SET 8 LINES PER INCH SPACING ON PRINTER.
      130
C

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C
WRITE (6,280)
SET PAGE CONTROL, PLOT X-Z, THEN X-Y.
CALL PAGES(56,0,1)
WRITE (6,250)
WRITE (6,220)
IFPA = 2 * IFPA
DO 160 K=1,11
DO 140 I=1,111
PLOT(I) = BLANK
140 IF (K.EQ.11) PLOT(I)=XAXIS
IF (K.AND. K.NE.11) PLOT(NXA) = YAXIS
Z = FLOAT(11-K)*DZ
DO 150 I=1,NFPA IFPA
IX = 1 + INT(0.5+(XFPA(I)-XMIN)/DX)
150 IF (K.EQ.11-INT(0.5+ZFPA(I)/DZ)) PLOT(IX) = POINT
160 WRITE (6,220)
WRITE (6,250)
IFPA = IFPA/2
DO 190 K=1,59
DO 170 I=1,111
PLOT(I) = BLANK
170 IF (K.EQ. NYA) PLOT(I) = XAXIS
Y = YMIN + FLOAT(59-K)*DY
DO 180 I=1,NFPA IFPA
IX = 1 + INT(0.5+(XFPA(I)-XMIN)/DX)
180 IF (K.EQ.59-INT(0.5+(YFPA(I)-YMIN)/DY)) PLOT(IX) = POINT
IF (MOD(K,2)
I = (I.NE.1) WRITE (6,240) PLOT
IF (I.EQ.1) WRITE (6,250) PLOT,Y
WRITE (6,250)
PRINT X,XAXIS VALUES.
D=10.*DX
C
DO 200 I=1,12
PLOT(I) = XMIN + D*FLOAT(I-1)
200 WRITE (6,270) (PLOT(I),I=1,12),DX,DY
RE-SET PRINTER SPACING BACK TO 6 LINES PER INCH.
WRITE (6,290)
RETURN
C
210 FORMAT(*ONEARLY CONSTANT X IN FLIGHT PATH. NC PLOT PRINTED.*/)
220 FORMAT(1X,113(1H*))
230 FORMAT(2H *,111A1,1H*,3X*Z **,F8.1,* M*)
240 FORMAT(2H *,111A1,1H*)
250 FORMAT(2X*AA*, 9X*BB*, 9X*CC*, 9X*DD*, 9X*EE*, 9X*FF*,
1 9X*GG*, 9X*HH*, 9X*II*, 9X*JJ*, 9X*KK*, 9X*LL*)

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260 FORMAT(2H * 11111,1H * 13X * Y = *,F8.1, * M*)
270 FORMAT(4X * X(A) = *,F8.1, * M*, 4X * X(B) = *,F8.1, * M*, 4X * X(C) = *,F8.1, * M*,
1 4X * X(D) = *,F8.1, * M*, 4X * X(E) = *,F8.1, * M*, 4X * X(F) = *,F8.1, * M*,
2 4X * X(G) = *,F8.1, * M*, 4X * X(H) = *,F8.1, * M*, 4X * X(I) = *,F8.1, * M*,
3 4X * X(J) = *,F8.1, * M*, 4X * X(K) = *,F8.1, * M*, 4X * X(L) = *,F8.1, * M*,
4 4X * X(O) = *,F8.1, * M*, 4X * X(P) = *,F8.1, * M*, 4X * X(Q) = *,F8.1, * M*,
280 FORMAT(11H * 13X * Y = *,F8.1, * M*)
290 FORMAT(11H * 13X * Y = *,F8.1, * M*)
END
SUBROUTINE RPLANE(T)
CCMPUTES INFORMATION ABOUT THE POSITION OF THE AIRCRAFT AT TIME = T
CC
COMMON/BLOCK2/NFPA,TMIN,TMAX,DTFPA
COMMON/BLOCK3/XGUN,YGUN,ZGUN
COMMON/NFPARM/XA,YA,ZA,RA2,RA,TIME
COMMON/IGXGYG/IG,XG(8),YG(8)
COMMON/XFPA(1201),YFPA(1201),ZFPA(1201),BFPA(1201),AFPA(1201),
1 PFPA(1201),VXFPA(1201),VYFPA(1201),VZFPA(1201)
* COMMON/CECM1/IRECM,IJ,GAINJ,IX,XSEC,CALX,PJh,
* X,Y,Z,ROL,PIT,PDG,
* FTGT,FJAM,GJ,SJT,SN
* CALL INTERP((T+TIMEI)/DTFPA)
XA=GETVAL(XFPA)-XGUN-XG(IG)
YA=GETVAL(YFPA)-YGUN-YG(IG)
ZA=GETVAL(ZFPA)-ZGUN
RA2=XAT*XA+YA*YA+ZA*ZA
RA=SQRT(RA2)
RETURN
END
FUNCTION ANGLIM(X)
LIMITS ANGLES TO PRINCIPAL ANGLES BETWEEN -PI AND +PI
CC
COMMON /CONSTS/ DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
IF(ABS(X)-PI)1,1,2
1 ANGLIM=X
RETURN
2 ANGLIM=X-PI2*FLOAT(IFIX((X+SIGN(PI,X))/PI2))
RETURN
END
FUNCTION RSHELL(T)
CCMPUTES RANGE TO SHELL AT TIME = T
CC
COMMON/VASBS/VMUZZ,ASHCON,BSHCON,CQUAD
DQUAD=1.0+T*(ASHCON+T*BSHCON)
RSHELL=T*VMUZZ/DQUAD

```

```

RETURN
END
FUNCTION VSHELL(T)
  CCMPUTES SPEED OF SHELL AT TIME=T. CAN ONLY BE USED AFTER A CALL
  TO RSHELL AT THE SAME TIME, SINCE RSHELL COMPUTES DQUAD FOR VSHELL
  COMMON/VSASBS/VMUZZ,ASHCON,BSHCON,CQUAD
  VSHELL=VMUZZ*(1.0-BSHCON*T)/(DQUAD*DQUAD)
  RETURN
END
SUBROUTINE INTERP(FINT)
  SETS CONSTANTS (FRACT, INDEX1, ANC INDEX2) FOR TWO POINT INTER-
  POLATION
  COMMON/MAGIC/FRACT,INDEX1,INDEX2
  INDEX1=FINT
  FRACT=FINT-FLOAT(INDEX1)
  INDEX1=INDEX1+1
  INDEX2=INDEX1+1
  RETURN
END
FUNCTION GETVAL(ARRAY)
  PERFORMS TWO POINT INTERPOLATION
  COMMON/MAGIC/FRACT,INDEX1,INDEX2
  DIMENSION ARRAY(1201)
  GETVAL=ARRAY(INDEX1)+FRACT*(ARRAY(INDEX2)-ARRAY(INDEX1))
  RETURN
END
BLOCK DATA
COMMON /BLOCK1/ ITITLE(10)
COMMON /BLOCK3/ XGUN,YGUN,ZGUN
COMMON /BLOCK4/ IGT,IEM,ICB,ISB,IGL,CIRCLE
COMMON /BLOCK5/ NRHOS,RHO(9)
COMMON /BLOCK6/ NTINTS,TINTER(10)
COMMON /BLOCK7/ IVACOM(8),VAT(9,5,9)
COMMON /BLOCK8/ TREATCT,TRACK1,TRACK2
COMMON /BLOCK9/ TROUND(6),THDMAX(6),PHDMAX(6),PHIMIN(6),PHIMAX(6),
1 PHDMIN(6),VELMAX(6),RANMIN(6),RANMAX(6),ATLAG(6),
2 ETHMAX(6),EPHMAX(6),RMODES(6)
COMMON /BLOCKA/ TFMAX1(6),TFMAX2(6),RVACON(6),RVBCON(6),VMUZEL(6)
COMMON /CONSTS/ DEGREE,RADIAN,PI,PI2,QTRPI,SQRT2
COMMON /HEADFO/ LINE,NUMBER
COMMON /BUDGET/ BDACON(6)
VALUES ON "ASIDQC" CARDS FROM A.S.I. DOCUMENTATION DRAFT. (BEE)

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```

DATA ATLAG /2*9999.99,1.33,9999.99,2*1.33/
DATA BDACON /00265,00501,0031,00697,00113,00113/
DATA DEGREE /57.295779513082, RACIAN/0.01745325251994/
DATA EPHMAX /6*.1/ IEM/4/, ICB/1/, ISB/4/, IGL/1/, CIRCLE/0./
DATA IGT /3/ IEM/4/, ICB/1/, ISB/4/, IGL/1/, CIRCLE/0./
DATA ITITLE /7*(1H ),10H ASD/XRDA,1H,1H /
DATA IVACOM /8*(1H )//
DATA LINE /66/, NUMBER/0/
DATA NRHOS /9/
DATA NTINTS /10/
DATA PHDMAX /43633,43633,78540,34910,31416,3491/
DATA PHIMAX /1.48353,1.57079,1.48353,1.48353,1.51844,1.43117/
DATA PHIMIN /-1.17453,-1.14835,-1.17453,-1.17453,-.08727,-.06981,-.05236/
DATA PI /3.1415926535898/, PI2/6.2831853071796/
DATA QTRPI /0.78539816339745/
DATA RANMAX /3000,3000,3300,5000,5500,59599./
DATA RANMIN /0.0,0.0,0.0,0.0,0.0,0.0/
DATA RHO /1.5,3333333333,25.2,1666666666,1.02,.01/
DATA RMODES /-1.400,-1.400,-1.500,0.0,0.0,0.0/
DATA RVACON /1.80209,251499,22958,089321,07845,050694/
DATA RVBACON /0.12392,-.006259,-.006889,004262,-.000421,-.000357/
DATA SORT2 /1.4142135623731/
DATA SQRT2 /1.4142135623731/
DATA TFMAX CASE, BUT NOT EXACT AGREEMENT WITH AFATL PROGRAM ON TEST (BEE)
DATA TFMAX2 /1.6,2.2,3.8,4.1,6.2,9.99/
DATA THDMAX /99.99,99.99,7.5,99.99,11.6,19.2/
DATA TINTER /0.5236,5236,1.39626,0.5236,0.5236,5236/
DATA TREACT /10.20,30.40,50.60,70.80,90.99,99/
DATA TROUND /0.75,4.2,75.857,4./
DATA VAI /405*0/
DATA VELMAX /300,300,350,250,300,999.9/
DATA VELMIN /0.0,0.0,0.0,99.9/
DATA VMUZEL /840,1000,930,880,560.0,800./
DATA XGUN /0./
DATA YGUN /0./
DATA ZGUN /0./
END
SLBROUTINE AVG(ITST,IPRINT,ISW)
DIMENSION IPRT(6)
DATA IBLNK,SUM,CNT/2H ,0.0,0.0/
DATA IALL/2HAL/
CECODE(10,900,IPRT(1)) PK
FORMAT(10.7)
DECODE(7,910,IPRT(4)) YGUN
FORMAT(F7.0)
IF(ISW.EQ.IALL) GOTO 600
IF(ITST.EQ.IBLNK) GOTO 200

```

900  
910

C C



```

150 IF(YGUN-NE,0.0) GOTO 150
    CNT=CNT+0.5
    SUM=SUM+PK/2.
    POLD=0.0
    GOTO 500
150 CONTINUE
    CNT=CNT+1.0
    SUM=SUM+PK
    POLD=PK
    GOTO 500
200 IF(CNT.LE.0.5) GOTO 500
    CNT=CNT-0.5
    SUM=SUM-POLD/2.0
    AVERG=SUM/CNT
    WRITE(6,800) CNT,AVERG
    FORMAT(75X,18H AVERAGE P(KILL) ON,F6.1,20H OFFSET LOCATIONS IS,
    800 X F13.7/)
    SUM=0.0
    CNT=0.0
    RETURN
500 IF(ITST-NE,IBLNK) GOTO 610
600 IF(CNT-NE,0.0) GOTO 201
    GOTO 500
610 SUM=SUM+PK
    CNT=CNT+1.0
    GOTO 500
END
SLBROUTINE MULTPH(1,REFC,EL,BIAS,SC2)
DIMENSION C(3),S(3),B(3)
DATA
    *,AK/-0.6931471806/,
    *,SQR2/1.414213562/,
    *,B/0.0244346,0.0314159,0.0785398/,
    *,C/0.0132557,0.0185345,0.0478558/,
    *,S/0.00872665,0.0104720,0.0244346/,
    BW=B(1)
    CAL=C(1)
    SQR=S(1)
    DIR=EXP(AK*(SQ/BW)**2)
    EL2=2.*EL
    RL=EXP(AK*((EL2+SQ)/BW)**2)
    DRU1S=EXP(AK*((EL2-SQ)/BW)**2)
    DRU2S=(DIR+REFC*RU)**2
    DRL1S=(DIR+REFC*RL)**2
    DRL2S=(DIR+REFC*RL)**2
    DIF1=(DRU1S-DRL1S)
    SUM1=(DRU1S+DRL1S)

```

```

CIF2 = (DRU2S-DRL2S)
SUM2 = (DIF1/SUM1)
S1GER1 = DIF2/SUM2
S1GER2 = CAL*S1GER1
ANG1ER1 = ABS(ANG1ER2-ANG1ER1)/2.
PPBY2 = PPBY2/SQRT2
BIAS = SD*SD
RETURN
END
SUBROUTINE ECM1
DIMENSION RGDB(3), PRW(3), FREQ(3), IRTYP(4), RNOISE(3)
DIMENSION TABJ(37,37), TABX(37,37)
COMMON /BLOCK1/ ITITLE(10)
COMMON /HEADFO/ LINE, NUMBER
COMMON
C/CECM1/ IREC, IJ, GAINJ, IX, XSEC, CALX, PJW,
X, Y, Z, ROL, PIT, HDG,
F1GT, FJAM, GJ, SJT, SN
**
**
NAMELIST /NAM1/ RGDB, PRW, FREQ, IRTYP, I, RG, WL, F1GT, FJAM, PJW
DATA
* RGDB/40., 38.5, 28./
*, PRW/10500., 17500., 25000./
*, FREQ/15.1E9, 9.3805E9, 2.838E9/
*, IRTYP/1, 2, 2, 3/
*, PI4/12.56637061/
*, RNOISE/-123.0, -130.6, -132.2/

C
I = IRTYP(IREC)
RG = 10.*(RGDB(I)/10.)
WL = 2.998E8/FREQ(I)
RN = RNOISE(I)
F1GT = PRW(I)*RG*WL*WL/PI4/PI4/PI4
FJAM = PJW*RG*WL*WL/PI4/PI4
CALL PAGES(3, 0, JJP)
IF(IJ.EQ.0) WRITE(6, 9003) GAINJ
FORMAT(/, *, JAMMER ANTENNA GAIN*, F7.3, * DB*)
IF(IJ.NE.0) WRITE(6, 9004)
FORMAT(/, *, JAMMER TABLE SPECIFIED*)
CALL PAGES(3, 0, JJP)
IF(IX.EQ.0) WRITE(6, 9005) XSEC
FORMAT(/, *, AIRCRAFT CROSS SECTION *, F9.2, * SQ.METERS*)
IF(IX.NE.0) WRITE(6, 9006) CALX
FORMAT(/, *, AIRCRAFT CROSS SECTION TABLE SPECIFIED. *, F9.2)
** PRINTED VALUES WILL BE MULTIPLIED BY CALX. CALX= *, F9.2)

```

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C
C
IF(IJ.EQ.0) GO TO 1
JAMMER TABLE
CALL TABLR(TABJ,37)
DO 3 I=1,37
DO 3 J=1,37
TABJ(I,J) = 10.**(TABJ(1,J)/10.)
3 CCNTINUE
1 GJ = 10.**(GAINJ/10.)
IF(IX.EQ.0) GO TO 2
C
C
X-SECTION TABLE
CALL TABLR(TABX,37)
2 RETURN
CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
ENTRY ECM2
C
NAMELIST/NAM2/ X,Y,Z,ROL,PIT,HDG,
*CX1,CY1,CZ1,CX2,CY2,CZ2,AZ,EL,GAINJ,XSEC,D2,SJ,ST,SJT
*,GJ
C
IF(IX.EQ.0 .AND. IJ.EQ.0) GO TO 5
CALL DIRCOS(X,Y,Z,0.,0.,0.,CX1,CY1,CZ1)
CALL CARROT(CX1,CY1,CZ1,ROL,PIT,HDG,CX2,CY2,CZ2)
CALL RECSPH(CX2,CY2,CZ2,AZ,EL)
IF(IJ.EQ.0) GO TO 6
CALL INTRP(TABJ,AZ,EL,37,GJ)
6 IF(IX.EQ.0) GO TO 5
CALL INTRP(TABX,AZ,EL,37,XSEC)
XSEC = XSEC*CALX
5 D2 = DIST2(X,Y,Z,0.,0.,0.)
SJ = FJAM*GJ/D2
ST = FTGT*XSEC/D2/D2
SN = 10.*ALOG10(ST)-RN
SJT = 10.*ALOG10(SJ/ST)
RETURN
END
SUBROUTINE JAMER2 (IRAD,AJS,SDSQ)
DIMENSION AJS1(3),SD1(3)
DIMENSION AJS2(4),SD2(4)
DIMENSION AJS3(2),SD3(2)
DIMENSION AJS4(4),SD4(4)
GC TO (1,2,3,4),IRAD
DATA

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* N1/3/ -5., 6., 9./
*, AJS1/ 0.002963, 0.01185, 0.1374 /
*, SD1/ INT2(N1,AJS1,SD1,AJS,SD)
1 GC TO 5
DATA
* N2/4/ -2., 6., 10., 16./
*, AJS2/ 0.002963, 0.009877, 0.08278, 0.1374 /
*, SD2/ INT2(N2,AJS2,SD2,AJS,SD)
2 GC TO 5
DATA
* N3/2/ 15., 30./
*, AJS3/ 0.0018, 0.01441 /
*, SD3/ INT2(N3,AJS3,SD3,AJS,SD)
3 GC TO 5
DATA
* N4/4/ -2., 6., 10., 16./
*, AJS4/ 0.006914, 0.02173, 0.1024, 0.1374 /
*, SD4/ INT2(N4,AJS4,SD4,AJS,SD)
4 SCSQ = SD*SD
5 RETURN
END
SUBROUTINE INT2(NVAL,X,Y,XVAL,YVAL)
DIMENSION X(NVAL),Y(NVAL)
YVAL = Y(1)
IF (X(1)-XVAL) 4,4,3
DO 1 I=1,NVAL
IF (X(I)-XVAL) 1,1,2
1 CONTINUE
1 YVAL = Y(NVAL)
GO TO 3
2 YVAL = Y(I-1) + (Y(I)-Y(I-1)) / (X(I)-X(I-1)) * (XVAL-X(I-1))
3 RETURN
END
SUBROUTINE JAMER1(PLEN,SDSQ)
SDR = PLEN*0.6826/2.*2.998E8
SDSQ = SDR*SDR
RETURN
END
SUBROUTINE TABLR(TABX,IDIM)
C SUBROUTINE TO READ AND PRINT A TABLE CF UP TO 37 X 37 ELEMENTS
C THE PROGRAM PROVIDES A DEFAULT VALUE FOR ELEMENTS OUTSIDE THE
C DEFINED TABLE.
C INPUTS ARE:

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249

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IF(LP.EQ.NPAGE)KH2=MAZ
IF(LP.EQ.NPAGE)JH2=180
WRITE(6,106)((JH,JH=JF1,JH2,IDELEAZ)
106 FORMAT(106) RCS MATRI X*,/* ELEV *,13I9)
JELU=MAXO(JEL-1,1)
MELU=MINO(MEL+1,ICIM)
ELPT=ELO-DELEL*(JEL-JELO)
C PRINT TABLE
DO 15 J=JELC,MELU
WRITE(6,104)ELPT,(TABX(J,K),K=KH1,KH2)
104 FORMAT(1X,F7.1,2X,13F9.2)
15 CONTINUE
20 CONTINUE
C CONV
DELEL=DELEL*CDTR
DELAZ=DELAZ*CDTR
ELO=ELO*CDTR
RETURN
999 CONTINUE
105 WRITE(6,105)
FORMAT(* >>> ERROR IN INPUT <<<*)
STOP
SUBROUTINE CIRCOS(X1,Y1,Z1,X2,Y2,Z2,COSA,COSB,COSG)
XD = X2-X1
YD = Y2-Y1
ZD = Z2-Z1
D = SQR(XD*XD+YD*YD+ZD*ZD)
CCSA = XD/D
COSB = YD/D
COSG = ZD/D
RETURN
END
SUBROUTINE CARROT(X1,Y1,Z1,ROL,PIT,HDG,X2,Y2,Z2)
C CARROT
C HEADNG
X = X1*COS(HDG) + Y1*SIN(HDG)
Y = -X1*SIN(HDG) + Y1*COS(HDG)
Z = Z1
C PITCH
XX = X*COS(PIT)
YY = Y
ZZ = X*SIN(PIT) + Y
C ROLL
X2 = XX
Y2 = -YY*SIN(ROL) + YY*COS(ROL)
Z2 = -YY*SIN(ROL) + ZZ*COS(ROL)

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RETURN
END
SUBROUTINE RECSPH(X,Y,Z,PHI,THE)
THE = ACOS(Z/SQRT(X*X+Y*Y+Z*Z))
THE = THE-1.5708
PHI = 0.0
SB = SQRT(X*X+Y*Y)
IF(SB.NE.C.0) PHI = X/SB
PHI = ACOS(PHI)
PHI = ABS(PHI)
RETURN
END
SUBROUTINE INTRP(TAB,AZ,EL,NVAL,VALUE)
COMMON/TABLES/ELO,DELAZ,DELEL,JEL
DIMENSION TAB(NVAL,1)
A = ABS(AZ)
E = EL
AAZ = A/DELAZ+1.
IAZ = AAZ
IEL = (E-ELO)/DELEL+JEL
IEL = IEL
IAZ = MINO(MAXO(IAZ,1),36)
IEL = MINO(MAXO(IEL,1),36)
V1 = TAB(IEL,IAZ)
V2 = TAB(IEL,IAZ+1)
V3 = TAB(IEL+1,IAZ)
V4 = TAB(IEL+1,IAZ+1)
S = AAZ-IAZ
V12 = V1+(V2-V1)*S
V34 = V3+(V4-V3)*S
S = EEL-IEL
VALUE = V12+(V34-V12)*S
RETURN
END
FUNCTION DIST2(X1,Y1,Z1,X2,Y2,Z2)
XC = X2-X1
YC = Y2-Y1
ZC = Z2-Z1
DIST2 = XC*XD+YD*YD+ZD*ZD
RETURN
END

```

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